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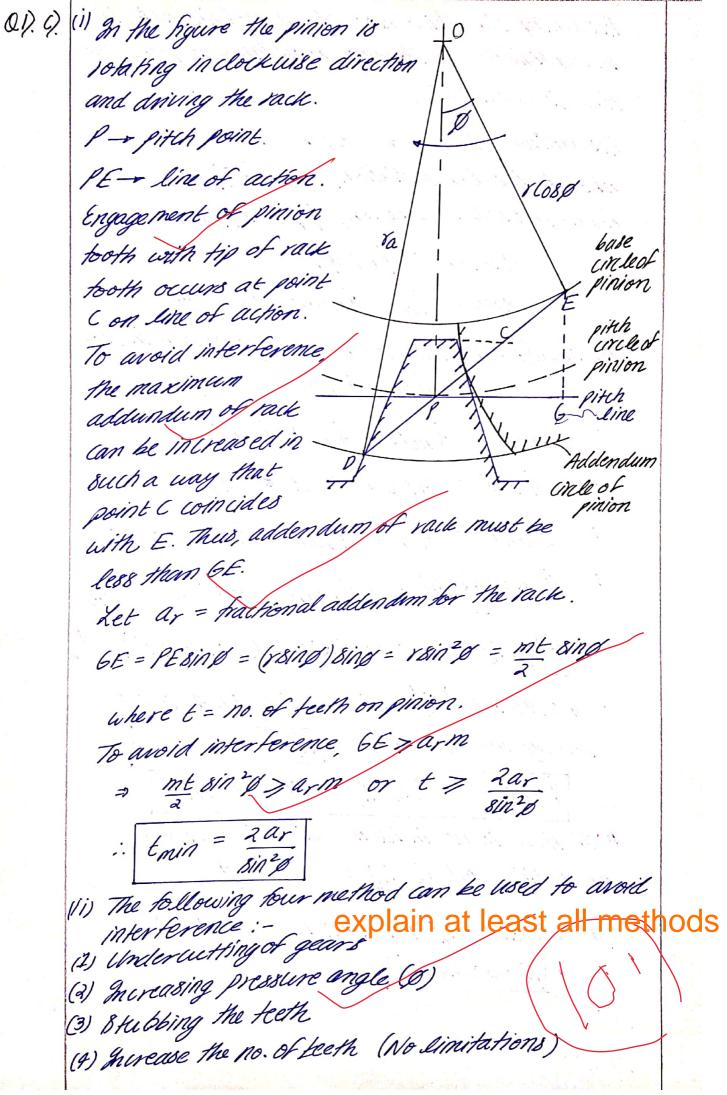
Q1).a). Let Link AR rotate about a fixed point A. Pis the point on a slider on the link. At any given instant, w = angular velocity V608 80 of the link. a = angular aucleration. of the link. v = linear velocity of WY COSSO The slider a = linear acc" of the slider. 80 r = radial dis tame of point P. In short time St, let SO be angular displacement of the link and Sr, the radial displacement of the slider in the doubward direction outward direction. w'= w + ast v = v + ast r'= r + sr 1 Acceleration parallel to AR guital velocity of I along AR = V final velocity of Palong AR = V'COSSO - W'r'sin SO : Acc? of Palong AR = (v+ ast) (os 50 - (w+ ast)(r+sr)sinso-v as St -- 0 = (08 St - 1 and sin 80 = St = Aun along AR = a - wrdo = a-w2r This is the centripetal acc? of slider. B Acceleration of I perpendicular to AR Initial velocity of PI to PR = wr final velocity of PI to AR = V sin 80 + WY (0880 Acc? of PL to AK = (V+ ast) (sin so - (w+dSt) (r+ sr) casso-wr as St -0 ⇒ CO888 - 1 1 8M88 - 88

= Accorp 1 to AR = vdo + wdr +ra ow two tra = 2 wo + tangential acc? we have an extra component of acc? of P in perpendicular direction to link AR apart from tangential acc! This component of acc! is known as corrolis acc? Coriolis Acceleration = 200 The corrolis acc. will be the if:-(1) Link AA rotates C.W. and slider moves radially outwards. (i) link AR rotates C.C.W. and slider moves radially inguards. otherwise, the coriolis component is (ve) Thus, these observations can be scummanised following rule: -The direction of the coriolis aut. component is obtained by rotating the radial velocity vector's through 90° in the direction of rotation of the link. Q1)(b) Assumption: - There is no heat generation in the wall. Wrad Kwall =1.2 W/mK Voolar = 800 W/m 2 T, = 450K space (7 = 0K) Vond. L=0.07m E= 0.85 X = 0.25

