

*Saket  
Centrxx*



**MADE EASY**

India's Best Institute for IES, GATE & PSUs

## ESE 2024 : Mains Test Series

UPSC ENGINEERING SERVICES EXAMINATION

### Mechanical Engineering

**Test-3 : Section A:** Fluid Mechanics and Turbo Machinery [All Topics]

**Section B :** Heat Transfer-1 + Refrigeration and Air-Conditioning-1 [Part Syllabus]

Thermodynamics-2 + Strength of Materials & Mechanics-2 [Part Syllabus]

Name :

Roll No

| Test Centres                              |                                  |                                    | Student's Signature |
|---|----------------------------------|------------------------------------|---------------------|
| Delhi <input checked="" type="checkbox"/> | Bhopal <input type="checkbox"/>  | Jaipur <input type="checkbox"/>    |                     |
| Pune <input type="checkbox"/>             | Kolkata <input type="checkbox"/> | Hyderabad <input type="checkbox"/> |                     |

#### Instructions for Candidates

- Do furnish the appropriate details in the answer sheet (viz. Name & Roll No).
- There are Eight questions divided in TWO sections.
- Candidate has to attempt FIVE questions in all in English only.
- Question no. 1 and 5 are compulsory and out of the remaining THREE are to be attempted choosing at least ONE question from each section.
- Use only black/blue pen.
- The space limit for every part of the question is specified in this Question Cum Answer Booklet. Candidate should write the answer in the space provided.
- Any page or portion of the page left blank in the Question Cum Answer Booklet must be clearly struck off.
- There are few rough work sheets at the end of this booklet. Strike off these pages after completion of the examination.

#### FOR OFFICE USE

| Question No. | Marks Obtained |
|--------------|----------------|
|--------------|----------------|

Section-A

|     |    |
|-----|----|
| Q.1 | 13 |
| Q.2 | 33 |
| Q.3 | 16 |
| Q.4 | —  |

Section-B

|     |    |
|-----|----|
| Q.5 | 34 |
| Q.6 | —  |
| Q.7 | 37 |
| Q.8 | —  |

Total Marks  
Obtained

133

Signature of Evaluator

Haseen

Cross Checked by

## IMPORTANT INSTRUCTIONS

CANDIDATES SHOULD READ THE UNDERMENTIONED INSTRUCTIONS CAREFULLY. VIOLATION OF ANY OF THE INSTRUCTIONS MAY LEAD TO PENALTY.

### DONT'S

1. Do not write your name or registration number anywhere inside this Question-cum-Answer Booklet (QCAB).
2. Do not write anything other than the actual answers to the questions anywhere inside your QCAB.
3. Do not tear off any leaves from your QCAB, if you find any page missing do not fail to notify the supervisor/invigilator.
4. Do not leave behind your QCAB on your table unattended, it should be handed over to the invigilator after conclusion of the exam.

### DO'S

1. Read the Instructions on the cover page and strictly follow them.
2. Write your registration number and other particulars, in the space provided on the cover of QCAB.
3. Write legibly and neatly.
4. For rough notes or calculation, the last two blank pages of this booklet should be used. The rough notes should be crossed through afterwards.
5. If you wish to cancel any work, draw your pen through it or write "Cancelled" across it, otherwise it may be evaluated.
6. Handover your QCAB personally to the invigilator before leaving the examination hall.

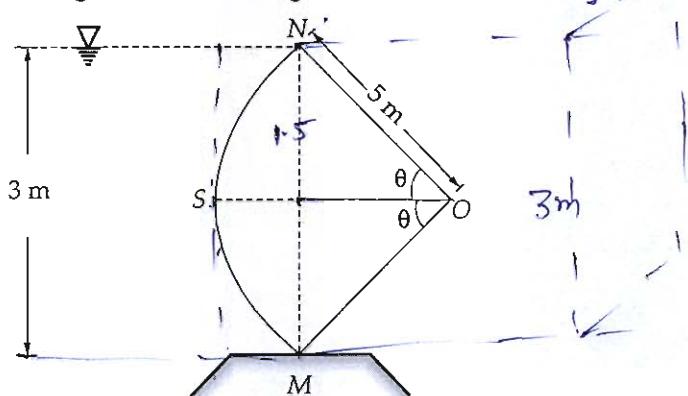
### # Comments :

- Representation is good.
- Improve accuracy
- Many compulsory questions not attempted
- Solve more such questions and improve speed and accuracy.

## Section : A

- 2.1 (a) A sector gate in the form of circular arc of radius 5 m retains water to a height of 3 m above its sill as shown in figure. Calculate the magnitude and direction of the resultant force per unit length of the gate. Assume a gate width of 1 m.

(2)



We Know

[12 marks]

$$F_H = \rho g A \bar{x} = 9810 \times 3 \times 1 \times 1.5 = 44.145 \text{ KN}$$

$F_v = \rho g (\text{Volume occupied by curve upto free surface})$

$$\sin \theta = \frac{1.5}{5} \Rightarrow \theta = 17.4576^\circ$$

$$2\theta = 34.9152^\circ$$

$$\text{Area of NSM} = \text{Area of ONSM} - \text{Area of NMO}$$

$$= \frac{2\theta}{2\pi} \times \pi r^2 - \frac{1}{2} \times 2 \times \sqrt{5^2 - 1.5^2} \times 1.5 \\ = 7.6173 - 7.1545 = 0.4628 \text{ m}^2$$

$$F_v = 9810 \times 0.4628 \times 1 = 4.5402 \text{ KN}$$

$$F_R = \sqrt{F_H^2 + F_v^2} = \sqrt{44.145^2 + 4.5402^2} = 44.3778 \text{ KN}$$

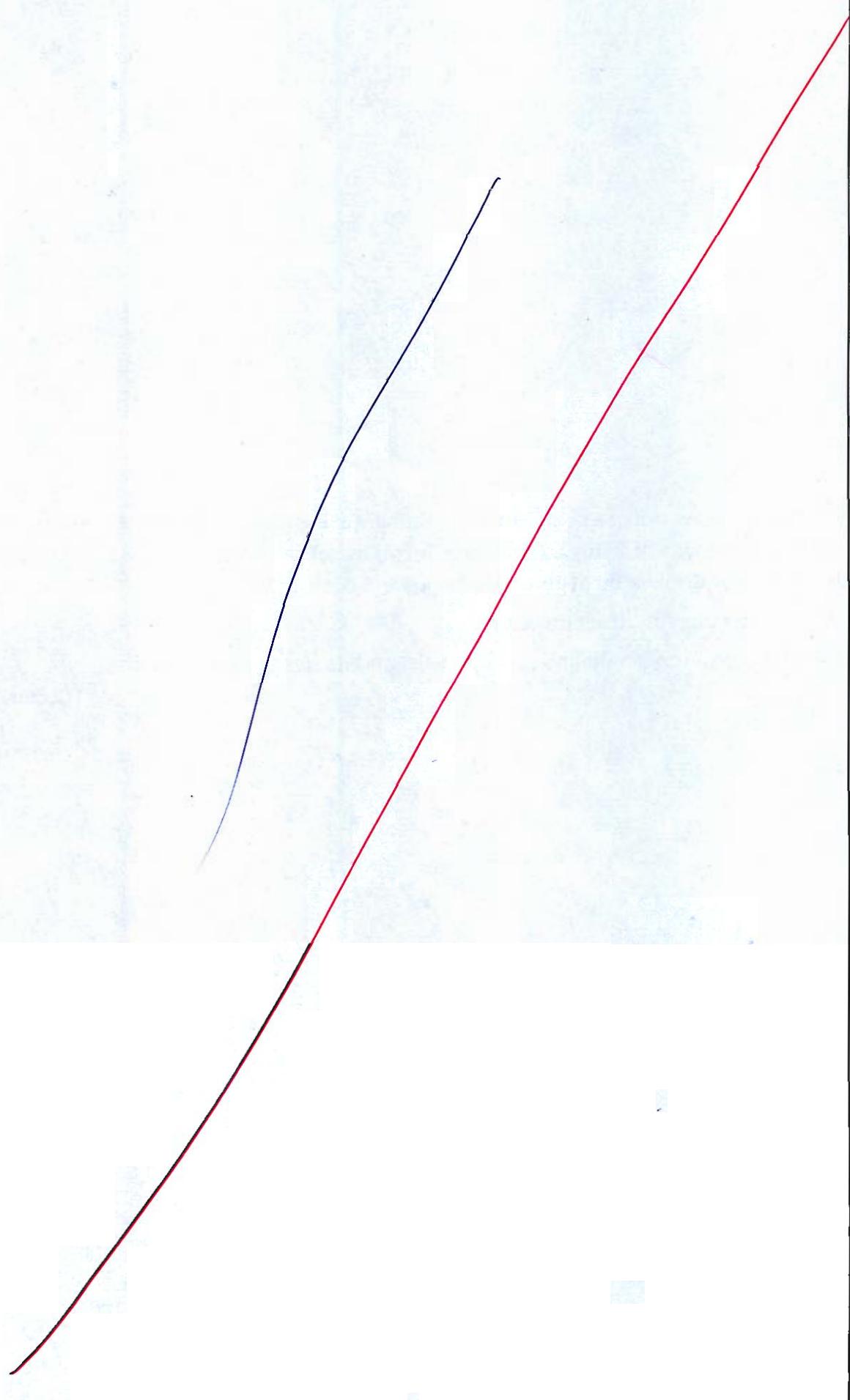
Ans

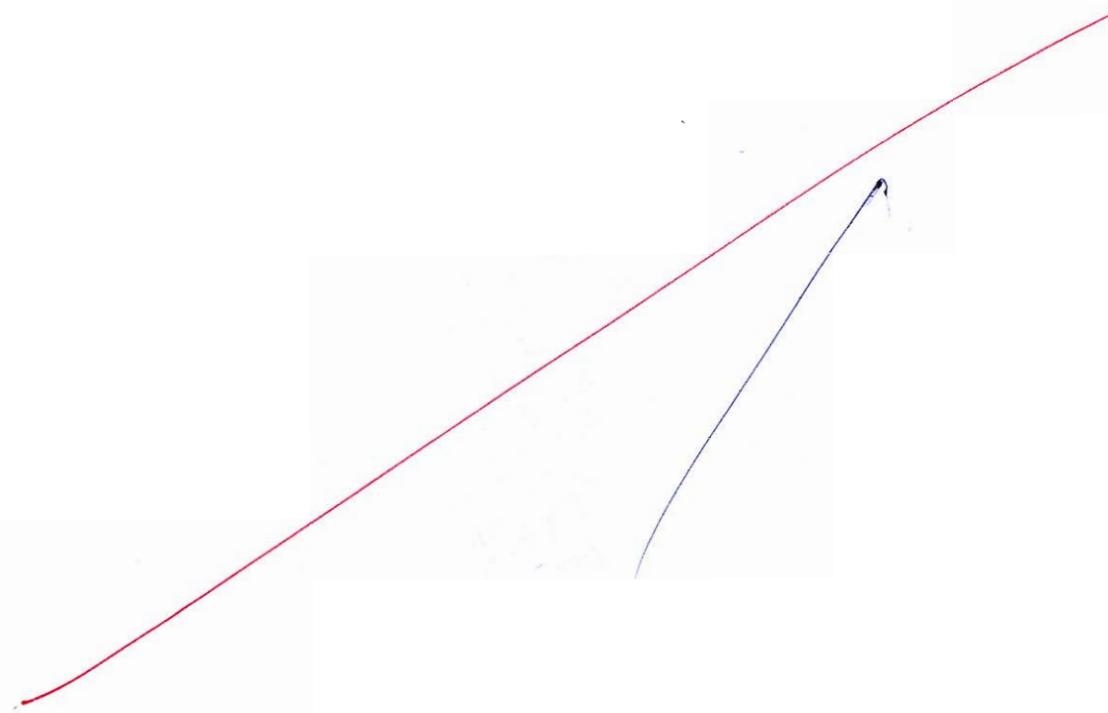
$$\tan \theta = \frac{F_v}{F_H} = \boxed{\theta = 5.872^\circ} \text{ direction.}$$

Ans

- Q.1 (b)** With the aid of a schematic diagram, explain the working principle of pulse jet engine and also draw the ideal and actual P-V diagrams.

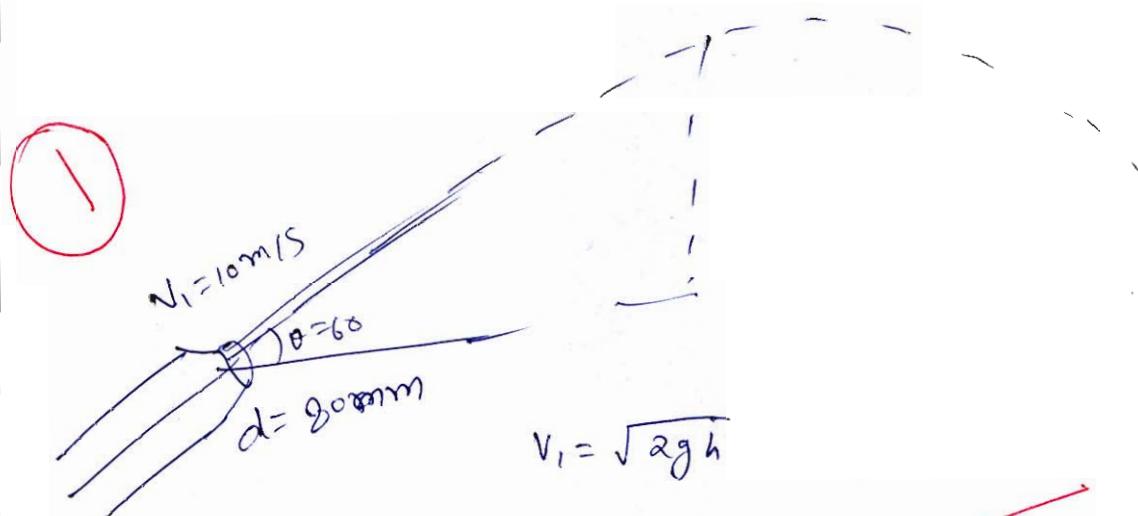
[12 marks]

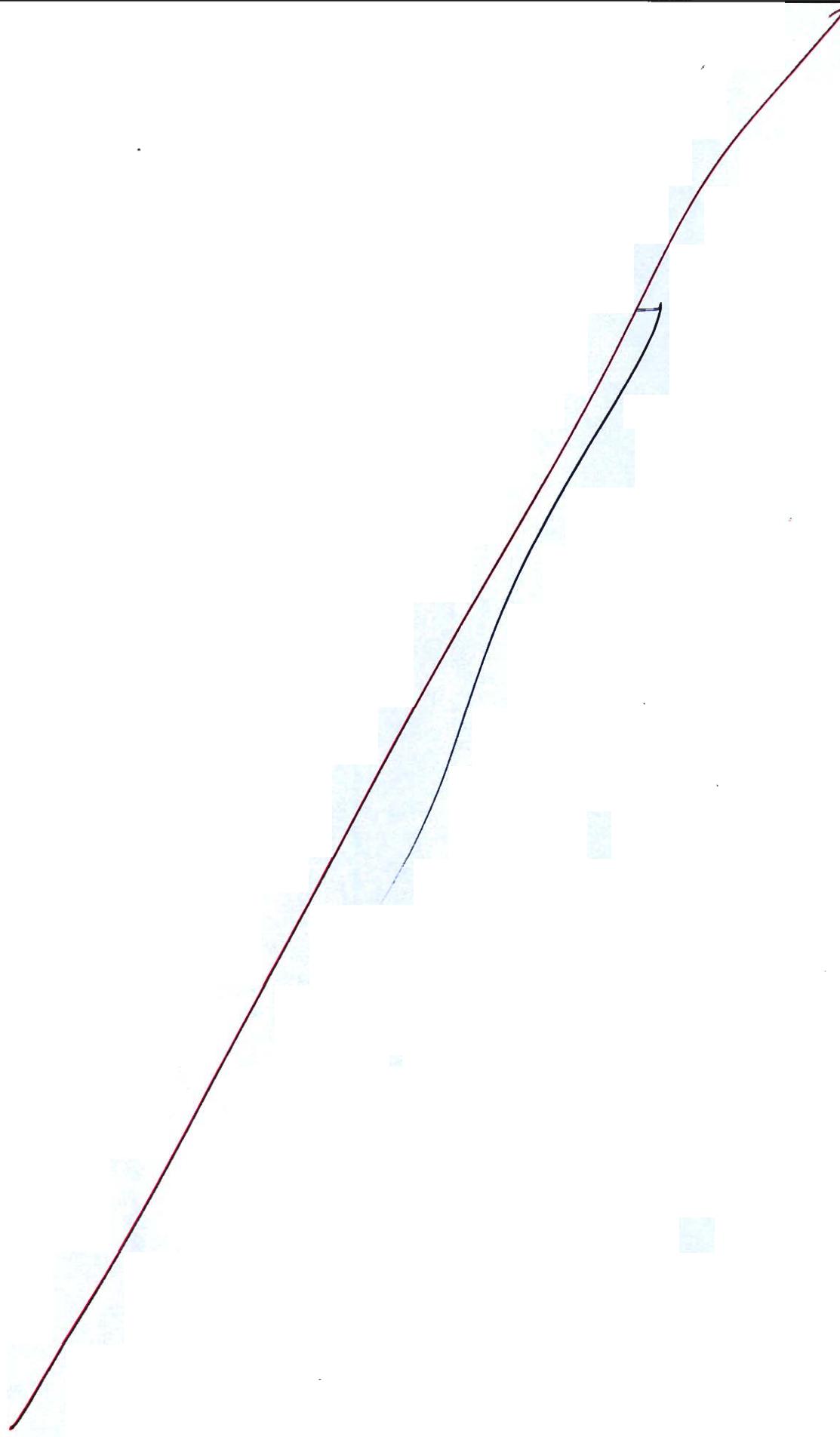




- Q.1 (c) A liquid jet issues out of a nozzle into atmosphere at an angle of  $60^\circ$  above the horizontal and with a velocity of 10 m/s. At the nozzle exit the jet has a diameter of 8 cm. Assuming the jet to be unbroken throughout the trajectory, determine
- the equation of the jet trajectory.
  - the maximum height attained by the jet and its size at that location.