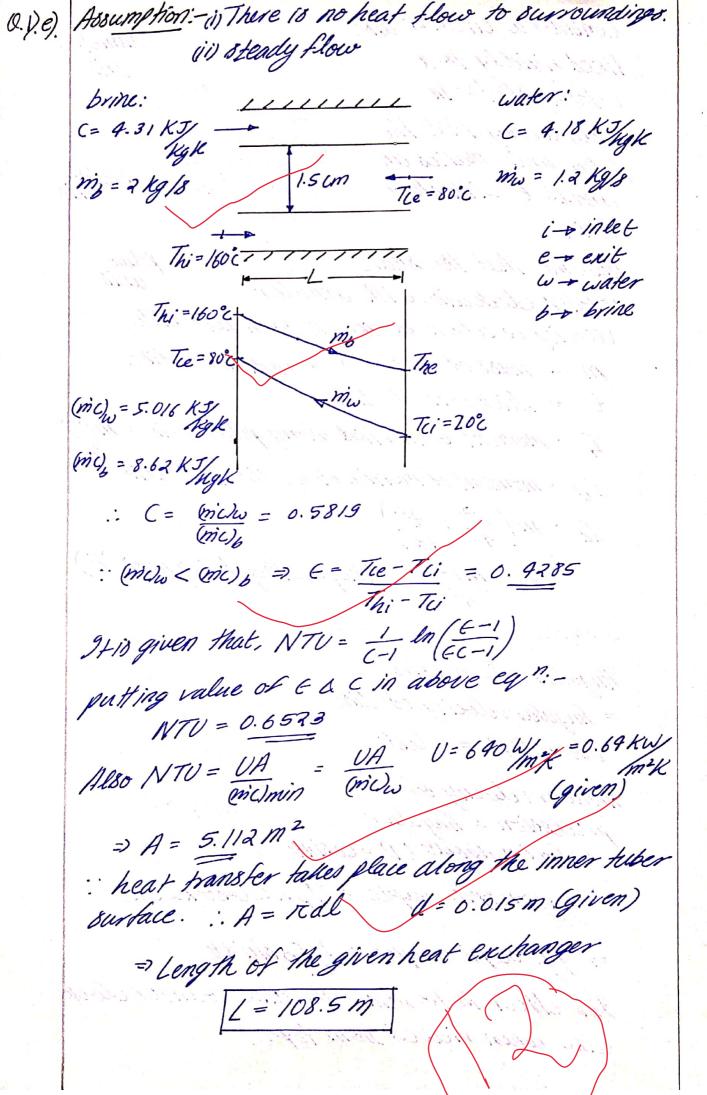
Assuming the thick ness of porcelain tito tiles to be negligible : Thermal resistance of porcelain tiles is negligible. Assuming T=0 = d+p=1 Absorbed solar radiation, ya = & yoular Keflected solar radiation, q, = 1980lar Woolar = qu + qx grotar = qua = grolar - gr (1) At steady state, for the houndary shown in fig. Vin = Wout ýsolar = ýrad + ýr + ýcond. ground = Ysolar grad & - ar Woond = Wa - Wrad (from en 11) à Kw (Ta-Ti) = x àsolar - € 5 Ta 4 (11) $\frac{1.2}{0.07} \left(7 - 450 \right) = 0.25 \times 800 - \left(4.8195 \times 10^{-8} \times 7_2^{-4} \right)$ =) 4.8195 X10 - 8 T + 17.1928 T = 7914.2857 solving the eq. we get T2 = 393.95K : 9 cond. = K (I-T1) = 960.85 W/m2 * If there is no invident light: - Woolar =0 by eq. (11) K(T2-T1) = -EOT24 4.8195 X10-8 Tay 17.428 Tz = 7714.28 = Ta = 383.08 K

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results

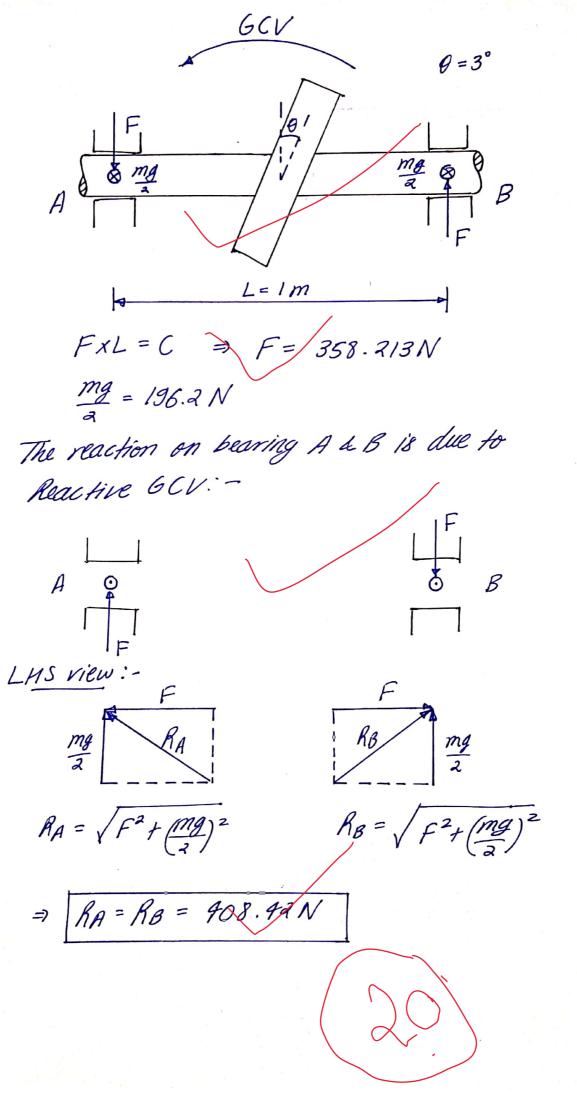


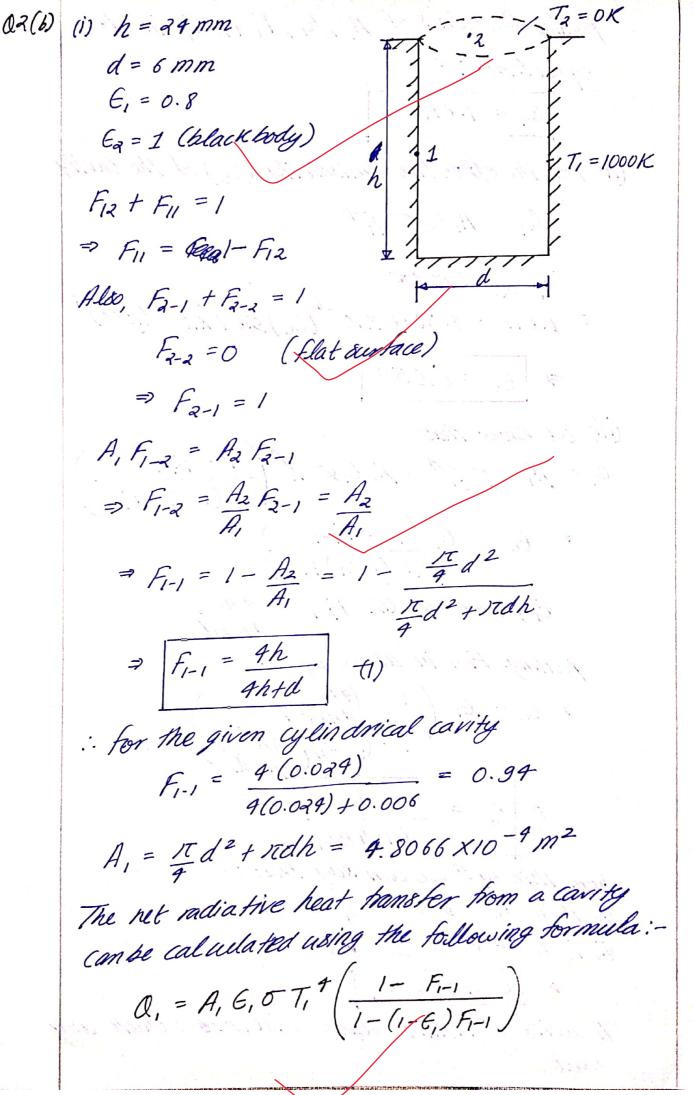
Q12(d) given, diameter of hole = 400 mm =d thickness of hole 215 plate t = 15 mm No. of holes/nin = 30 :. time taken for one hole = cycle time = $\epsilon_c = \frac{60}{2n} = 28e conds$ = 0.6 Kgfm/ = 5.886 Nm, Erengy/ Time taken for junking . Nmax = 160 rpm opension = $t_a = 0.1$ second t = 15Enery Norm = 140 rpm Energy = Energy x redt fig. schemo the diagram for the given punching machine => Energy = 110948.48 Nm Power of the motor, P = Energy/hole = 55474 W Maximum fluctuation of Energy DE = P(tc-ta) = 105901.06 NM Also DE = 1 I (wnax - wmin2) where I = moment of mertia $\omega = \frac{2\pi N}{1-} = angular velocity$ $\Rightarrow \Delta E = \frac{1}{2} I \left(\frac{2\pi}{60} \right)^2 \left(N_{\text{max}}^2 - N_{\text{min}}^2 \right)$ $\Rightarrow \Delta E = \frac{1}{2} I \left(\frac{127}{60} \right)^{2} \left(160^{2} - 190^{2} \right)$ => I = 3203.8 Ky-m2 : The fly wheel required for the given ofercation must have a moment of mertia:-I = 3203.8 kg-m2



Q.2 (a) Considera circular disc dametral fixed rigidly to a acis rolating shaft in such a way that the polar ancis makes on 8 haft axis angle & with shaft x asus. Assume that the shaft rotates clockwise with angwer velocity w when a newed from left side. m - mass of disc r- radius of disc t - thickness of disc. If = moment of inertial along polar axis = mr2 Id = moment of inertia along dimenal axis Id = m/2+ 12/ Id = mr2 (neglecting thickness t') (i) spinning about polar axis:-Angular vel. of spin = Angular velocity of disc about OP = wlost Angular velocity of Precession precession = Angular relocity about OD = Wind gyroscopic couple = Ip x yoloso xw sino ⇒ gyroscopic couple - 1 Ip 81128 It's effect is to rotate the disc counter clock wise when viewed from top.

(i) spinning about dia metal axis
Andigalar velocity of spin
= Angular vel. about OD = W &NO
Angular velocity of precession
= Angular velocity of disc
about OP = W 6080
: gyroscopic couple = Id w Coso w SIMO
⇒ gyroscopic couple = 1 Id w²sin20
Its effect is to votate the disc clockwise when
viewed from the top.
:. Resultant gyroscopic couple on disc
$C = \frac{1}{2} (I_p - Id) \omega^2 8 i n^2 \theta$
$C = \frac{1}{2} \left(\frac{mr^2 - mr^2}{2} \right) \omega^2 8 / h 2 \theta$
$C = \frac{mr^2 \omega^2 8ih ? \theta}{8}$ hence proved.
: Ip > Id : net effect will be counter clockwise
whition of the and with
gris given that, m = 40kg
H=3
$\omega = \frac{217N}{60} = 104.719 rad/8$
$\Rightarrow C = \frac{40(0.25)^2(104.719)^28/106^\circ}{8}$
=) C = 358.213 N-m Counter clockwise when viewed from top.
when viewed from top.