[12 marks]

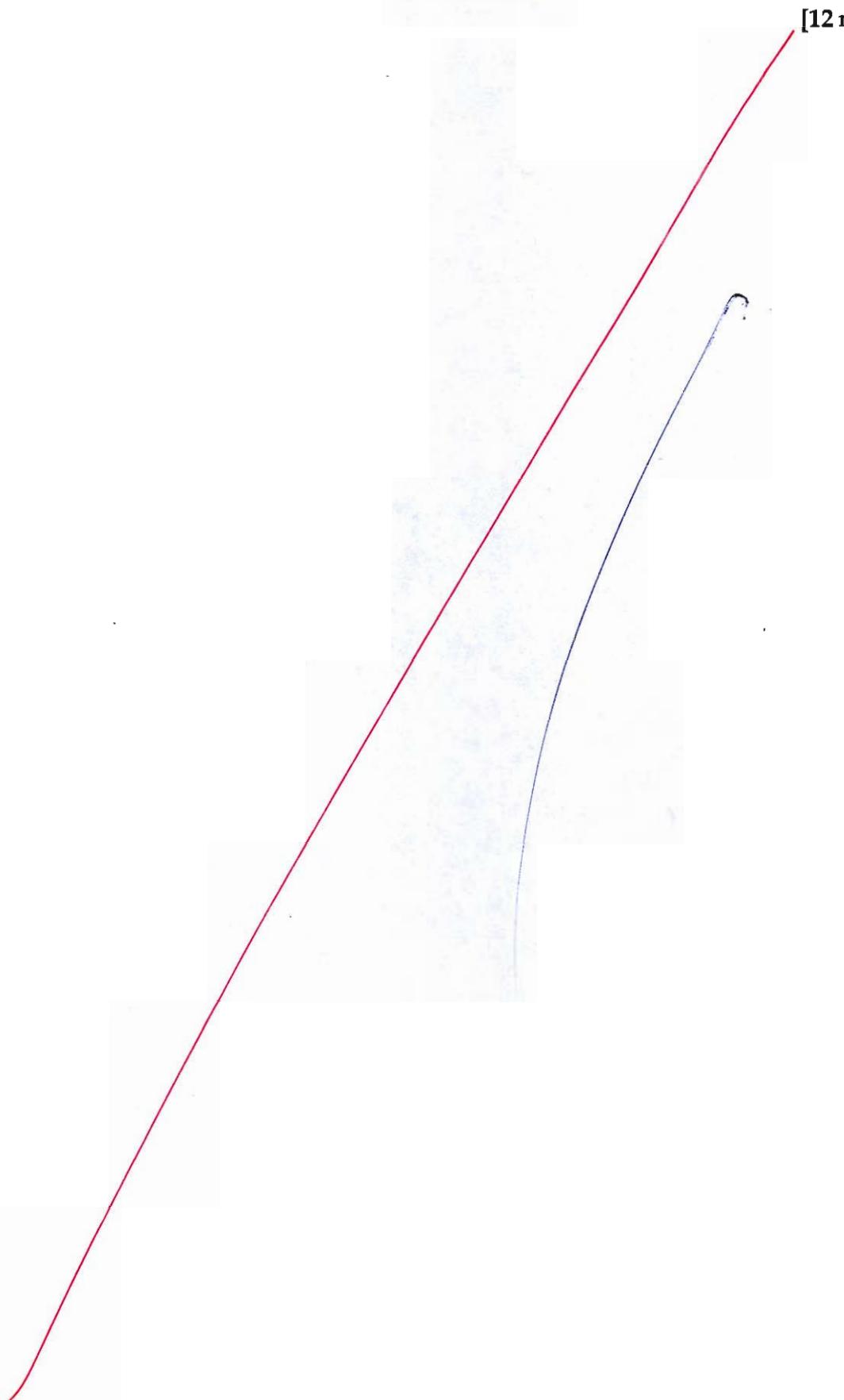




Q.1 (d)

A single acting reciprocating pump has its piston executing a simple harmonic motion. Show that the ratio of the work done against friction when the air vessels are fitted to that in the absence of air vessels is  $3/2\pi^2$ .

[12 marks]





Q.1 (e)

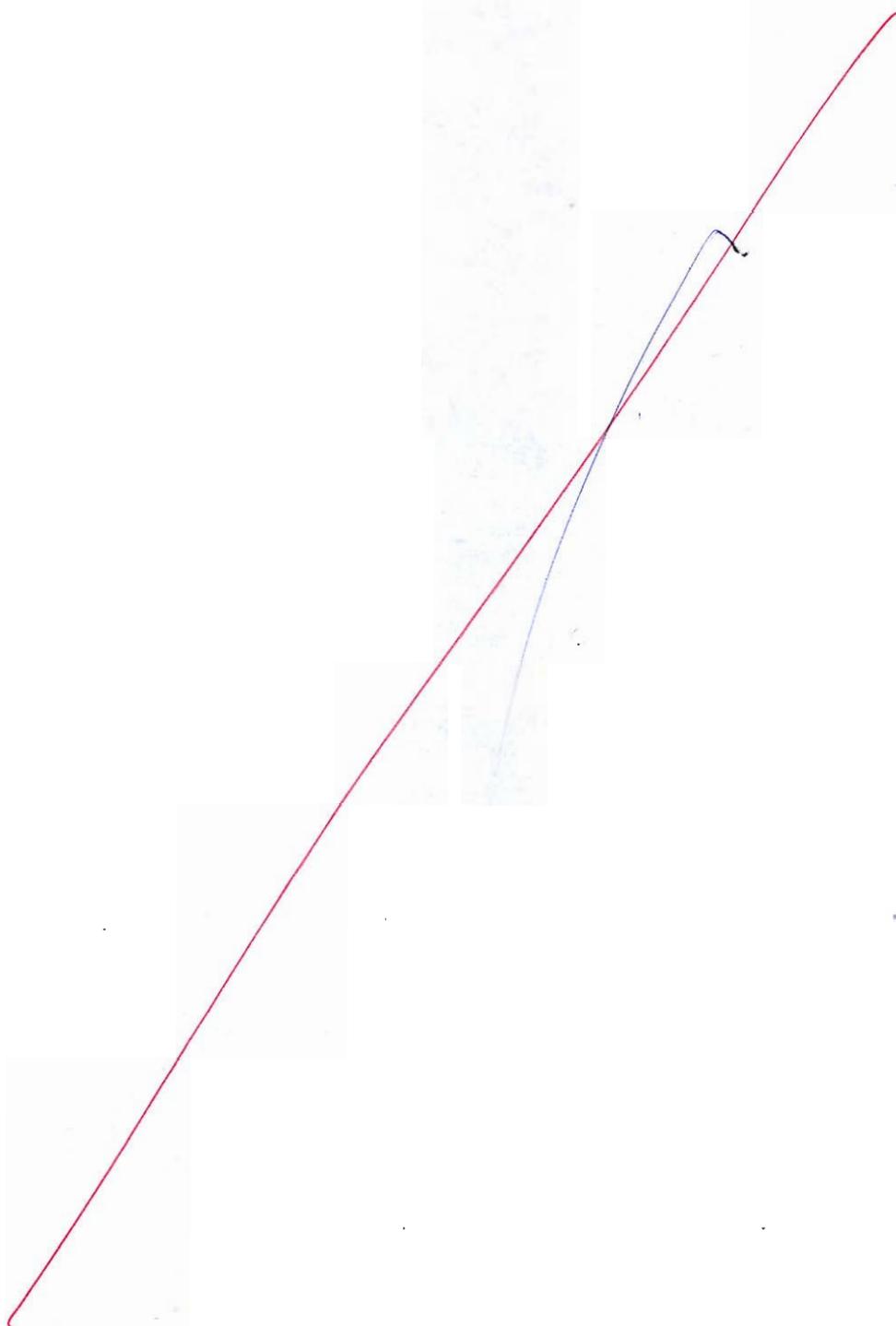
Examine whether or not the following velocity profiles satisfy the essential boundary conditions for velocity distribution in laminar boundary layer on a flat plate.

(i)  $\frac{U}{U_0} = 1 + \left(\frac{y}{\delta}\right) - 2\left(\frac{y}{\delta}\right)^2$

(ii)  $\frac{U}{U_0} = \sin\left(\frac{\pi y}{2\delta}\right)$

where  $u$  is the velocity at height  $y$  above the surface,  $U_0$  is the free stream velocity and  $\delta$  is the nominal boundary layer thickness.

[12 marks]



Q.2 (a) A Pelton wheel has 4 nozzles each 52 mm in diameter with coefficient of velocity 0.98. Bucket speed is 0.46 times the jet speed and bucket mean diameter is 0.85 m. Reduction in relative velocity due to bucket friction is 16% with jet deflection of  $165^\circ$  and having mechanical efficiency of 94%. The water is supplied through a pipeline 360 m long from a reservoir, the level of which is 300 m above the nozzles. If the friction coefficient is 0.0058 and the head lost in friction amounts to 32 m. Calculate :

- the diameter of the pipeline,
- the bucket power, hydraulic and overall efficiencies of the wheel,
- the unit speed, the unit power and specific speed.

$$\text{nozzles} = 4 \quad C_v = 0.98 \quad D = 0.85 \text{ m} \quad [20 \text{ marks}]$$

$$d = 52 \text{ mm} \quad u = 0.46 V_1 \quad S = 165^\circ, K = 0.84$$

$$\eta_m = 0.94, L = 360 \text{ m}, H_{in} = 300 \text{ m}, f' = 0.0058$$

$$h_f = 32 \text{ m}$$

$$(i) h_f = \frac{4 f' L V_p^2}{2 g D_p} \quad H = 300 - 32 = 268 \text{ m}$$

$$\eta_m = \frac{SP}{RP}$$

$$V_1 = C_v \sqrt{2gH} = 0.98 \sqrt{2 \times 9.81 \times 268}$$

$$V_1 = 71.0629 \text{ m/s}$$

$$32 = \frac{4 \times 0.0058 \times 360 \times \cancel{(1 + 0.16)^2} \cdot 19215}{2 \times 9.81 \times D_p \times D_p^2}$$

Apply continuity at Pipe & Nozzle

~~$$\frac{\cancel{4}}{4} \times (0.052)^2 \times 71.0629 = \frac{1}{4} D_p^2 \times V_p$$~~

$$V_p = \frac{19215}{D_p^2}$$

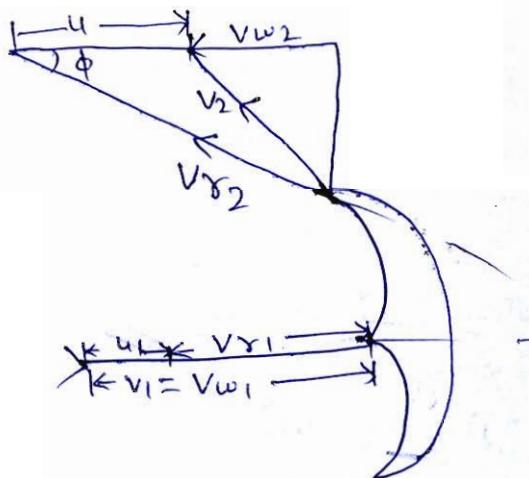
$$D_p = 0.1367 \text{ m}$$

$$\text{or } 136.73 \text{ mm}$$

Refer Solution

Ans

~~$$(ii) u = 0.46 \times 71.0629 = 32.6889 \text{ m/s}$$~~



$$\phi = 15^\circ$$

$$v_{w1} = 71.0629 \text{ m/s}$$

$$v_{r1} = v_1 - u_1$$

$$= 38.3739 \text{ m/s}$$

$$v_{r2} = kv_{r1} = 0.84 \times 38.3739 = 32.2341 \text{ m/s}$$

$$\cos \phi = \frac{u + v_{w2}}{v_{r2}} \Rightarrow \cos 15^\circ = \frac{32.2341 + v_{w2}}{32.2341}$$

$$v_{w2} = -1.5531 \text{ m/s, } v_{w2} \text{ in opposite direction}$$

$$RP = \rho Q (v_{w1} + v_{w2}) u$$

$$Q = \frac{\pi}{4} (0.05^2)^2 \times 71.0629 = 0.1509 \text{ m}^3/\text{s}$$

$$\text{Discharge from 4 nozzles} = 4Q = 0.6036 \text{ m}^3/\text{s}$$

$$RP = 1000 \times 0.6036 (71.0629 - 1.5531) \times 32.2341$$

$$RP = 1371.657 \text{ kW} \quad \boxed{\text{Ans.}}$$

$$WP = \rho g Q H = 9.81 \times 1000 \times 0.6036 \times 268 \\ = \boxed{1586.912 \text{ kW}}$$

$$\gamma_{\text{hydraulic}} = \frac{RP}{WP} = \frac{1371.657}{1586.912} \times 100 = \boxed{86.43\%} \quad \boxed{\text{Ans.}}$$

$$\gamma_{\text{overall}} = \gamma_h \times \gamma_m = 0.8643 \times 0.94 = \boxed{81.24\%} \quad \boxed{\text{Ans.}}$$

$$(iii) \text{ Unit speed} \Rightarrow \text{Nu} = \frac{N}{\sqrt{H}}$$

$$u = \frac{\pi D N}{60} \Rightarrow 32.6889 = \frac{\pi \times 85 \times N}{60}$$

$$N = 734.4847 \text{ rpm}$$

$$\text{Nu} = \frac{734.4847}{\sqrt{268}} = \underline{\underline{44.865}} \text{ /} \text{m minute}$$

$$(iii) \text{ Unit Power, } P_u = \frac{P}{H^{3/2}}$$

*Bucket power × η₀*

$$= \frac{1371.657}{(268)^{1.5}} = \underline{\underline{0.3126}}$$

$$(iii) \text{ Specific Speed - } N_s = \frac{N \sqrt{P}}{H^{5/4}}$$

$$= \frac{734.4847 \times \sqrt{1371.657}}{268^{1.25}} = \underline{\underline{25.086}} \text{ rpm}$$

- Q.2 (b) (i) Explain the phenomenon of boundary layer separation. Describe four methods of controlling of boundary layer separation.
- (ii) Two pipes each of length  $L$  and diameters  $D_1$  and  $D_2$  are arranged in parallel. The loss of head when a total quantity of water  $Q$  flows through them being  $h_1$ . If the pipes are arranged in series the same quantity of water ' $Q$ ' flows through them, the loss of head is  $h_2$ . If  $D_1 = 1.5D_2$ , determine the ratio of  $h_1$  to  $h_2$ . Neglect minor losses and assume the friction factor  $f$  to be constant and to have the same value of for both the pipes.

[8 + 12 marks]

6

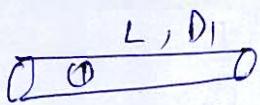
(i). whenever any fluid moving in converging section area is decrease so velocity increases & pressure decreases, the pressure gradient is negative ( $-\frac{\partial P}{\partial n}$ ). It is called favourable pressure gradient.

• When this fluid come in diverging section, the pressure starts increasing & pressure gradient is adverse. ( $+\frac{\partial P}{\partial n}$ ).

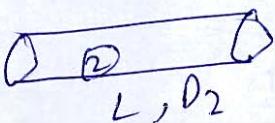
• The fluid near the boundary not have capability to flow due to this adverse pressure gradient, fluid will stop near the boundary & reverse its direction, this whole phenomenon is called boundary layer separation.