putting the values of A, , E, , Fi-1, T, in above equation :-Q, = 1.611 W (i) For the effective emissivity (Ee) of the Cavity a, = A, Ge 5 T, 9 = to = 0.00 => 1.611 = 4.8066 × 10 -4 (Ee) 5.67 × 10 -8 (1000) 4 \Rightarrow $\epsilon_e = 0.059$ (11) We Know that $Q_1 = \beta_1 \in \beta \mathcal{T}_1^A = \beta_1 \in \beta \mathcal{T}_1^A \left(\frac{1 - F_{1-1}}{1 - (1 - F_1) F_{1-1}} \right)$ $\Rightarrow \quad \epsilon_e = \epsilon_i \left(\frac{1 - F_{i-1}}{1 - (1 - \epsilon_i) F_{i-1}} \right)$ Fer from eq. (1) Fi-1 = 4h 4h+d putting. Fi-1 in above eq.:- $\Rightarrow \epsilon_e = \epsilon_1 \left(\frac{1 - \left(\frac{4h}{h} \right)}{1 - \left(1 - \epsilon_1 \right) \left(\frac{4h}{h + d} \right)} \right)$ $\Rightarrow \epsilon = \frac{\epsilon_i d}{d + 4h\epsilon_i}$ from this ey!" we can say that:-(i) if h = 0 => Ee = E1 (ii) if $h \rightarrow \infty \Rightarrow \epsilon_e \rightarrow 0$ $\Rightarrow 0 < \epsilon_e < \epsilon_1$ refer solution A cavity of infinite depth will note radiate any heat.

0.2(c) Assumption: - heat conduction is in & direction only. A = Aoe ax (given) (i) Assuming a strips of thickness du at distance x de from left side of rod.

Energy balance for the stup:-Oz = Oxt dx (: lateral surface is insulated) $\Rightarrow -KA\frac{dT}{dx} = -KA\frac{dT}{dx} + \frac{d}{dx} \left(-KA\frac{dT}{dx} \right)$ $\Rightarrow KAd^{2T} + (dA)KdT = 0 \quad (:A = f(x))$ Let 0 = To -T 3) de sid $\Rightarrow \frac{d\theta}{dx} = -\frac{dT}{dx}$ $\frac{d^2\theta}{dx^2} = -\frac{d^2t}{dx^2}$ $\Rightarrow KA \frac{d^2\theta}{dx^2} + \left(\frac{dA}{dx}\right) K \frac{d\theta}{dx} = 0.$

dividing the equation by
$$KA:-$$

$$\frac{d^2\theta}{dx^2} + \frac{(d^2dx)}{A} \frac{d\theta}{dx} = 0$$

$$\frac{d^2\theta}{dx^2} + \frac{A_0 a e^{ax}}{A_0 e^{ax}} \frac{d\theta}{dx} = 0$$

$$\frac{d^2\theta}{dx^2} + a \frac{d\theta}{dx} = 0$$

$$\frac{d^2\theta}{dx} + a \frac{d\theta}{dx} =$$

$$\Rightarrow \theta = \frac{\theta_{L}(e^{-ax}_{-1})}{(e^{-aL}_{-1})}$$

$$(e^{-aL}_{-1})$$

$$\Rightarrow T_{0} - T = (T_{0} - T_{0})(e^{-ax}_{-1})$$

$$(e^{-aL}_{-1})$$

$$(e^{-aL}_{-1})$$

$$hence proved.$$
(i) Assumption:— Steady state

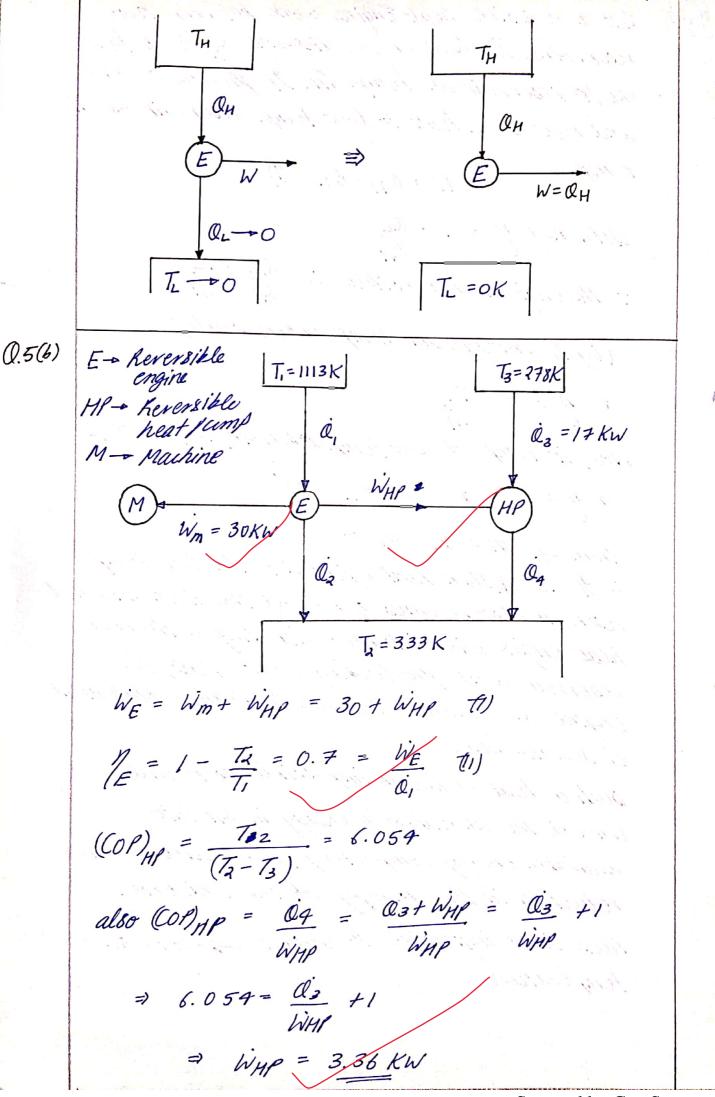
Now heat generation is taking place in the rod

as per the following given equi-
as per the following giv

Integrating the ear," we get :-It is given that at x=0 the rod is insulated $(4x)_{x=0} = 0 \Rightarrow 0 = -KA\frac{dT}{dx}_{x=0}$ $\Rightarrow \left(\frac{dT}{dx}\right)_{x=0} = 0 \qquad \& (T)_{x=0} = T_0$ Using this boundary cond? in eq. (i) :- $0 = -a \tau_0 + \frac{q_0}{q_K} + C$ $\Rightarrow C = a To \int \frac{q_0}{n\kappa}$ using this in eq. (1):- $\frac{dT}{dx} = -aT + \frac{q_0}{ak} \left(e^{-ax} \right) + aT_0 - \frac{q_0}{ak}$ $\Rightarrow \frac{dT}{dx} = a(\theta) + \frac{q_0}{q_0} \left(e^{-ax} - 1\right) (0)$ from fourier law for conduction:refer $Q_{x} = -KA \frac{dT}{dx}$ $q_{\chi} = -KA_{o}e^{a\chi}\left(a\theta\right) + \frac{\dot{q}_{o}}{n\mu}\left(e^{-a\chi}-1\right)$ =) \(\gamma_{\chi} = -akA_0e^{ax}\) + \(\frac{a}{6}A_0\) (e^{ax}-1) The above ear. is the desired ear for ax

Let a reversible heat Engine work between two Q.5(a) reservoirs. The heat engine absorbs Oy heat from the the reservoir at temp. TH. It produces work w and rejects Qu heat to low temp. reservoir at The temp. W= QH-QL (1) efficiency $\gamma = 1 - \frac{Q_L}{D_{\perp}}$: the engine is reversible : QL = TL (by Murmodynamic temperature scale) = 1 = 1 - Th Now, if temp. of low temp. reservoir, i.e, The =OK = 1 = 1 = W= QH (11) from eq. 6/4 (1) QL =0. : If The ox then heat rejected to the low temp. reservoir also becomes zero and the guess assumed heat engine will interact with single thermal reservoir while producing work. Thus, the engine will violate the kelvin Planck statement of second law. Such a heat engine is impossible as it will Convert the unavarlable energy in Qu into available energy completely, which will lead to reduction in the entropy of the universe. Thus it is impossible to achieve absolute zero temperature. for more

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: from eq. (1) WE = 33.36 KW using value of vie in ear (11) Q1 = 47.66 KW Q = Q, - WE = 14.297KW Q4 = Q3 + Wyp = 20.363 KW (1) Rate of heat supply from 840°C source = Q, a, = 97.66 KW (ii) hate of heat rejection to sink at 60°C = 02+04 Q2 + Q4 = 34.66 KW Q.5(6) (1) P2=79, P1=P3 V3=7V, V $V_1 = V_2$ fig. T-8 diagram fig. P-V diagram (i) Assumption: - Air is an ideal gas $V_1 = \frac{mRT_1}{n} = 0.861 \, \text{m}^2 = V_2$ m = 1KgV3 = 7V, = 6.027 m3 P = 100 KPa

T1 = 300K

for poless
$$1-2$$
 $V = C$ $\frac{P_{2}}{P_{1}} = \frac{T_{2}}{T_{1}}$
 $\Rightarrow T_{A} = 2100 \text{ K}$