### Methods to Control the Boundary Layer Separation

- Accelerate the fluid particles near the boundary.
- Divergence angle keep minimum as 5 to 7°.
- Inject new accelerate fluid particles.
- Eject deaccelerate fluid particles.
- Move the boundary in the direction of flow.



$$Q_1 + Q_2 = Q, h_1$$



$$D_1 = 1.5 D_2 \quad \checkmark$$

$$\frac{h_1}{h_2} = ?$$

parallel - loss - h<sub>1</sub>, series - h<sub>2</sub> discharge is same.

$$h_1 = \frac{f L Q_1^2}{12.1 D_1^5} = \frac{f L Q_2^2}{12.1 D_2^5}$$

$$h_2 = \frac{f L Q_1^2}{12.1 D_1^5} + \frac{f L Q_2^2}{12.1 D_2^5} = \frac{f L (Q_1^2 + Q_2^2)}{12.1}$$

$$h_2 = \frac{f L Q_1^2}{91.884375 D_2^5} + \frac{f L Q_2^2}{12.1 D_2^5} = \frac{f L (Q_1^2 + 7.59375 Q_2^2)}{91.88437 D_2^5}$$

$$\frac{h_1}{h_2} = \frac{f L Q_2^2}{12.1 D_2^5}$$

$$\frac{f L (Q_1^2 + 7.59375 Q_2^2)}{91.88437 D_2^5}$$

$$\frac{91.88437 Q_2^2}{12.1 (Q_1^2 + 7.59375 Q_2^2)}$$

$$\frac{7.59375 Q_2^2}{Q_1^2 + 7.59375 Q_2^2}$$

$$h_1 = \frac{f L Q_2^2}{12.1 D_1^5}, h_2 = \frac{f L Q_2^2}{12.1 D_1^5} + \frac{f L Q_2^2}{12.5 D_2^2}$$

*Refer Solution*

$$\frac{h_1}{h_2} = \frac{\cancel{fLQ^2}}{\cancel{12 \cdot 1 D_{1.5}}} = \frac{1}{D_{1.5}}$$
$$\frac{\cancel{fLQ^2}}{\cancel{12 \cdot 1 D_{1.5}}} + \frac{\cancel{fLQ^2}}{\cancel{12 \cdot 1 D_{2.5}}} = \frac{1}{D_{1.5}} + \frac{1}{D_{2.5}}$$

$$\frac{\frac{1}{(1.5)^5}}{\frac{1}{(1.5)^5} + 1} = 0.1163$$

Ans

Refer solution

- Q.2 (c)** A nozzle of the impulse stage of a turbine receive steam at 20 bar and 300°C and discharge it at 12 bar and 240°C. The efficiency of nozzle is 96% and nozzle angle is 18°. The blade speed is that required for maximum work, and the inlet angle of the blades is that required for entry of the steam without shock. The blade exit angle is 6° less than the inlet blade angle. The blade friction factor is 0.88. For a steam flow rate of 1440 kg/hr, calculate (a) Axial thrust, (b) Diagram power, and (c) Diagram efficiency. [Refer Steam table attached]

Water/Steam at  $p = 1.2 \text{ MPa}$  ( $T_{\text{sat}} = 187.957^\circ\text{C}$ )

| $T$     | $v$                    | $u$    | $h$    | $s$      | $T$  | $v$                    | $u$    | $h$    | $s$     |
|---------|------------------------|--------|--------|----------|------|------------------------|--------|--------|---------|
| °C      | $\text{m}^3/\text{kg}$ | kJ/kg  | kJ/kg  | kJ/kg K  | °C   | $\text{m}^3/\text{kg}$ | kJ/kg  | kJ/kg  | kJ/kg K |
| 0       | 0.00099960             | -0.02  | 1.18   | -0.00008 | 270  | 0.20111                | 2739.2 | 2980.5 | 6.9155  |
| 5       | 0.00099949             | 21.01  | 22.21  | 0.07623  | 280  | 0.20540                | 2756.1 | 3002.6 | 6.9558  |
| 10      | 0.00099977             | 41.99  | 43.19  | 0.15098  | 290  | 0.20964                | 2772.9 | 3024.5 | 6.9951  |
| 15      | 0.00100039             | 62.93  | 64.13  | 0.22428  | 300  | 0.21386                | 2789.7 | 3046.3 | 7.0335  |
| 20      | 0.00100129             | 83.84  | 85.04  | 0.29623  | 310  | 0.21804                | 2806.4 | 3068.0 | 7.0710  |
| 25      | 0.00100246             | 104.74 | 105.94 | 0.36692  | 320  | 0.22220                | 2823.0 | 3089.6 | 7.1078  |
| 30      | 0.00100388             | 125.62 | 126.82 | 0.43639  | 330  | 0.22634                | 2839.6 | 3111.2 | 7.1438  |
| 35      | 0.00100551             | 146.50 | 147.71 | 0.50471  | 340  | 0.23045                | 2856.2 | 3132.7 | 7.1792  |
| 40      | 0.00100736             | 167.38 | 168.59 | 0.57194  | 350  | 0.23455                | 2872.7 | 3154.2 | 7.2139  |
| 45      | 0.00100939             | 188.27 | 189.48 | 0.63811  | 360  | 0.23863                | 2889.2 | 3175.6 | 7.2480  |
| 50      | 0.00101162             | 209.16 | 210.37 | 0.70326  | 370  | 0.24270                | 2905.8 | 3197.0 | 7.2816  |
| 55      | 0.00101402             | 230.04 | 231.26 | 0.76743  | 380  | 0.24675                | 2922.3 | 3218.4 | 7.3147  |
| 60      | 0.00101660             | 250.93 | 252.17 | 0.83067  | 390  | 0.25079                | 2939.0 | 3239.9 | 7.3472  |
| 65      | 0.00101933             | 271.87 | 273.09 | 0.89299  | 400  | 0.25482                | 2955.5 | 3261.3 | 7.3793  |
| 70      | 0.00102223             | 292.79 | 294.02 | 0.95444  | 410  | 0.25883                | 2972.1 | 3282.7 | 7.4109  |
| 75      | 0.00102529             | 313.74 | 314.97 | 1.0150   | 420  | 0.26284                | 2988.8 | 3304.2 | 7.4421  |
| 80      | 0.00102851             | 334.70 | 335.93 | 1.0748   | 430  | 0.26684                | 3005.5 | 3325.7 | 7.4728  |
| 85      | 0.00103188             | 355.67 | 356.91 | 1.1338   | 440  | 0.27083                | 3022.2 | 3347.2 | 7.5032  |
| 90      | 0.00103540             | 376.67 | 377.91 | 1.1921   | 450  | 0.27482                | 3038.9 | 3368.7 | 7.5332  |
| 95      | 0.00103907             | 397.69 | 398.94 | 1.2496   | 460  | 0.27879                | 3055.8 | 3390.3 | 7.5628  |
| 100     | 0.00104290             | 418.74 | 419.99 | 1.3064   | 470  | 0.28276                | 3072.6 | 3411.9 | 7.5921  |
| 105     | 0.00104688             | 439.81 | 441.07 | 1.3625   | 480  | 0.28673                | 3089.4 | 3433.5 | 7.6210  |
| 110     | 0.00105102             | 460.92 | 462.18 | 1.4179   | 490  | 0.29069                | 3106.4 | 3455.2 | 7.6496  |
| 115     | 0.00105531             | 482.06 | 483.33 | 1.4728   | 500  | 0.29464                | 3123.3 | 3476.9 | 7.6779  |
| 120     | 0.00105976             | 503.25 | 504.52 | 1.5270   | 520  | 0.30253                | 3157.5 | 3520.5 | 7.7336  |
| 125     | 0.00106438             | 524.46 | 525.74 | 1.5806   | 540  | 0.31041                | 3191.8 | 3564.3 | 7.7881  |
| 130     | 0.00106916             | 545.73 | 547.01 | 1.6337   | 560  | 0.31826                | 3226.4 | 3608.3 | 7.8416  |
| 135     | 0.00107410             | 567.04 | 568.33 | 1.6863   | 580  | 0.32611                | 3261.3 | 3652.6 | 7.8940  |
| 140     | 0.00107923             | 588.41 | 589.71 | 1.7383   | 600  | 0.33394                | 3296.3 | 3697.0 | 7.9455  |
| 145     | 0.00108453             | 609.83 | 611.13 | 1.7899   | 620  | 0.34177                | 3331.6 | 3741.7 | 7.9961  |
| 150     | 0.00109001             | 631.32 | 632.63 | 1.8410   | 640  | 0.34958                | 3367.1 | 3786.6 | 8.0459  |
| 155     | 0.00109569             | 652.87 | 654.18 | 1.8916   | 660  | 0.35739                | 3402.9 | 3831.8 | 8.0948  |
| 160     | 0.00110157             | 674.49 | 675.81 | 1.9419   | 680  | 0.36518                | 3439.0 | 3877.2 | 8.1430  |
| 165     | 0.00110765             | 696.19 | 697.52 | 1.9917   | 700  | 0.37297                | 3475.3 | 3922.9 | 8.1904  |
| 170     | 0.00111395             | 717.97 | 719.31 | 2.0411   | 720  | 0.38076                | 3511.9 | 3968.8 | 8.2371  |
| 175     | 0.00112047             | 739.84 | 741.18 | 2.0902   | 740  | 0.38853                | 3548.8 | 4015.0 | 8.2832  |
| 180     | 0.00112723             | 761.80 | 763.15 | 2.1390   | 760  | 0.39631                | 3585.9 | 4061.5 | 8.3286  |
| 185     | 0.00113424             | 783.87 | 785.23 | 2.1874   | 780  | 0.40407                | 3623.3 | 4108.2 | 8.3734  |
| 187.957 | 0.00113850             | 796.96 | 798.33 | 2.2159   | 800  | 0.41184                | 3661.0 | 4155.2 | 8.4176  |
|         |                        |        |        |          | 820  | 0.41959                | 3698.9 | 4202.4 | 8.4612  |
|         |                        |        |        |          | 840  | 0.42735                | 3737.1 | 4249.9 | 8.5042  |
|         |                        |        |        |          | 860  | 0.43510                | 3775.6 | 4297.7 | 8.5468  |
|         |                        |        |        |          | 880  | 0.44285                | 3814.3 | 4345.7 | 8.5888  |
|         |                        |        |        |          | 900  | 0.45059                | 3853.3 | 4394.0 | 8.6303  |
|         |                        |        |        |          | 920  | 0.45834                | 3892.6 | 4442.6 | 8.6713  |
|         |                        |        |        |          | 940  | 0.46608                | 3932.1 | 4491.4 | 8.7119  |
|         |                        |        |        |          | 960  | 0.47381                | 3971.9 | 4540.5 | 8.7520  |
|         |                        |        |        |          | 980  | 0.48155                | 4011.9 | 4589.8 | 8.7917  |
|         |                        |        |        |          | 1000 | 0.48928                | 4052.3 | 4639.4 | 8.8310  |

Water/Steam at  $p = 2.0 \text{ MPa}$  ( $T_{\text{sat}} = 212.377^\circ\text{C}$ )

| $T$     | $v$                    | $u$    | $h$    | $s$      |
|---------|------------------------|--------|--------|----------|
| °C      | $\text{m}^3/\text{kg}$ | kJ/kg  | kJ/kg  | kJ/kg K  |
| 0       | 0.00099919             | -0.01  | 1.99   | -0.00003 |
| 5       | 0.00099910             | 21.01  | 23.01  | 0.07622  |
| 10      | 0.00099939             | 41.97  | 43.97  | 0.15091  |
| 15      | 0.00100001             | 62.89  | 64.89  | 0.22416  |
| 20      | 0.00100093             | 83.79  | 85.79  | 0.29607  |
| 25      | 0.00100210             | 104.68 | 106.68 | 0.36671  |
| 30      | 0.00100352             | 125.54 | 127.55 | 0.43615  |
| 35      | 0.00100516             | 146.42 | 148.43 | 0.50444  |
| 40      | 0.00100700             | 167.29 | 169.30 | 0.57163  |
| 45      | 0.00100904             | 188.15 | 190.17 | 0.63776  |
| 50      | 0.00101126             | 209.04 | 211.06 | 0.70289  |
| 55      | 0.00101366             | 229.91 | 231.94 | 0.76704  |
| 60      | 0.00101623             | 250.81 | 252.84 | 0.83024  |
| 65      | 0.00101897             | 271.71 | 273.75 | 0.89254  |
| 70      | 0.00102187             | 292.64 | 294.68 | 0.95396  |
| 75      | 0.00102492             | 313.56 | 315.61 | 1.0145   |
| 80      | 0.00102813             | 334.51 | 336.57 | 1.0743   |
| 85      | 0.00103149             | 355.48 | 357.54 | 1.1333   |
| 90      | 0.00103501             | 376.46 | 378.53 | 1.1915   |
| 95      | 0.00103867             | 397.47 | 399.55 | 1.2490   |
| 100     | 0.00104249             | 418.51 | 420.59 | 1.3057   |
| 105     | 0.00104647             | 439.57 | 441.66 | 1.3618   |
| 110     | 0.00105059             | 460.67 | 462.77 | 1.4173   |
| 115     | 0.00105487             | 481.79 | 483.90 | 1.4721   |
| 120     | 0.00105931             | 502.96 | 505.08 | 1.5263   |
| 125     | 0.00106392             | 524.16 | 526.29 | 1.5799   |
| 130     | 0.00106868             | 545.41 | 547.55 | 1.6330   |
| 135     | 0.00107362             | 566.71 | 568.86 | 1.6855   |
| 140     | 0.00107872             | 588.06 | 590.22 | 1.7375   |
| 145     | 0.00108401             | 609.47 | 611.64 | 1.7890   |
| 150     | 0.00108948             | 630.94 | 633.12 | 1.8401   |
| 155     | 0.00109513             | 652.48 | 654.67 | 1.8907   |
| 160     | 0.00110099             | 674.08 | 676.28 | 1.9409   |
| 165     | 0.00110705             | 695.76 | 697.97 | 1.9907   |
| 170     | 0.00111332             | 717.51 | 719.74 | 2.0401   |
| 175     | 0.00111982             | 739.36 | 741.60 | 2.0892   |
| 180     | 0.00112655             | 761.31 | 763.56 | 2.1379   |
| 185     | 0.00113353             | 783.34 | 785.61 | 2.1863   |
| 190     | 0.00114076             | 805.49 | 807.77 | 2.2344   |
| 195     | 0.00114827             | 827.75 | 830.05 | 2.2822   |
| 200     | 0.00115607             | 850.14 | 852.45 | 2.3298   |
| 210     | 0.00117262             | 895.31 | 897.66 | 2.4244   |
| 212.377 | 0.00117675             | 906.15 | 908.50 | 2.4468   |

| $T$  | $v$                    | $u$    | $h$    | $s$     |
|------|------------------------|--------|--------|---------|
| °C   | $\text{m}^3/\text{kg}$ | kJ/kg  | kJ/kg  | kJ/kg K |
| 270  | 0.11726                | 2718.6 | 2953.1 | 6.6409  |
| 280  | 0.12005                | 2737.0 | 2977.1 | 6.6849  |
| 290  | 0.12280                | 2755.2 | 3000.8 | 6.7273  |
| 300  | 0.12551                | 2773.2 | 3024.2 | 6.7684  |
| 310  | 0.12818                | 2790.9 | 3047.3 | 6.8083  |
| 320  | 0.13082                | 2808.5 | 3070.1 | 6.8472  |
| 330  | 0.13344                | 2825.9 | 3092.8 | 6.8851  |
| 340  | 0.13603                | 2843.2 | 3115.3 | 6.9221  |
| 350  | 0.13860                | 2860.5 | 3137.7 | 6.9583  |
| 360  | 0.14115                | 2877.6 | 3159.9 | 6.9937  |
| 370  | 0.14369                | 2894.7 | 3182.1 | 7.0285  |
| 380  | 0.14621                | 2911.8 | 3204.2 | 7.0627  |
| 390  | 0.14872                | 2928.9 | 3226.3 | 7.0962  |
| 400  | 0.15121                | 2945.9 | 3248.3 | 7.1292  |
| 410  | 0.15370                | 2962.9 | 3270.3 | 7.1616  |
| 420  | 0.15617                | 2980.0 | 3292.3 | 7.1935  |
| 430  | 0.15864                | 2997.0 | 3314.3 | 7.2250  |
| 440  | 0.16109                | 3014.1 | 3336.3 | 7.2560  |
| 450  | 0.16354                | 3031.1 | 3358.2 | 7.2866  |
| 460  | 0.16598                | 3048.2 | 3380.2 | 7.3168  |
| 470  | 0.16842                | 3065.4 | 3402.2 | 7.3466  |
| 480  | 0.17085                | 3082.5 | 3424.2 | 7.3760  |
| 490  | 0.17327                | 3099.7 | 3446.2 | 7.4050  |
| 500  | 0.17568                | 3116.8 | 3468.2 | 7.4337  |
| 520  | 0.18050                | 3151.4 | 3512.4 | 7.4901  |
| 540  | 0.18530                | 3186.1 | 3556.7 | 7.5453  |
| 560  | 0.19009                | 3221.0 | 3601.2 | 7.5994  |
| 580  | 0.19486                | 3256.2 | 3645.9 | 7.6523  |
| 600  | 0.19961                | 3291.5 | 3690.7 | 7.7043  |
| 620  | 0.20436                | 3327.1 | 3735.8 | 7.7553  |
| 640  | 0.20910                | 3362.8 | 3781.0 | 7.8054  |
| 660  | 0.21383                | 3398.8 | 3826.5 | 7.8547  |
| 680  | 0.21855                | 3435.1 | 3872.2 | 7.9032  |
| 700  | 0.22326                | 3471.7 | 3918.2 | 7.9509  |
| 720  | 0.22797                | 3508.4 | 3964.3 | 7.9978  |
| 740  | 0.23267                | 3545.5 | 4010.8 | 8.0441  |
| 760  | 0.23737                | 3582.7 | 4057.4 | 8.0897  |
| 780  | 0.24206                | 3620.2 | 4104.3 | 8.1347  |
| 800  | 0.24674                | 3658.0 | 4151.5 | 8.1790  |
| 820  | 0.25142                | 3696.1 | 4198.9 | 8.2228  |
| 840  | 0.25610                | 3734.4 | 4246.6 | 8.2660  |
| 860  | 0.26078                | 3772.9 | 4294.5 | 8.3087  |
| 880  | 0.26545                | 3811.8 | 4342.7 | 8.3509  |
| 900  | 0.27012                | 3850.9 | 4391.1 | 8.3925  |
| 920  | 0.27478                | 3890.2 | 4439.8 | 8.4336  |
| 940  | 0.27944                | 3929.8 | 4488.7 | 8.4743  |
| 960  | 0.28411                | 3969.7 | 4537.9 | 8.5145  |
| 980  | 0.28876                | 4009.9 | 4587.4 | 8.5543  |
| 1000 | 0.29342                | 4050.2 | 4637.0 | 8.5936  |

[20 marks]

$$P_1 = 20 \text{ bar}$$

$$P_2 = 12 \text{ bar}$$

$$T_1 = 300^\circ\text{C}$$

$$T_2 = 240^\circ\text{C}$$

$\eta_{nozzle} = 96\%$ ,  $\alpha = 18^\circ$  find  $u$ ?  $\phi$ ?

$$\phi = \theta - \alpha, k = .88, m_s = 1440 \text{ kg/hr}$$

$$u = \frac{v_1 + v_2}{2}$$

(a) Axial thrust (b) Diagram Power

(c) Dia efficiency

From steam table,  $h_1 = 3024.2 \text{ kJ/kg}$

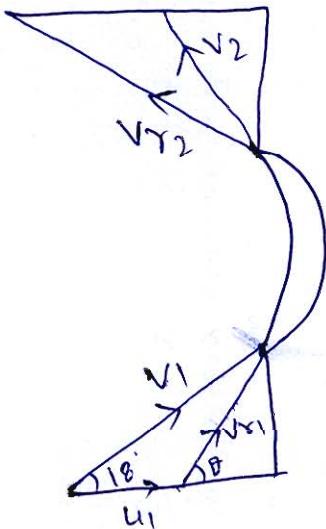
$$h_2 = 2912.7 \text{ kJ/kg}$$

$$\text{Power} = m_s (h_1 - h_2) = \frac{1440}{3600} (3024.2 - 2912.7)$$

$$\text{Power} = 44.6 \times \eta_{nozzle} = 44.6 \times .96$$

$$= 42.816 \text{ kW}$$

Ans.



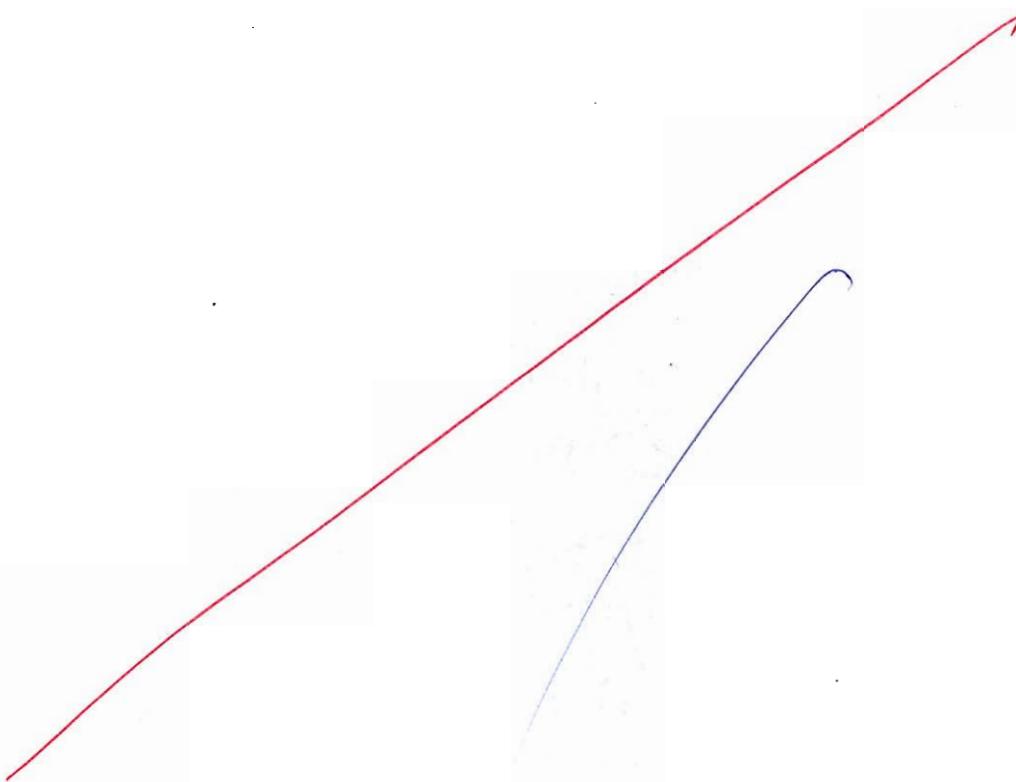
Refer Solution

$$\text{Power} = m_s u^2 F = 42.816$$

$$\frac{144\phi}{3600} \times u^2 = 42.816$$

$$u = 327.169 \text{ m/s}$$

Incomplete solution



- Q.3 (a)** A centrifugal compressor running at 10500 rpm delivers  $840 \text{ m}^3/\text{min}$  of free air. The air is compressed from 100 kPa and  $7^\circ\text{C}$  to a compression ratio of 4 with an isentropic efficiency of 84%. Impeller has radial blade at outlet and flow velocity of 62 m/s may be assumed to be constant throughout. The slip factor is 0.92. At inlet, blade area coefficient is 0.86. The outer radius of impeller is twice the inner. Determine (i) Final temperature of air, (ii) Theoretical power required, (iii) Diameter of impeller at inlet and outlet, (iv) Breadth of impeller at inlet (v) Blade angle of impeller at inlet and (vi) Blade angle diffuser at inlet. [For air take:  $C_p = 1.005 \text{ kJ/kgK}$ ,  $\gamma = 1.4$ ]

[20 marks]

given  $N = 10500 \text{ rpm}$ ,  $Q = 840 \text{ m}^3/\text{min}$

$$P_1 = 100 \text{ kPa} \quad n = 4, \eta_{isn} = .84$$

$$T_1 = T_c = 280 \text{ K}, \phi = 90^\circ, V_{f1} = V_{f2} = 62 \text{ m/s}$$

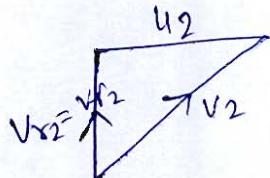
$$\phi_s = .92 \quad KA_1 = .86 \quad D_2 = 2D_1$$

(i) Find  $T_2'$ , Theoretical power  $\rightarrow$   $C_p = 1.005 \text{ kJ/kg K}$

$$\frac{T_2}{T_1} = (n)^{\frac{1}{\gamma-1}} \Rightarrow T_2 = 280(4)^{\frac{1}{1.4}} = 487.5083 \text{ K}$$

$$.84 = \frac{C_p(T_2 - T_1)}{(T_2' - T_1)} \Rightarrow T_2' = 527.033 \text{ K} \quad \text{or} \quad 254.033^\circ\text{C}$$

$$\text{Theoretical Power} = C_p(T_2' - T_1) = 1.005(527.033 - 280) \\ = \boxed{248.2681 \text{ KJ/Kg}} \text{ Ans.}$$



$$V_{r2} = V_{f2} = 62 \text{ m/s} = V_1$$

$$\text{Power} = \Phi_s \cdot u_2^2 \Rightarrow 248.2681 \times 10^3 = 0.92 \times u_2^2 \\ u_2 = 519.477 \text{ m/s}$$

$$u_2 = \frac{\pi D_2 N}{60} \Rightarrow 519.477 = \frac{\pi \times D_2 \times 10500}{60}$$

$$D_2 = \boxed{0.9448 \text{ m}} \quad D_1 = \boxed{0.4724 \text{ m}} \text{ Ans.$$

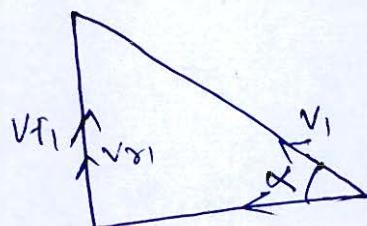
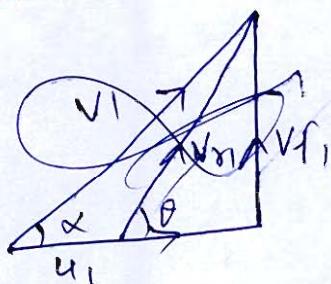
$$Q = \pi D_1 b_1 \times V_{f1} \times K A_1$$

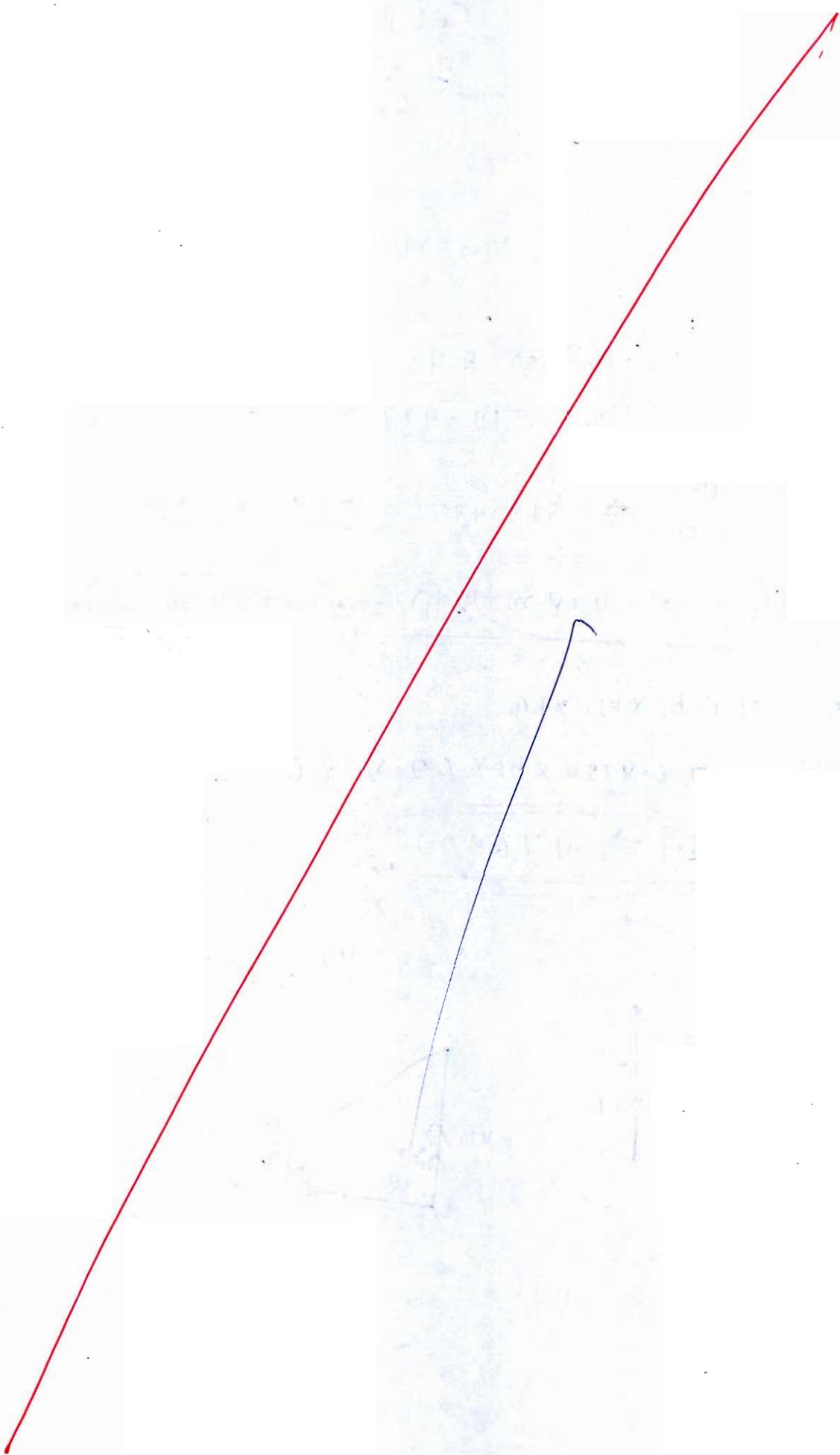
$$\frac{840}{60} = \pi \times 0.4724 \times b_1 \times 62 \times 0.86$$

$$b_1 = \boxed{0.1769 \text{ m}} \text{ Ans.$$

Refer

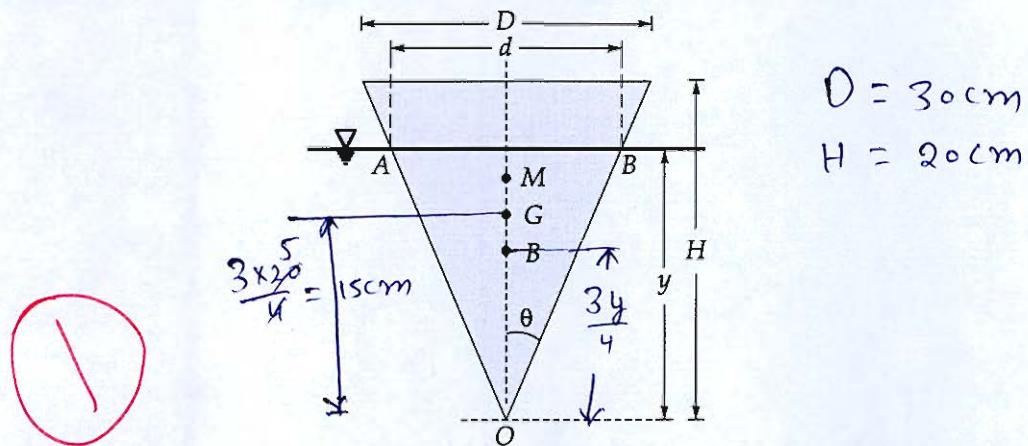
Solution





Q.3 (b)

A solid cone of diameter 30 cm and height 20 cm float with its vertex downwards in water as shown in figure. Analyze whether the cone would be stable and float in water with its axis vertical if the specific gravity of cone material is 0.8.

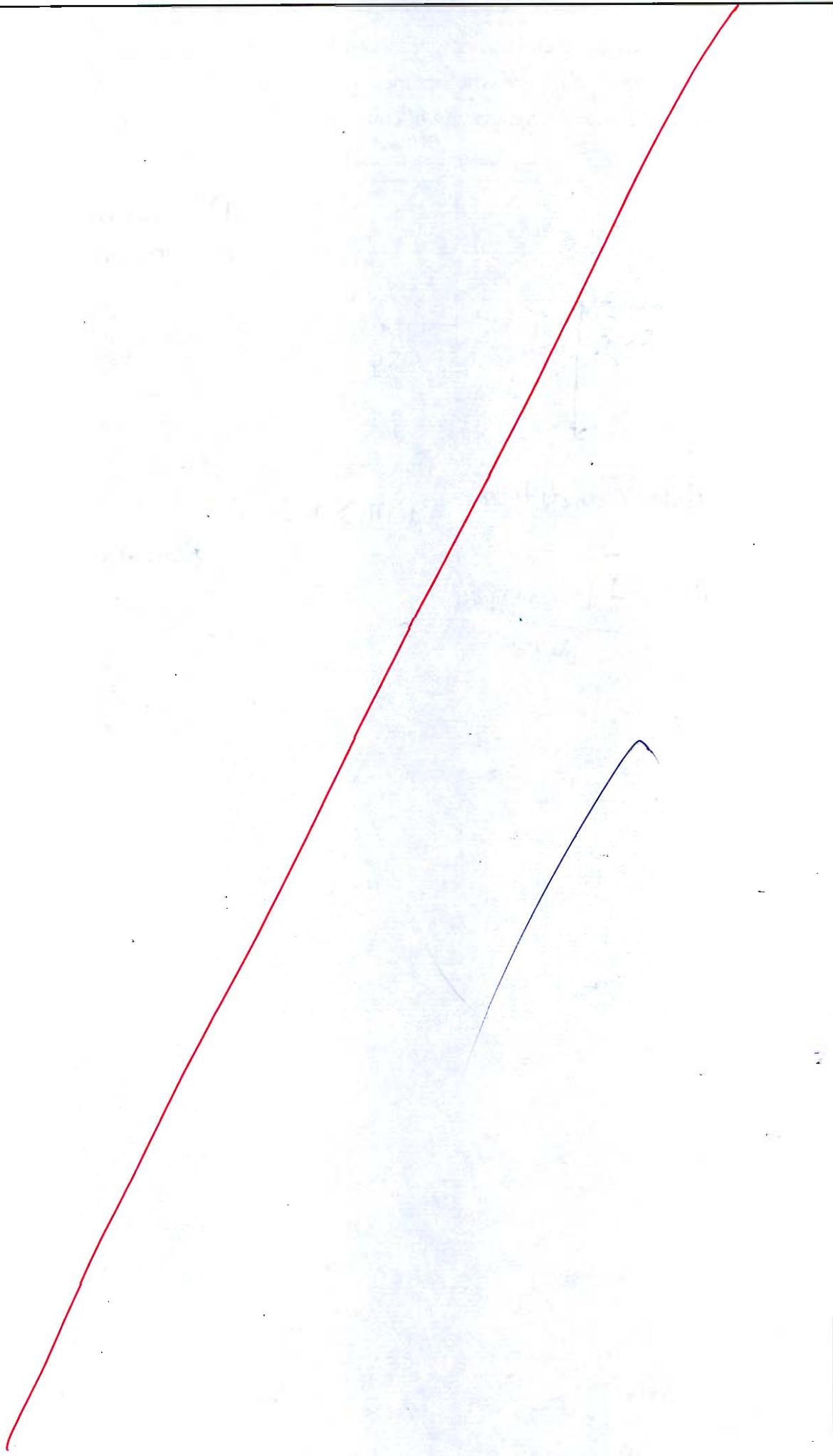


for stable condition,  $B_M > B_G$

[20 marks]

$$B_M = \frac{I_{\text{free surface}}}{V_f d}$$

$$B_G = 15 - \frac{3}{4}y$$



Q.3 (c) Explain briefly the function of a draft tube. How Kaplan turbine differs with Francis turbine. A Kaplan turbine operating under a head of 7.6 m develops 1835 kW with an overall efficiency of 88%. The turbine is set 2.6 m above the tail water level and vacuum gauge inserted at turbine outlet records a suction head of 3.17 m. Calculate the efficiency of the draft tube if it has an inlet diameter of 3.1 m and the loss of head due to friction in the draft tube equals 25% of kinetic head at outlet.