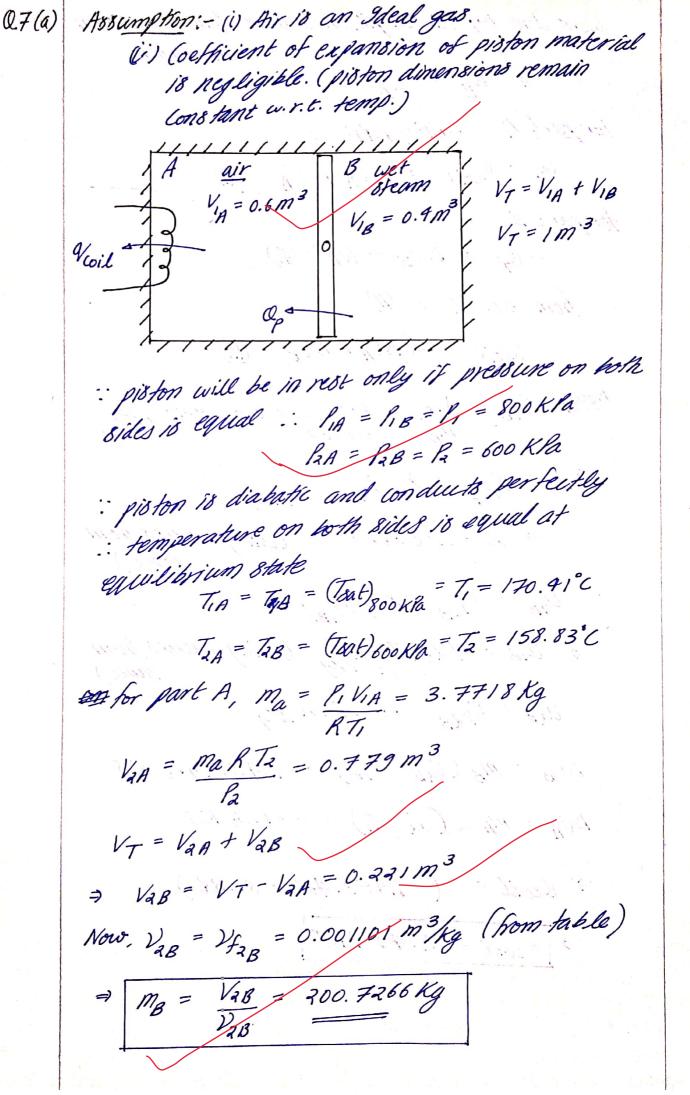
Proless $1-2$ $W_{12} = 0$ $(: V = C)$
 $Q_{12} = mC_{V}(T_{2}-T_{1}) = +1292.4 \text{ KJ}$

Process $2+3$ $Q_{23} = Q_{23} = (\Delta Q_{3} + W_{23})$
 $(\Delta U)_{2-3} = 0$ $(:: T = C)$
 $\Rightarrow Q_{2-3} = W_{2-3} = mI_{A}V_{2} \ln(\frac{V_{3}}{V_{A}})$
 $\Rightarrow Q_{3-3} = W_{2-3} = +1172.80 \text{ KJ}$

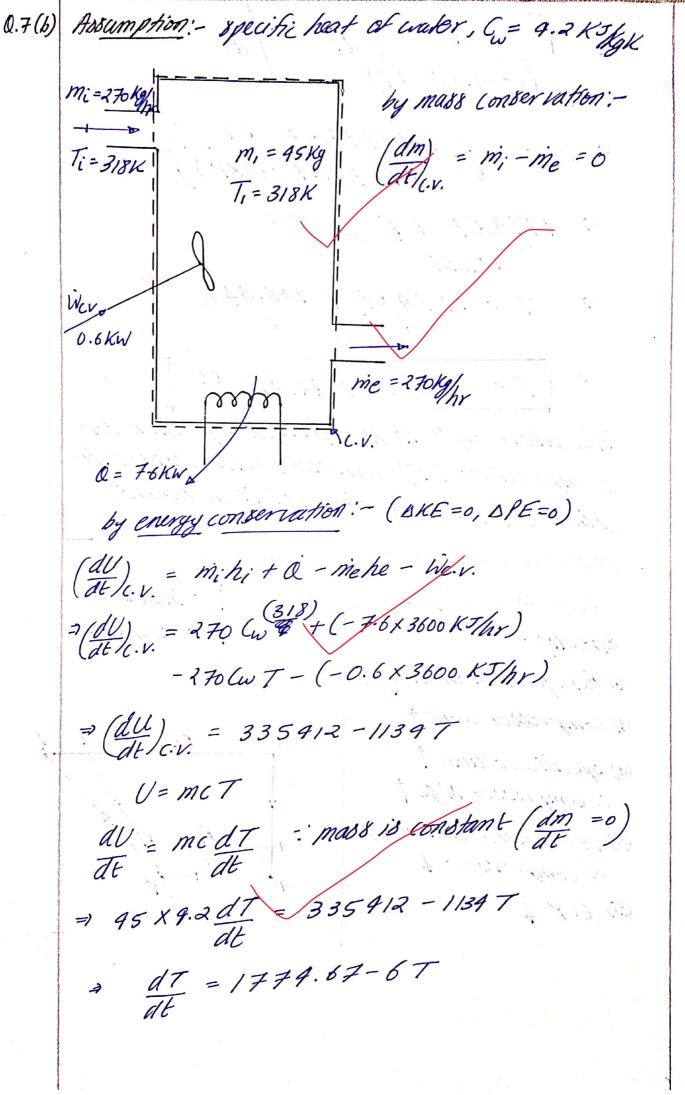
Process $3+1$ $Q_{3-2} = Q_{3-2} = +1172.80 \text{ KJ}$
 $W_{3-1} = P_{1}(V_{1}-V_{3}) = -516.6 \text{ KJ}$
 $W_{12} = W_{12} + W_{2-2} + W_{2-1}$
 $\Rightarrow W_{13} = G_{12} + G_{2-3}$
 $\Rightarrow W_{14} = G_{12} + G_{2-3}$
 $\Rightarrow Q_{1} = 2965.2 \text{ KJ}$
 $Q_{1} = W_{1} = G_{1} = G_{1} = G_{2} = G$

specified capacity = 160 TR 0.5)(d) Cooling water Flow rate =/20 like : mwater = 20 Kg/8 (Tin)w = 25°C (Tout) = 35°C Imotor = 300 KW ((mech = 92%) Assumption - There is no heat loss to the surrounding in the condenser. :. heat lost by working fluid = heat gained by water (w = 4.18 KJ/kgK ac = mw Cw (Tout - Tin) => Qc = 836 KW Work of compressor (Wc) mech = Power of motor (P) 1 We = mech motor = 276KW Qc = Wet QE where QE = heat absorbed in en elapourator or refrigeration capacity. QE = Qc - WC = 560 KW = 160 TR The actual refrigerating capacity of the given ar conditioning system is as per specifications. RC = 160TR

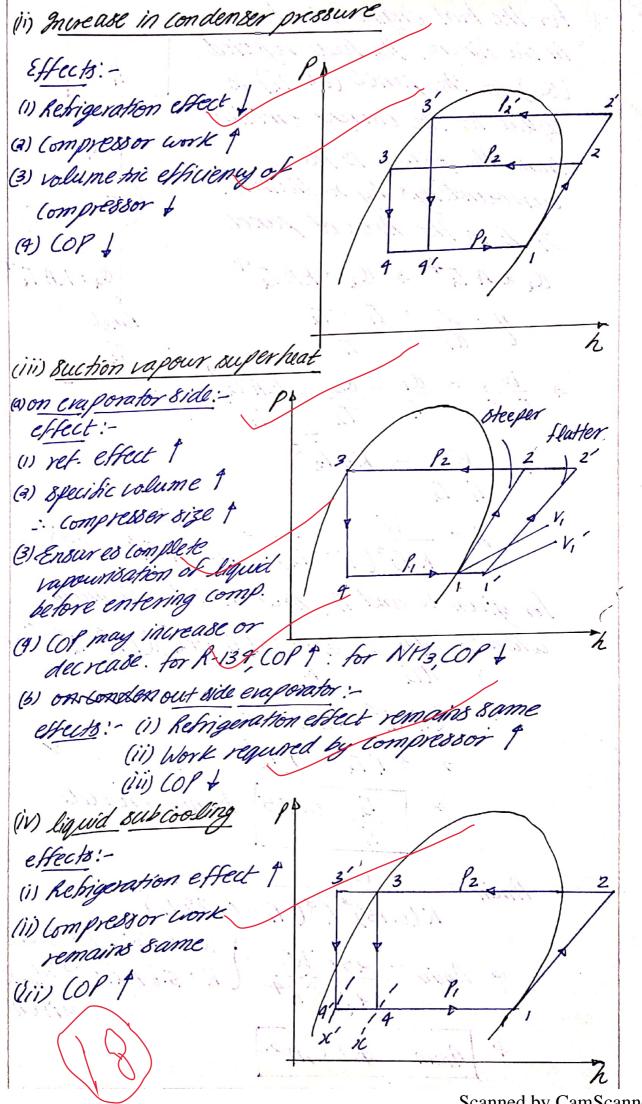
In the condensor of a refrigeration system, heat Q.5(e) of the reprigerent is rejected. The reprigerent vapour at discharge of compressor is superheated. De superheating of the vapour takes place in hist few wills of condenser. Then the saturated vapour is converted to saturated liquid/by rejecting latent heat of condensation. In some condensers out too ling may also take place. The liquid refrigerant thus obtained can be used in evapourator to absorb heat. * The type of condensers based on cooling medium: (i) Air wooled condenses (ii) Water world condenser (ii) Evapourative condenser. Air out (humid) Eva pourative condenser fan compartment The refrigerant first Eliminator rejects its heat to A A A A water and then water rejects its heat to air, Retrigerant mainly in the form of evapourated water. Airin Air leaves with high humidity as in a Make-up Cooling tower. Water Thus an evaporative tig. Evaporative condenser combines the Condenser functions of condenser and cooling lower. Evaporative condensers are commonly used in large ammonia plants.