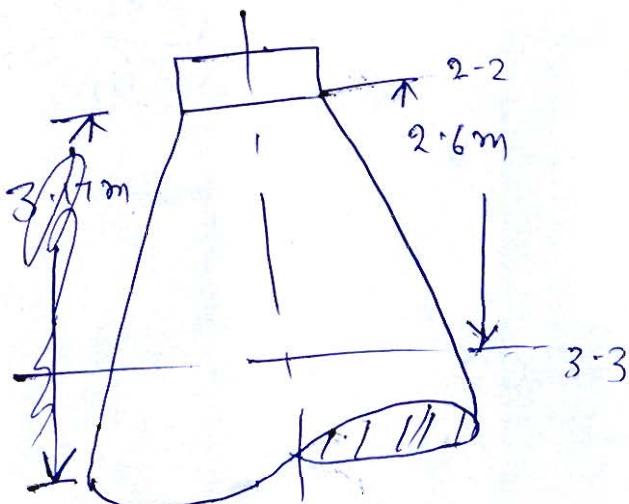
### Functions of Draft tube

[20 marks]

- (i) It utilizes the kinetic energy, which is gone waste on the runner exit ( $\frac{V_2^2}{2g}$ ).
- (ii) It ensures the water out smoothly from the runner to tail race level.
- (iii) It avoid the splashing of water.
- (iv) It reduce the wastage of water.

10

given data,  $H = 7.6 \text{ m}$ ,  $P = 1835 \text{ kW}$ ,  $\eta_o = .88$



$$\frac{P_2}{\rho g} = 3.17 \text{ m} (\text{vacuum}) \\ = 7.13 \text{ m (abs)} \\ D_2 = 3.1 \text{ m}$$

$$h_f = .25 \times 3.17 = \\ .7925 \text{ m} \\ \text{find } \eta_{DT}$$

$$\eta_{DT} = \frac{\frac{V_2^2}{2g} - h_f - \frac{V_3^2}{2g}}{\frac{V_2^2}{2g}}$$

$$\eta_o = \frac{P}{\rho g Q H}$$

$$.88 = \frac{1835 \times 10^3}{10^3 \times 9.81 \times Q \times 7.6}$$

$$Q = 27.9686 \text{ m}^3/\text{s}$$

$$V_2 = \frac{Q}{A_2} \Rightarrow \frac{27.968 \times 4}{\pi (3.1)^2} = 3.7055 \text{ m/s}$$

Apply Bernoulli's eqn b/w 2 & 3

$$\frac{V_2^2}{2g} + \frac{P_2}{\rho g} + 2.6 = \frac{V_3^2}{2g} + \frac{P_3}{\rho g} + h_f$$

$$\frac{3.7055^2}{2 \times 9.81} + \cancel{2.6} + 2.6 - 10.3 \cancel{- 0.25} = \frac{V_3^2}{2g} + \frac{0.25}{2g}$$

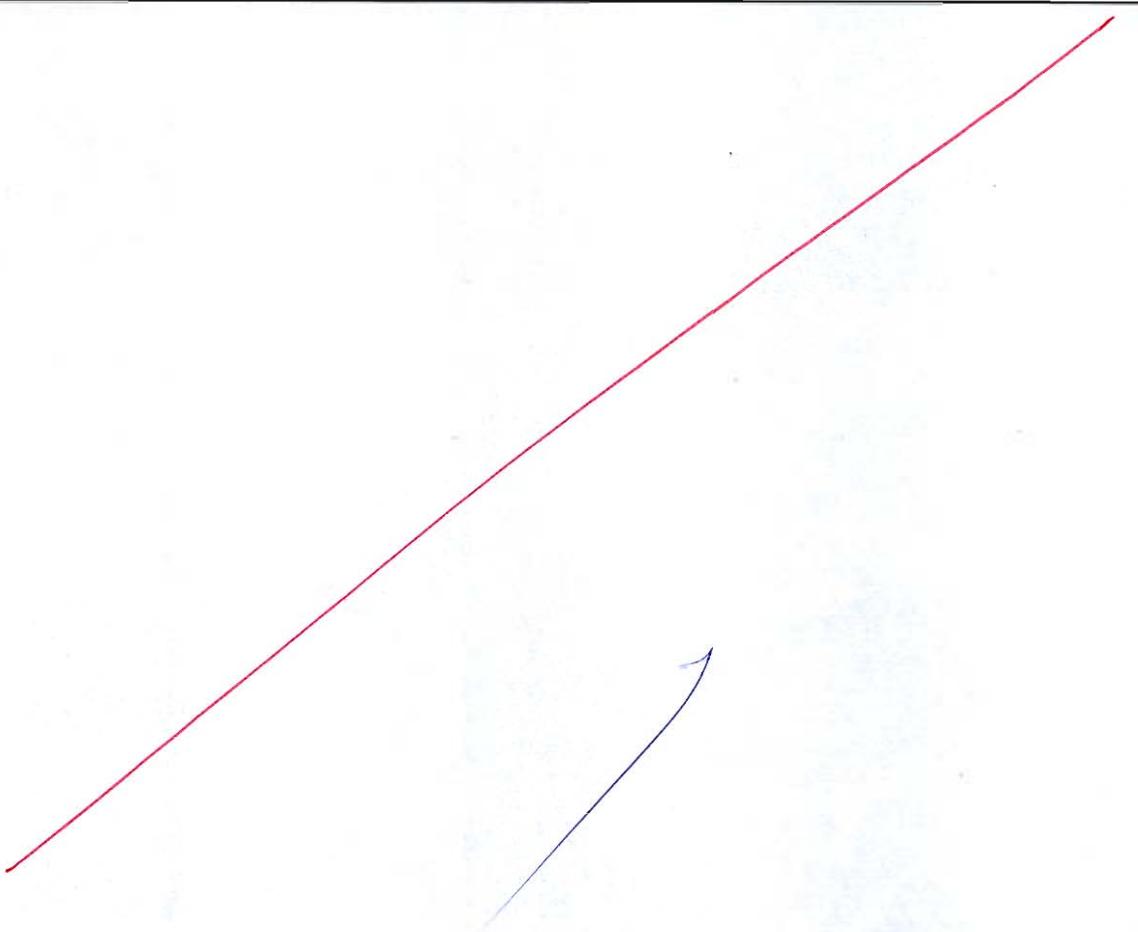
$$0.12983 = 1.25 \frac{V_3^2}{2g}$$

$$\frac{V_3^2}{2g} = 0.10386 \text{ m}$$

$$h_f = 0.02596 \text{ m}$$

$$\gamma_{DT} = \frac{3.7055 - 0.02596 - 0.10386}{3.7055}$$

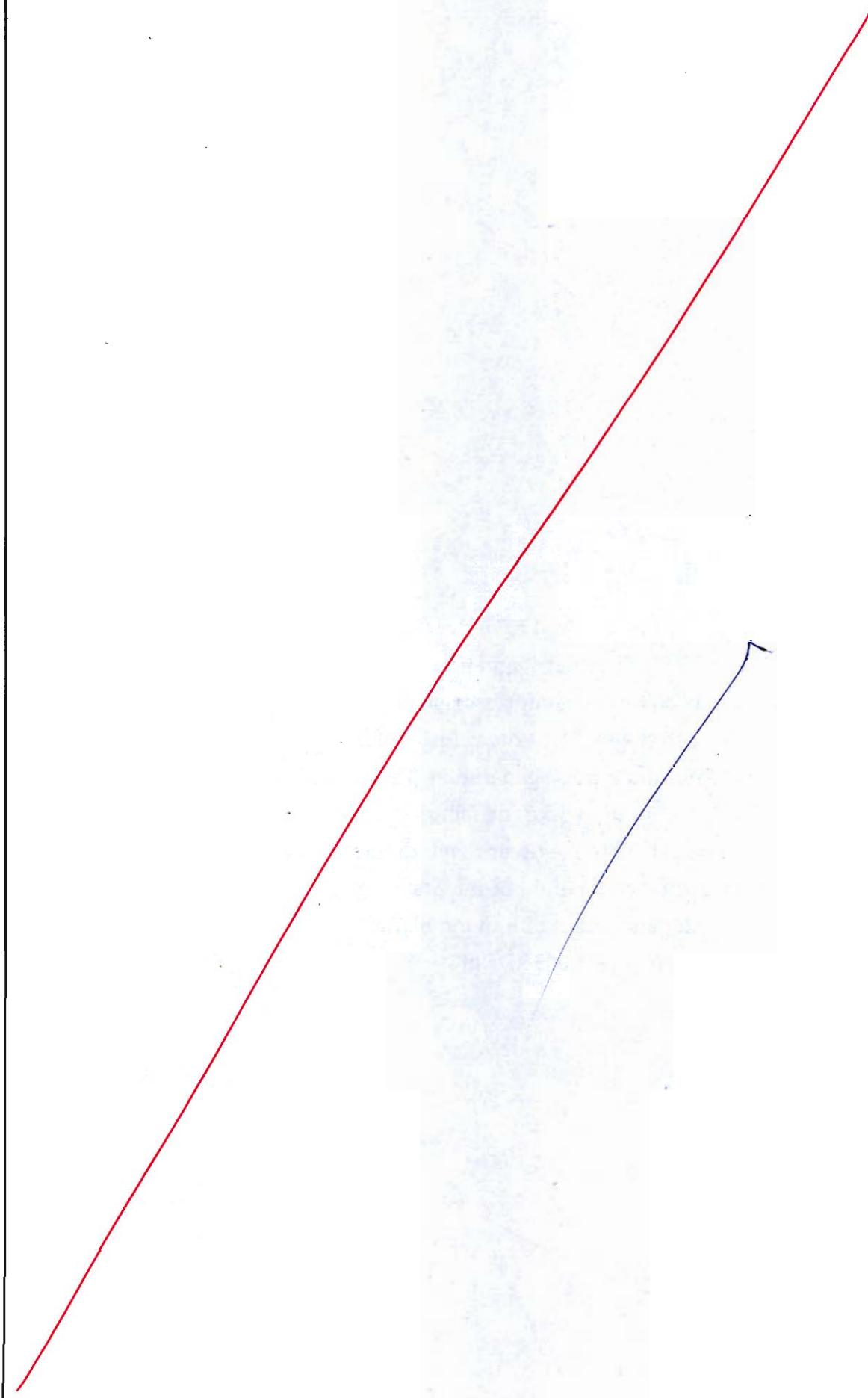
96.49%  
Ans

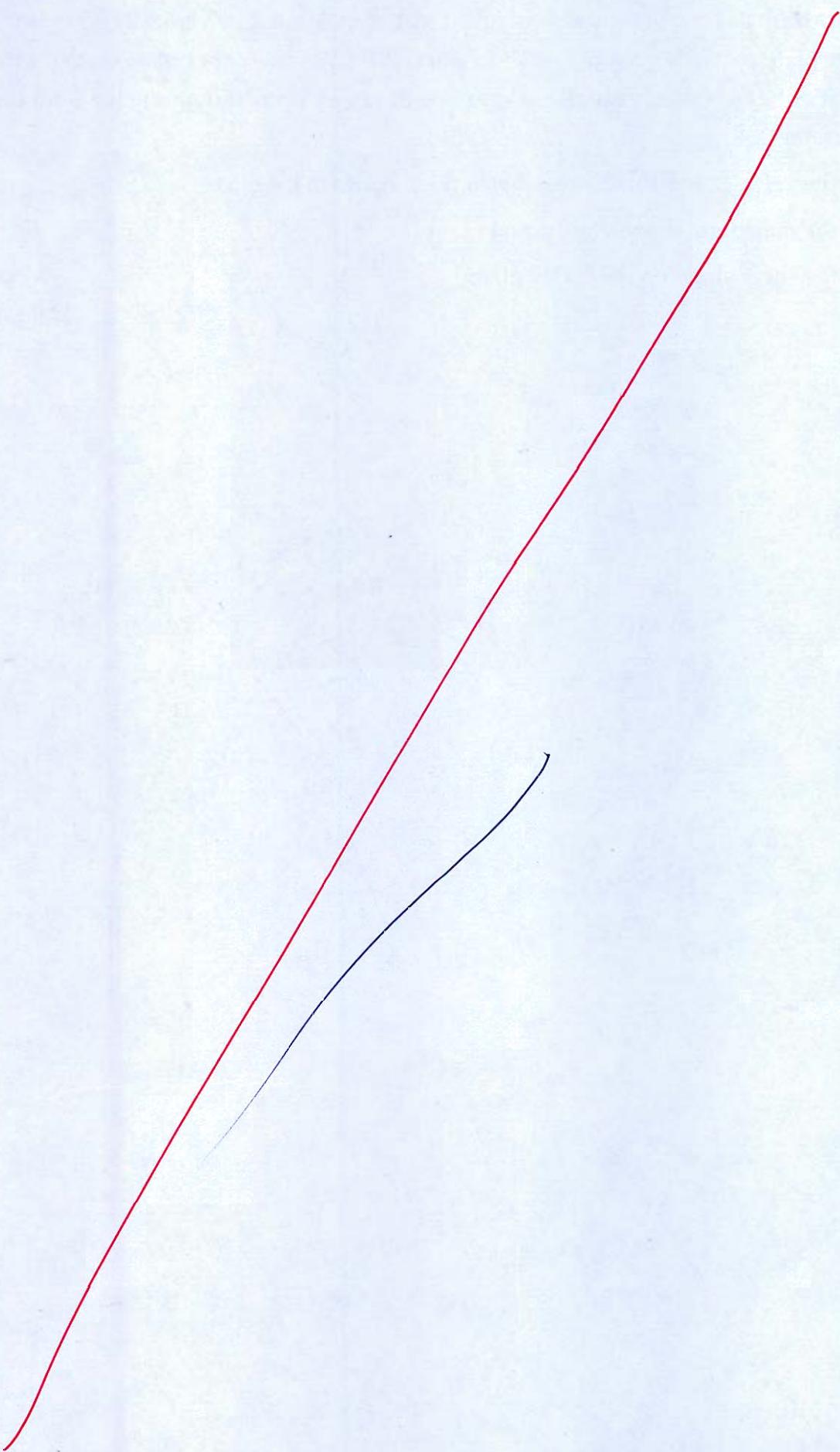


**Q.4 (a)** An axial flow compressor is required to deliver air at the rate 48 kg/s and provide a total pressure ratio of 5 : 1. The inlet stagnation conditions being 290 K and 1 bar. The isentropic efficiency is 88%. The compressor shall have 10 stages with equal rise in total temperature in each stages. The axial velocity of flow is 160 m/s and blade speed is kept at 210 m/s to minimise noise generation. The stage degree of reaction at mean blade height is 50%. Assuming workdone factor as 0.86, calculate all the fluid angles of the first stage. Also, calculate the tip and hub diameter, if hub-tip diameter ratio is 0.82. Also, determine the blade height of the first stage and the speed in rpm. Draw velocity diagram at inlet and outlet of moving blade.

Take  $R = 0.287 \text{ kJ/kgK}$  and  $c_p = 1.005 \text{ kJ/kgK}$ .