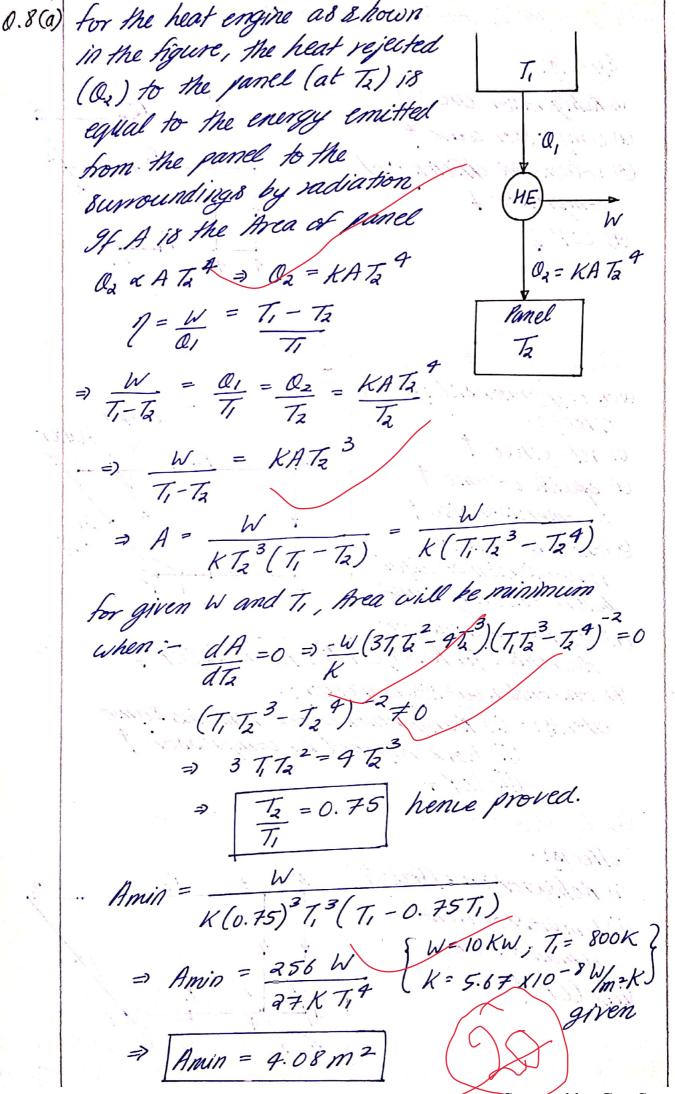
work done by air on piston = work done by piston = WB = - WA for part A DKE= DIE =0 Op - Qcoil = DUA + WA -(1) for part B - Op = DUB - WA TOD from eq. (1) A (1) Ocoil = - (DUA+ DUB) (III) N_{000} , $V_{18} = \frac{V_{18}}{m_R} = \frac{0.4}{200.73} = 0.0019928 \, m_{kg}^3$ VIB = VIB + X (VBIB - VIB) $\Rightarrow \chi = 0.003669$ (using values from : UB = UIB + X (Ufg, 2) => U1B = 726.78 KJ/g (using values from table) Ung = UfaB = 669.72 KJ/Kg DUB = MB (UAB - UB) = - 11953. 466 KJ DUA = MACV (Ta-Ti) = -31.36 KJ -> Ocoil = - (-11453.466 - 8136) => Qcoil = 11.48 MJ



 $\int \frac{1}{(1774.67-67)} dT = \int dt$ $\Rightarrow t = -\frac{1}{6} \ln \left(\frac{1774.67 - 67}{1774.67 - 6(318)} \right)$ 1774.67-6T = e-6t -133.33 T = 1774.67 + 133.33 e -66 T = 295.77 + 22.22e-6t The above eq! is the desired equation for temperature (T) variation with respect to time (t), where time (t) is in hours. Q.7 (c). factors affecting performance of VCRS cycle:-1 - increases (i) Pecre ase in evaporator pressure - decreases. Effects:-(1) Resigeration estell | (3) Compressor work + (3) specific volume : Compressor size 1 (4) volume tric estiliency of Compressor & (5) (OP 1.



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Applying first law of Mermodynamics on the rod:
$$Q = \Delta U + W$$
 $Q = 0$ (insulated $W = 0$
 $Q = 0$ ($T_3 - T_1$) $Q = 0$
 $Q = 0$ ($T_4 - T_2$) $Q = 0$

$$\Rightarrow T_3 = T_1 + T_2$$

Entropy any analysis:

$$\Delta Suniv = \Delta Syod + \Delta Shurr.$$

$$\Rightarrow \Delta Suniv = \Delta Syod.$$
for rod:- $\Delta S = dmc lm(\frac{T_S}{T_K})$

$$dm = \beta A dx \qquad L$$

$$\Rightarrow \Delta S = \int dS = \int \beta A c ln(\frac{T_S}{T_K}) dx$$

$$\Rightarrow \Delta S = \int A C ln T_S - \int \beta A c (ln T_K) dx$$

$$\Rightarrow \Delta S = \int A c l ln T_S - \int \beta A c (ln T_S) dx$$

$$Tx = T_I - (T_I - T_S) x \Rightarrow dTx = -(T_I - T_S) dx$$

$$\Rightarrow dx = \frac{L}{(T_I - T_S)} dTx$$

$$\Rightarrow \Delta S = \beta A c l ln T_S - \int \beta A c (ln T_K) (\frac{L}{T_I - T_S}) dTx$$

$$\Rightarrow \Delta S = \beta A c l ln T_S - \int \beta A c (ln T_K) (\frac{L}{T_I - T_S}) dTx$$

$$\Rightarrow \Delta S = \beta A c l ln T_S - (\beta A c l) [T_K ln T_K - T_K] T_I$$

$$\Rightarrow \Delta S = m c ln T_S - (m c) [T_I ln T_I - T_L ln T_S - (T_I - T_S)]$$

$$\Rightarrow \Delta S = m c ln T_S - m c [(T_I ln T_I) - (T_I ln T_I) - (T_I ln T_I)]$$

$$\Rightarrow \Delta S = \Delta S m c ln T_S - m c [(T_I ln T_I) - (T_I ln T_I) - (T_I ln T_I)]$$

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$$\Rightarrow \Delta S = \Delta S m c ln T_S - m c [(T_I ln T_I) - T_I ln T_I]$$

for the
$$R-R$$
 yell $1-2-3-4$
 $R_1 = (R_1)$
 $R_2 = 1+8.73$ K5/kg

 $R_3 = R_4 = (R_1)$
 $R_3 = R_4 = (R_1)$
 $R_3 = R_4 = (R_2)$
 $R_3 = R_4 = (R_3)$
 $R_4 = R_2$
 $R_5 = R_5$
 $R_$