[20 marks]



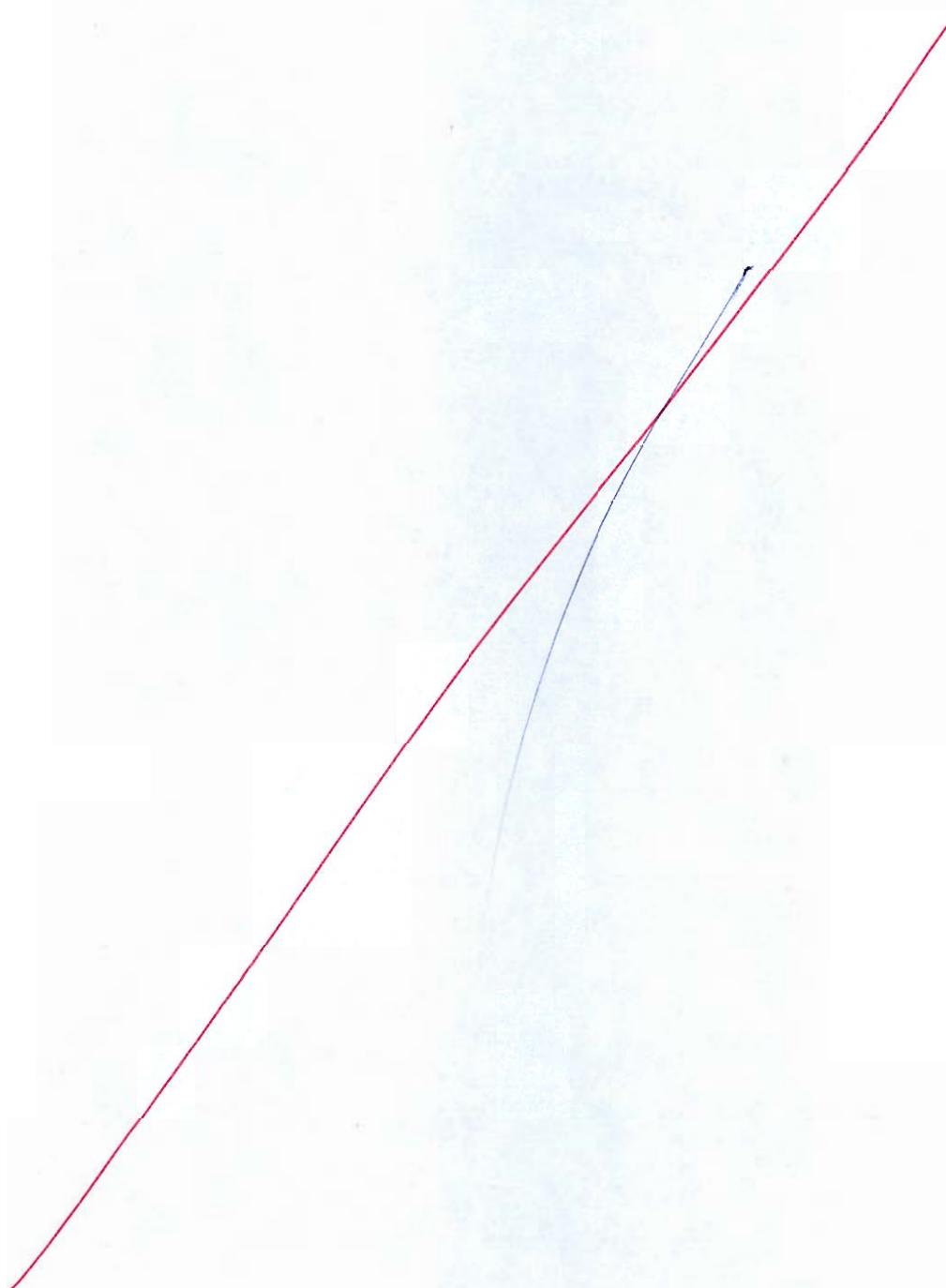


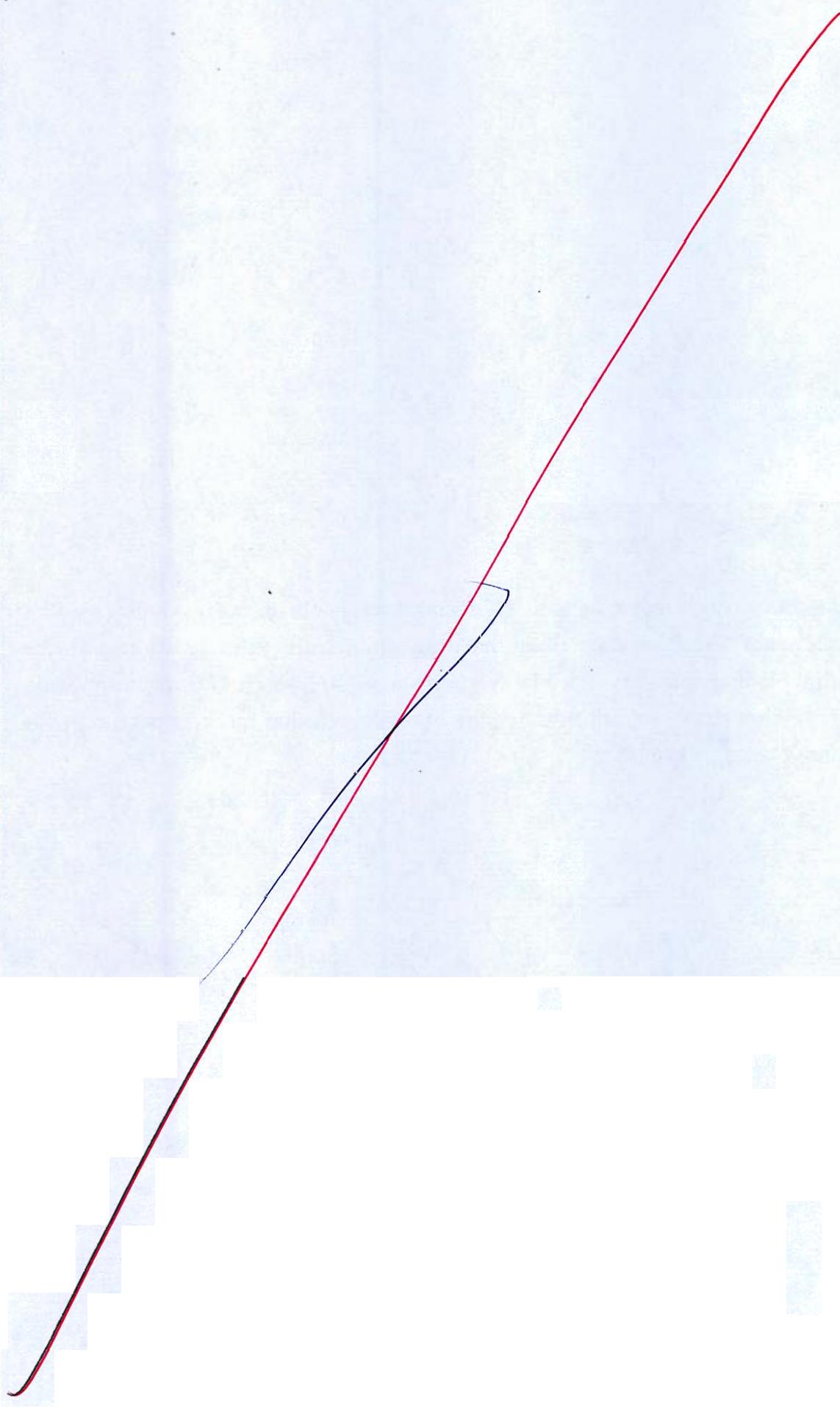
Q.4 (b)

Laminar flow of a fluid of viscosity  $0.8 \text{ kg/ms}$  and density  $1300 \text{ kg/m}^3$  occurs between a pair of plates of extensive width. The plates are  $15 \text{ mm}$  apart and are inclined at  $45^\circ$  to the horizontal. Pressure guages mounted at two points  $1.2 \text{ m}$  vertically apart on the upper plate record a pressure of  $75 \text{ kPa}$  and  $250 \text{ kPa}$ . The upper plate moves with a velocity of  $2 \text{ m/s}$  relative to the lower plate but in a direction opposite the fluid flow. Determine

- (i) the velocity and shear stress distribution between the plate.
- (ii) the maximum flow velocity, and
- (iii) the shear stress on the upper plate.

[20 marks]

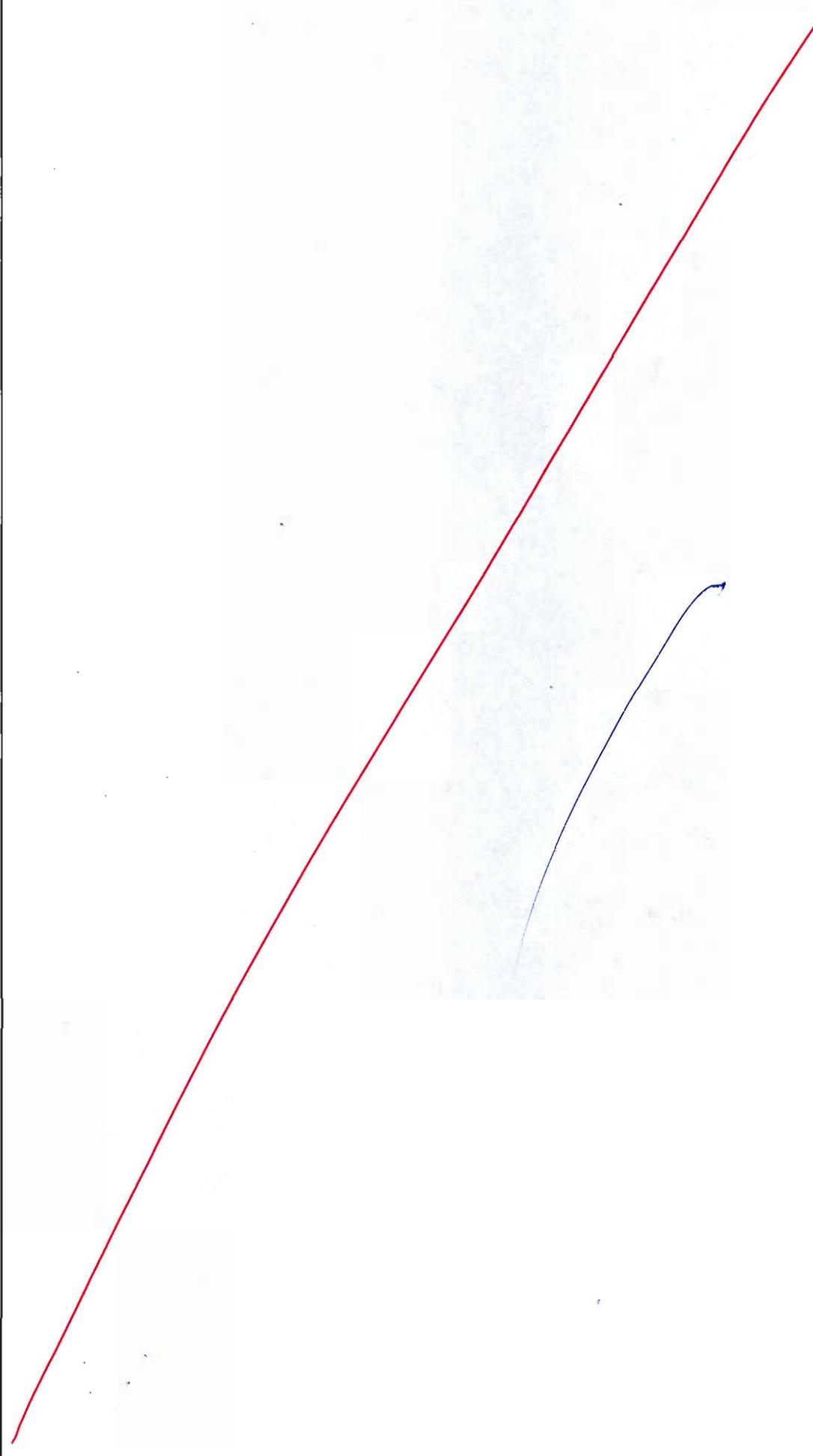


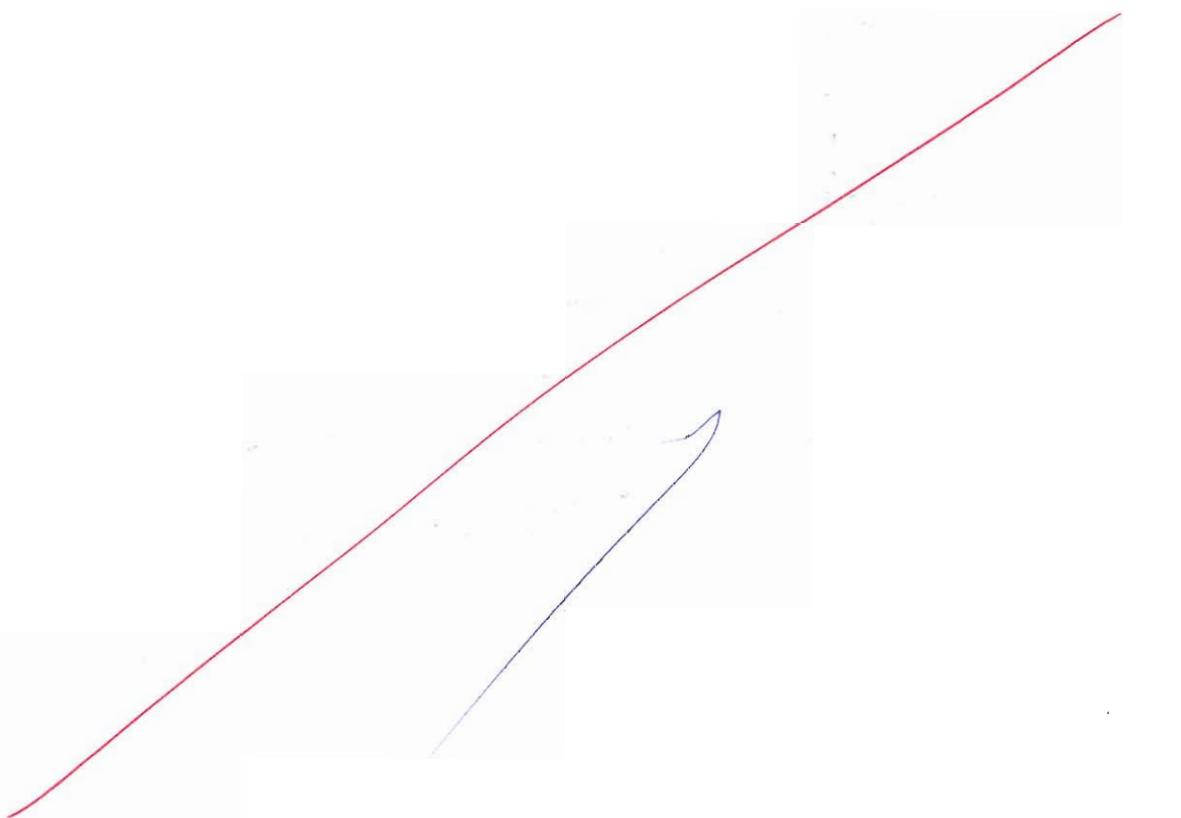


- Q.4 (c) What are the advantages of using a 50% reaction stage, obtain the maximum blading efficiency of a 50% reaction stage. Show that the diagram work per unit mass of steam for maximum blading efficiency of a 50% reaction stage is  $U^2$ , where  $U$  is the mean blade efficiency. Also, draw the velocity diagram of a 50% reaction turbine operating with maximum blading efficiency.

[20 marks]







## Section : B

Q.5 (a)

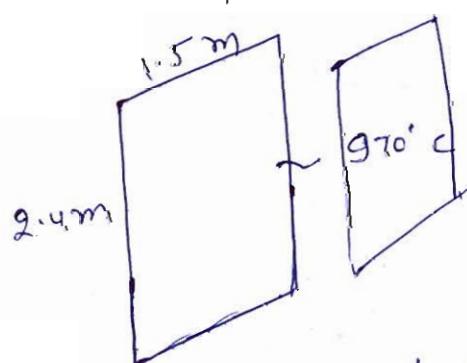
A nuclear reactor with its core constructed of parallel vertical plates 2.4 m high and 1.50 m wide has been designed on free convection heating of liquid bismuth. The maximum temperature of the plate surface is limited to 970°C, while the lowest allowable temperature of bismuth is 330°C. Calculate the maximum possible heat dissipation from both sides of each plate. For convective coefficient the appropriate correlation is,

$$Nu = 0.13(Gr \cdot Pr)^{1/3}$$

The thermo-physical properties at mean film temperature of 650°C for bismuth are :  
 $\rho = 10^4 \text{ kg/m}^3$ ;  $\mu = 3.12 \text{ kg/m-h}$ ;  $c_p = 150.7 \text{ J/kgK}$ ;  $k = 13.02 \text{ W/mK}$

[12 marks]

(10)



$$T_{\text{allow}} = 330^\circ\text{C}$$

$$\beta = \frac{1}{\frac{1243 + 603}{2}} = \frac{1}{923}$$

Max heat dissipation = ?

$$\frac{h L}{k} = 0.13(Gr \cdot Pr)^{1/3}$$

$$Gr = \frac{g \beta \Delta T L^3}{(\mu \rho)^2} = \frac{9.81 \times 8.33 \times (2.4)^3}{\left( \frac{3.12}{3600 \times 10^4} \right)^2 \times 923} = \frac{12519.206}{3.44 \times 10^{-12}} = 3.64 \times 10^{12}$$

$$f_{cr} = \frac{MCP}{K} = \frac{3.12 \times 150.7}{3600 \times 13.02} = 0.010031$$

$$\frac{h \times 2.4}{13.02} = .13 \left( \frac{12519.206}{6259.603 \times 10^{12}} \times 0.010031 \right)^{1/3}$$

$$h = \cancel{28.031 \cdot 315} \text{ w/m}^2 K \quad 35316.9697 \text{ w/m}^2 K$$

$$Q = 4 \times h A (\Delta T) = 4 \times \cancel{28.031 \cdot 315} \times 2.4 \times 1.5 \times$$

$$G_{40} =$$

$$325.4811 \text{ mW}$$

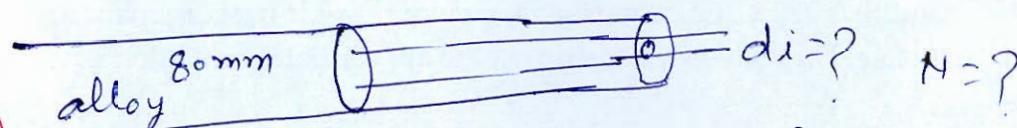
Ans

$$35316.9697$$

2  
(twice only)  
@ both sides

- Q.5 (b) A solid alloy shaft of 80 mm in diameter is coupled with a hollow steel shaft of same external diameter in series. If the angle of twist of the steel shaft per unit length is 72% of that of the alloy shaft, then find the inner diameter of the steel shaft. What will be the speed to transmit 720 kW, if the limiting shear stresses in the alloy and the steel are to be 64 MPa and 80 MPa, respectively? Take  $G_{\text{steel}} = 1.8 G_{\text{alloy}}$ .

[12 marks]



$$\left(\frac{\theta}{L}\right)_{\text{Steel}} = .72 \left(\frac{\theta}{L}\right)_{\text{Alloy}}$$

$$\rho = 720 \text{ kW}$$

$$T_{\text{alloy}} = 64 \text{ MPa}$$

$$G_{\text{Steel}} = 1.8 G_{\text{Alloy}} \quad T_{\text{Steel}} = 80 \text{ MPa}$$

$$\frac{\theta}{L} = \frac{T}{GJ}$$

$$\frac{T}{G_{\text{Steel}} \times \frac{\pi}{32} \times (80^4 - d_1^4)} = \frac{T}{G_{\text{Alloy}} \times \frac{\pi}{32} \times 80^4} \times .72$$

$$\frac{1}{1.8(80^4 - d_1^4)} = \frac{.72}{80^4}$$

$$\frac{80^4}{1.8 \times .72} = 80^4 - d_1^4 \Rightarrow d_1 = 55.304 \text{ mm}$$

Ans.

$$P = \frac{2\pi NT}{60}$$

$$T_{\text{alloy}} = \frac{16 T}{\pi d^3}$$

$$T = 6433.981 \text{ Nm}$$

$$64 = \frac{16 \times T}{\pi \times (0.08)^3}$$

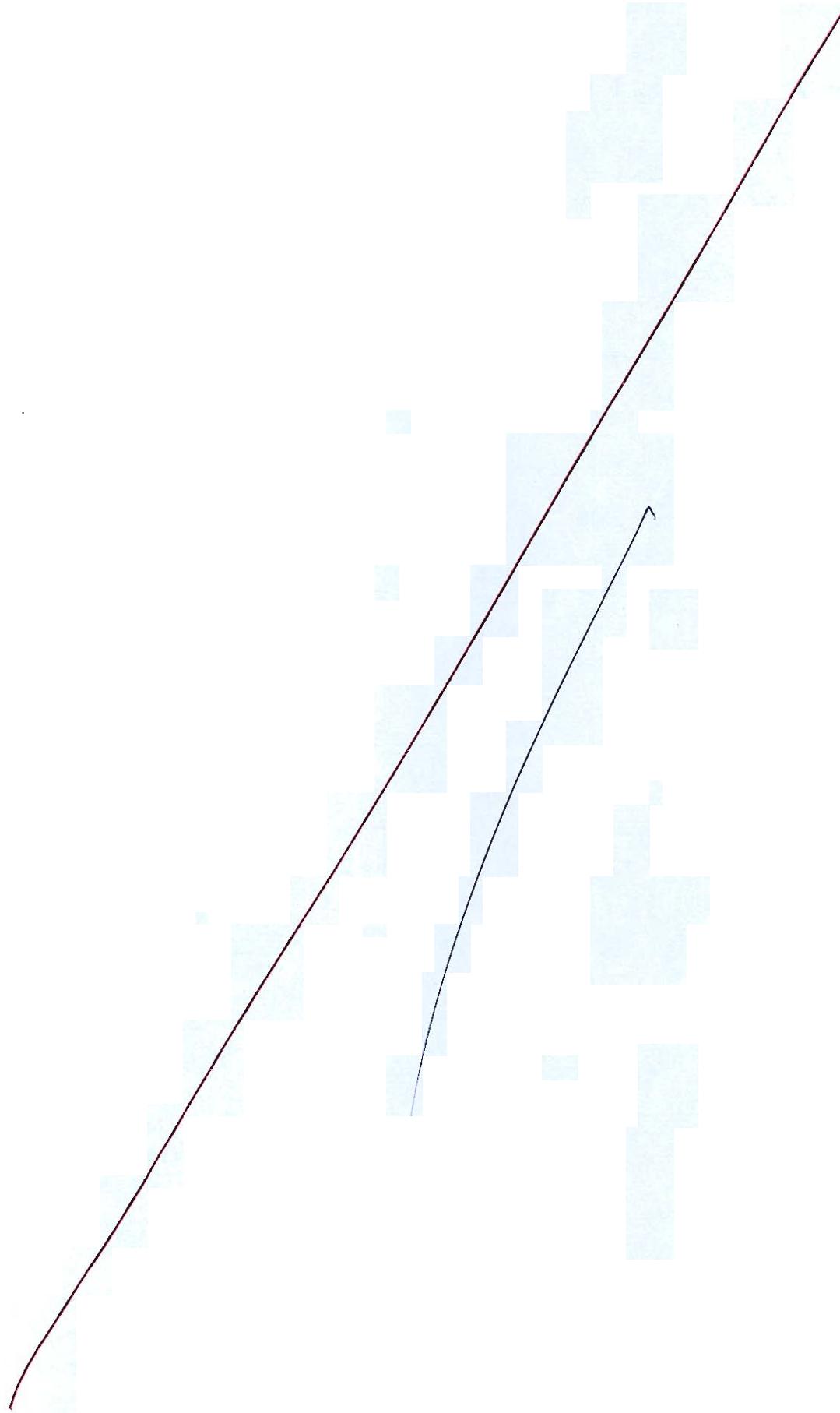
$$720 \times 10^3 = \frac{2\pi N \times 6433.981}{60}$$

$$N = 1068.621 \text{ rpm} \quad \underline{\text{Ans.}}$$

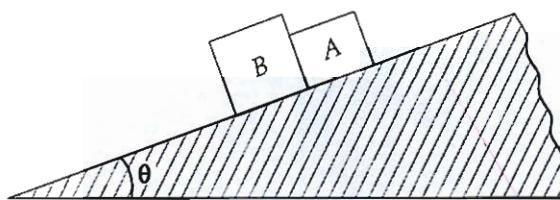
✓

- Q.5 (c) Briefly explain the working of thermostatic expansion valve for refrigerant flow control. Draw a neat sketch and write the functions of thermostatic expansion valve.

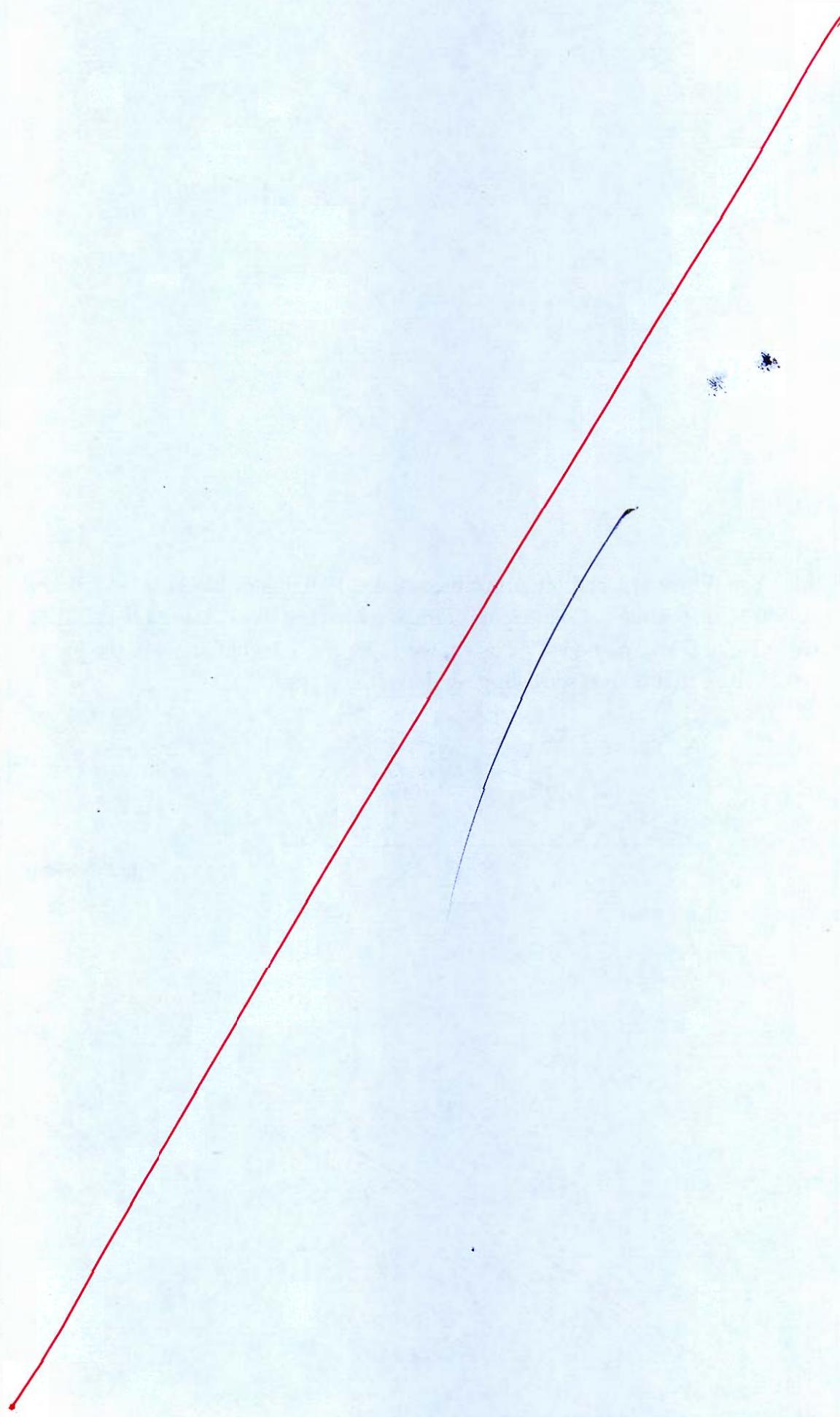
[12 marks]



- 5 (d) Two blocks A and B are placed on an inclined plane. Weights of blocks A and B are 450 N and 560 N, respectively. Coefficients of friction between block A and plane is 0.25 and between block B and plane is 0.32 as shown in figure. To what angle  $\theta$ , the plane should be raised so that bodies start slipping down the plane?



[12 marks]



Q.5 (e)

A piston cylinder assembly contains 0.2 kg superheated steam at 30 bar and 300°C. This assembly is placed in thermal contact with a reservoir at 300°C and the steam is allowed to expand to 1 bar and 300°C, calculate the maximum work that can be obtained from the steam. (Refer steam table attached)

Water/Steam at  $p = 0.10 \text{ MPa}$  ( $T_{\text{sat}} = 99.606^\circ\text{C}$ )

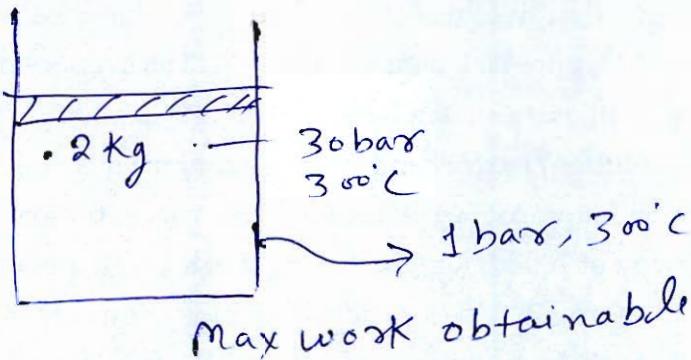
| $T$    | $v$        | $u$    | $h$    | $s$      |
|--------|------------|--------|--------|----------|
| °C     | m³/kg      | kJ/kg  | kJ/kg  | kJ/kg K  |
| *0     | 0.00100016 | -0.04  | 0.06   | -0.00015 |
| 5      | 0.00100003 | 21.02  | 21.12  | 0.07625  |
| 10     | 0.00100030 | 42.02  | 42.12  | 0.15108  |
| 15     | 0.00100090 | 62.98  | 63.08  | 0.22445  |
| 20     | 0.00100180 | 83.91  | 84.01  | 0.29646  |
| 25     | 0.00100296 | 104.82 | 104.92 | 0.36720  |
| 30     | 0.00100437 | 125.72 | 125.82 | 0.43673  |
| 35     | 0.00100600 | 146.62 | 146.72 | 0.50510  |
| 40     | 0.00100785 | 167.52 | 167.62 | 0.57237  |
| 45     | 0.00100988 | 188.41 | 188.51 | 0.63858  |
| 50     | 0.00101211 | 209.32 | 209.42 | 0.70377  |
| 55     | 0.00101452 | 230.23 | 230.33 | 0.76798  |
| 60     | 0.00101709 | 251.15 | 251.25 | 0.83125  |
| 65     | 0.00101984 | 272.08 | 272.18 | 0.89361  |
| 70     | 0.00102274 | 293.02 | 293.12 | 0.95509  |
| 75     | 0.00102581 | 313.98 | 314.08 | 1.0157   |
| 80     | 0.00102903 | 334.95 | 335.05 | 1.0755   |
| 85     | 0.00103241 | 355.95 | 356.05 | 1.1346   |
| 90     | 0.00103594 | 376.96 | 377.06 | 1.1928   |
| 95     | 0.00103962 | 398.00 | 398.10 | 1.2504   |
| 99.606 | 0.00104315 | 417.40 | 417.50 | 1.3028   |
| 99.606 | 1.6939     | 2505.5 | 2674.9 | 7.3588   |
| 100    | 1.6959     | 2506.2 | 2675.8 | 7.3610   |
| 105    | 1.7204     | 2514.1 | 2686.1 | 7.3885   |
| 110    | 1.7447     | 2521.8 | 2696.3 | 7.4155   |
| 115    | 1.7690     | 2529.6 | 2706.5 | 7.4418   |
| 120    | 1.7932     | 2537.3 | 2716.6 | 7.4678   |
| 125    | 1.8172     | 2545.0 | 2726.7 | 7.4932   |
| 130    | 1.8412     | 2552.6 | 2736.7 | 7.5183   |
| 135    | 1.8652     | 2560.2 | 2746.7 | 7.5429   |
| 140    | 1.8891     | 2567.8 | 2756.7 | 7.5672   |
| 145    | 1.9129     | 2575.4 | 2766.7 | 7.5911   |
| 150    | 1.9367     | 2582.9 | 2776.6 | 7.6148   |
| 155    | 1.9604     | 2590.5 | 2786.5 | 7.6380   |
| 160    | 1.9841     | 2598.0 | 2796.4 | 7.6610   |
| 165    | 2.0077     | 2605.5 | 2806.3 | 7.6838   |
| 170    | 2.0313     | 2613.1 | 2816.2 | 7.7062   |
| 175    | 2.0549     | 2620.6 | 2826.1 | 7.7284   |
| 180    | 2.0785     | 2628.1 | 2836.0 | 7.7503   |
| 185    | 2.1020     | 2635.6 | 2845.8 | 7.7719   |
| 190    | 2.1255     | 2643.1 | 2855.7 | 7.7934   |
| 195    | 2.1490     | 2650.7 | 2865.6 | 7.8146   |
| 200    | 2.1724     | 2658.3 | 2875.5 | 7.8356   |
| 210    | 2.2193     | 2673.3 | 2895.2 | 7.8769   |
| 220    | 2.2661     | 2688.4 | 2915.0 | 7.9174   |
| 230    | 2.3128     | 2703.5 | 2934.8 | 7.9572   |
| 240    | 2.3595     | 2718.7 | 2954.6 | 7.9962   |
| 250    | 2.4062     | 2733.9 | 2974.5 | 8.0346   |
| 260    | 2.4528     | 2749.1 | 2994.4 | 8.0723   |
| 270    | 2.4993     | 2764.5 | 3014.4 | 8.1094   |

| $T$  | $v$    | $u$    | $h$    | $s$     |
|------|--------|--------|--------|---------|
| °C   | m³/kg  | kJ/kg  | kJ/kg  | kJ/kg K |
| 270  | 2.4993 | 2764.5 | 3014.4 | 8.1094  |
| 280  | 2.5459 | 2779.8 | 3034.4 | 8.1459  |
| 290  | 2.5924 | 2795.2 | 3054.4 | 8.1818  |
| 300  | 2.6388 | 2810.6 | 3074.5 | 8.2172  |
| 310  | 2.6853 | 2826.2 | 3094.7 | 8.2520  |
| 320  | 2.7317 | 2841.7 | 3114.9 | 8.2864  |
| 330  | 2.7782 | 2857.3 | 3135.1 | 8.3202  |
| 340  | 2.8246 | 2873.0 | 3155.5 | 8.3536  |
| 350  | 2.8710 | 2888.7 | 3175.8 | 8.3866  |
| 360  | 2.9173 | 2904.6 | 3196.3 | 8.4191  |
| 370  | 2.9637 | 2920.3 | 3216.7 | 8.4512  |
| 380  | 3.0100 | 2936.3 | 3237.3 | 8.4829  |
| 390  | 3.0564 | 2952.3 | 3257.9 | 8.5142  |
| 400  | 3.1027 | 2968.3 | 3278.6 | 8.5452  |
| 410  | 3.1490 | 2984.4 | 3299.3 | 8.5757  |
| 420  | 3.1953 | 3000.6 | 3320.1 | 8.6059  |
| 430  | 3.2416 | 3016.7 | 3340.9 | 8.6358  |
| 440  | 3.2879 | 3033.1 | 3361.9 | 8.6653  |
| 450  | 3.3342 | 3049.4 | 3382.8 | 8.6946  |
| 460  | 3.3805 | 3065.8 | 3403.9 | 8.7235  |
| 470  | 3.4267 | 3082.3 | 3425.0 | 8.7521  |
| 480  | 3.4730 | 3098.9 | 3446.2 | 8.7804  |
| 490  | 3.5193 | 3115.5 | 3467.4 | 8.8084  |
| 500  | 3.5655 | 3132.1 | 3488.7 | 8.8361  |
| 520  | 3.6580 | 3165.8 | 3531.6 | 8.8908  |
| 540  | 3.7505 | 3199.6 | 3574.7 | 8.9445  |
| 560  | 3.8430 | 3233.7 | 3618.0 | 8.9972  |
| 580  | 3.9354 | 3268.2 | 3661.7 | 9.0489  |
| 600  | 4.0279 | 3302.8 | 3705.6 | 9.0998  |
| 620  | 4.1203 | 3337.8 | 3749.8 | 9.1499  |
| 640  | 4.2127 | 3373.0 | 3794.3 | 9.1991  |
| 660  | 4.3052 | 3408.5 | 3839.0 | 9.2476  |
| 680  | 4.3976 | 3444.2 | 3884.0 | 9.2954  |
| 700  | 4.4900 | 3480.4 | 3929.4 | 9.3424  |
| 720  | 4.5824 | 3516.8 | 3975.0 | 9.3888  |
| 740  | 4.6747 | 3553.4 | 4020.9 | 9.4345  |
| 760  | 4.7671 | 3590.3 | 4067.0 | 9.4797  |
| 780  | 4.8595 | 3627.6 | 4113.5 | 9.5242  |
| 800  | 4.9519 | 3665.0 | 4160.2 | 9.5681  |
| 820  | 5.0443 | 3702.8 | 4207.2 | 9.6115  |
| 840  | 5.1366 | 3740.8 | 4254.5 | 9.6544  |
| 860  | 5.2290 | 3779.2 | 4302.1 | 9.6968  |
| 880  | 5.3213 | 3817.8 | 4349.9 | 9.7386  |
| 900  | 5.4137 | 3856.6 | 4398.0 | 9.7800  |
| 920  | 5.5061 | 3895.8 | 4446.4 | 9.8209  |
| 940  | 5.5984 | 3935.2 | 4495.0 | 9.8613  |
| 960  | 5.6908 | 3974.8 | 4543.9 | 9.9013  |
| 980  | 5.7831 | 4014.8 | 4593.1 | 9.9408  |
| 1000 | 5.8754 | 4055.1 | 4642.6 | 9.9800  |

Water/Steam at  $p = 3.0 \text{ MPa}$  ( $T_{\text{sat}} = 233.853^\circ\text{C}$ )

| $T$     | $v$                    | $u$    | $h$    | $s$     | $T$  | $v$                    | $u$    | $h$    | $s$     |
|---------|------------------------|--------|--------|---------|------|------------------------|--------|--------|---------|
| °C      | $\text{m}^3/\text{kg}$ | kJ/kg  | kJ/kg  | kJ/kg K | °C   | $\text{m}^3/\text{kg}$ | kJ/kg  | kJ/kg  | kJ/kg K |
| 0       | 0.00099869             | 0.01   | 3.01   | 0.00003 | 270  | 0.0750660              | 2689.7 | 2914.9 | 6.3987  |
| 5       | 0.00099861             | 21.00  | 24.00  | 0.07619 | 280  | 0.0771620              | 2710.7 | 2942.2 | 6.4486  |
| 10      | 0.00099892             | 41.94  | 44.94  | 0.15081 | 290  | 0.0791960              | 2731.0 | 2968.6 | 6.4959  |
| 15      | 0.00099955             | 62.85  | 65.85  | 0.22400 | 300  | 0.0811790              | 2750.8 | 2994.3 | 6.5412  |
| 20      | 0.00100047             | 83.73  | 86.73  | 0.29586 | 310  | 0.0831190              | 2770.1 | 3019.5 | 6.5847  |
| 25      | 0.00100165             | 104.60 | 107.60 | 0.36645 | 320  | 0.0850220              | 2789.1 | 3044.2 | 6.6266  |
| 30      | 0.00100307             | 125.45 | 128.46 | 0.43584 | 330  | 0.0868930              | 2807.7 | 3068.4 | 6.6672  |
| 35      | 0.00100471             | 146.31 | 149.32 | 0.50409 | 340  | 0.0887370              | 2826.2 | 3092.4 | 6.7066  |
| 40      | 0.00100656             | 167.16 | 170.18 | 0.57124 | 350  | 0.0905560              | 2844.4 | 3116.1 | 6.7449  |
| 45      | 0.00100860             | 188.02 | 191.05 | 0.63734 | 360  | 0.0923550              | 2862.4 | 3139.5 | 6.7823  |
| 50      | 0.00101082             | 208.89 | 211.92 | 0.70243 | 370  | 0.0941340              | 2880.4 | 3162.8 | 6.8187  |
| 55      | 0.00101322             | 229.75 | 232.79 | 0.76654 | 380  | 0.0958970              | 2898.2 | 3185.9 | 6.8544  |
| 60      | 0.00101579             | 250.63 | 253.68 | 0.82971 | 390  | 0.0976450              | 2915.9 | 3208.8 | 6.8892  |
| 65      | 0.00101852             | 271.52 | 274.58 | 0.89198 | 400  | 0.0993790              | 2933.6 | 3231.7 | 6.9234  |
| 70      | 0.00102141             | 292.43 | 295.49 | 0.95336 | 410  | 0.10110                | 2951.1 | 3254.4 | 6.9570  |
| 75      | 0.00102446             | 313.35 | 316.42 | 1.0139  | 420  | 0.10281                | 2968.7 | 3277.1 | 6.9900  |
| 80      | 0.00102766             | 334.28 | 337.36 | 1.0736  | 430  | 0.10451                | 2986.2 | 3299.7 | 7.0224  |
| 85      | 0.00103101             | 355.23 | 358.32 | 1.1326  | 440  | 0.10620                | 3003.7 | 3322.3 | 7.0542  |
| 90      | 0.00103452             | 376.21 | 379.31 | 1.1908  | 450  | 0.10789                | 3021.1 | 3344.8 | 7.0856  |
| 95      | 0.00103818             | 397.20 | 400.31 | 1.2482  | 460  | 0.10956                | 3038.6 | 3367.3 | 7.1165  |
| 100     | 0.00104199             | 418.21 | 421.34 | 1.3050  | 470  | 0.11123                | 3056.1 | 3389.8 | 7.1470  |
| 105     | 0.00104595             | 439.26 | 442.40 | 1.3610  | 480  | 0.11289                | 3073.6 | 3412.3 | 7.1770  |
| 110     | 0.00105006             | 460.35 | 463.50 | 1.4164  | 490  | 0.11455                | 3091.1 | 3434.8 | 7.2066  |
| 115     | 0.00105433             | 481.46 | 484.62 | 1.4712  | 500  | 0.11620                | 3108.6 | 3457.2 | 7.2359  |
| 120     | 0.00105876             | 502.60 | 505.78 | 1.5254  | 520  | 0.11948                | 3143.8 | 3502.2 | 7.2933  |
| 125     | 0.00106334             | 523.80 | 526.99 | 1.5790  | 540  | 0.12274                | 3179.0 | 3547.2 | 7.3493  |
| 130     | 0.00106809             | 545.03 | 548.23 | 1.6320  | 560  | 0.12599                | 3214.3 | 3592.3 | 7.4041  |
| 135     | 0.00107301             | 566.31 | 569.53 | 1.6845  | 580  | 0.12922                | 3249.8 | 3637.5 | 7.4577  |
| 140     | 0.00107810             | 587.64 | 590.87 | 1.7365  | 600  | 0.13245                | 3285.4 | 3682.8 | 7.5103  |
| 145     | 0.00108336             | 609.03 | 612.28 | 1.7880  | 620  | 0.13566                | 3321.3 | 3728.3 | 7.5618  |
| 150     | 0.00108881             | 630.47 | 633.74 | 1.8390  | 640  | 0.13886                | 3357.4 | 3774.0 | 7.6124  |
| 155     | 0.00109444             | 651.99 | 655.27 | 1.8896  | 660  | 0.14205                | 3393.8 | 3819.9 | 7.6621  |
| 160     | 0.00110027             | 673.57 | 676.87 | 1.9397  | 680  | 0.14523                | 3430.3 | 3866.0 | 7.7109  |
| 165     | 0.00110630             | 695.22 | 698.54 | 1.9895  | 700  | 0.14841                | 3467.0 | 3912.2 | 7.7590  |
| 170     | 0.00111255             | 716.95 | 720.29 | 2.0388  | 720  | 0.15157                | 3504.0 | 3958.7 | 7.8062  |
| 175     | 0.00111901             | 738.77 | 742.13 | 2.0878  | 740  | 0.15474                | 3541.2 | 4005.4 | 7.8528  |
| 180     | 0.00112571             | 760.68 | 764.06 | 2.1365  | 760  | 0.15790                | 3578.7 | 4052.4 | 7.8987  |
| 185     | 0.00113265             | 782.69 | 786.09 | 2.1849  | 780  | 0.16105                | 3616.4 | 4099.5 | 7.9439  |
| 190     | 0.00113984             | 804.81 | 808.23 | 2.2329  | 800  | 0.16420                | 3654.3 | 4146.9 | 7.9885  |
| 195     | 0.00114731             | 827.04 | 830.48 | 2.2807  | 820  | 0.16734                | 3692.6 | 4194.6 | 8.0325  |
| 200     | 0.00115506             | 849.39 | 852.86 | 2.3282  | 840  | 0.17048                | 3731.0 | 4242.4 | 8.0759  |
| 210     | 0.00117149             | 894.50 | 898.01 | 2.4227  | 860  | 0.17362                | 3769.6 | 4290.5 | 8.1187  |
| 220     | 0.00118931             | 940.19 | 943.76 | 2.5164  | 880  | 0.17675                | 3808.6 | 4338.9 | 8.1610  |
| 230     | 0.00120873             | 986.60 | 990.23 | 2.6097  | 900  | 0.17988                | 3847.9 | 4387.5 | 8.2028  |
| 233.853 | 0.00121669             | 1004.6 | 1008.3 | 2.6455  | 920  | 0.18301                | 3887.3 | 4436.3 | 8.2441  |
| 233.853 | 0.06666440             | 2603.2 | 2803.2 | 6.1856  | 940  | 0.18613                | 3927.0 | 4485.4 | 8.2849  |
| 240     | 0.0682300              | 2619.8 | 2824.5 | 6.2274  | 960  | 0.18925                | 3967.0 | 4534.8 | 8.3252  |
| 250     | 0.0706270              | 2644.6 | 2856.5 | 6.2893  | 980  | 0.19237                | 4007.2 | 4584.3 | 8.3651  |
| 260     | 0.0728950              | 2667.7 | 2886.4 | 6.3459  | 1000 | 0.19549                | 4047.6 | 4634.1 | 8.4045  |
| 270     | 0.0750660              | 2689.7 | 2914.9 | 6.3987  |      |                        |        |        |         |

[12 marks]



(12)

Max work obtainable.

$$u_1 = 2750.8, s_1 = 6.5412$$

$$u_2 = 2810.6, s_2 = 8.2172$$

$$w_{\max} = (u_2 - u_1) - T_0(s_2 - s_1)$$

$$= (2810.6 - 2750.8) - 573(8.2172 - 6.5412)$$

$$= 900.548 \text{ kJ/kg} = \underline{\quad}$$

$$\text{work} = \frac{180.1096 \text{ kJ}}{\text{Ans}}$$

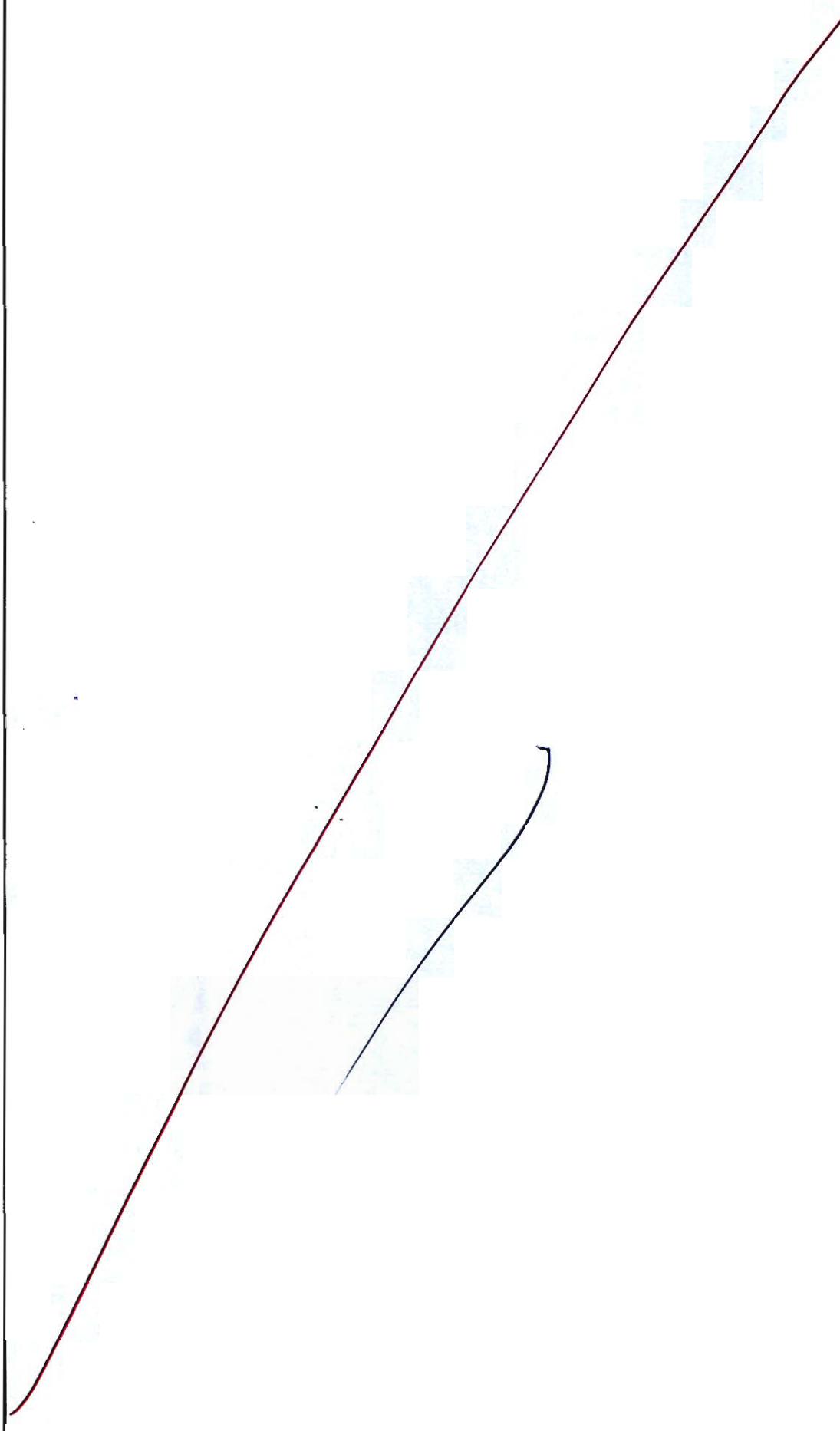


Q.6 (a)

In an aqua-ammonia absorption refrigeration system of 10 tonnes refrigeration capacity the vapour leaving the generator are 100% pure  $\text{NH}_3$  saturated at  $40^\circ\text{C}$ . The evaporator, absorber, condenser and generator temperatures are  $-20^\circ\text{C}$ ,  $30^\circ\text{C}$ ,  $40^\circ\text{C}$  and  $170^\circ\text{C}$  respectively. At absorber exit (strong solution) the concentration of ammonia in solution is 0.4 and enthalpy of 25 kJ/kg. At generator exit (weak solution) the concentration of ammonia in solution is 0.2 and enthalpy of 700 kJ/kg. The enthalpy of saturated liquid and saturated vapour ammonia at  $40^\circ\text{C}$  are 371.9 kJ/kg and 1473.3 kJ/kg respectively. The enthalpy of saturated vapour ammonia from evaporator exit at  $-20^\circ\text{C}$  is 1420 kJ/kg.

- (i) Determine the mass flow rate of ammonia in the evaporator.
- (ii) Carry out overall mass conservation and mass conservation of ammonia in absorber to determine mass flow rates of weak and strong solution.
- (iii) Determine the heat rejection in absorber and condenser, heat added in generator and C.O.P.

[20 marks]

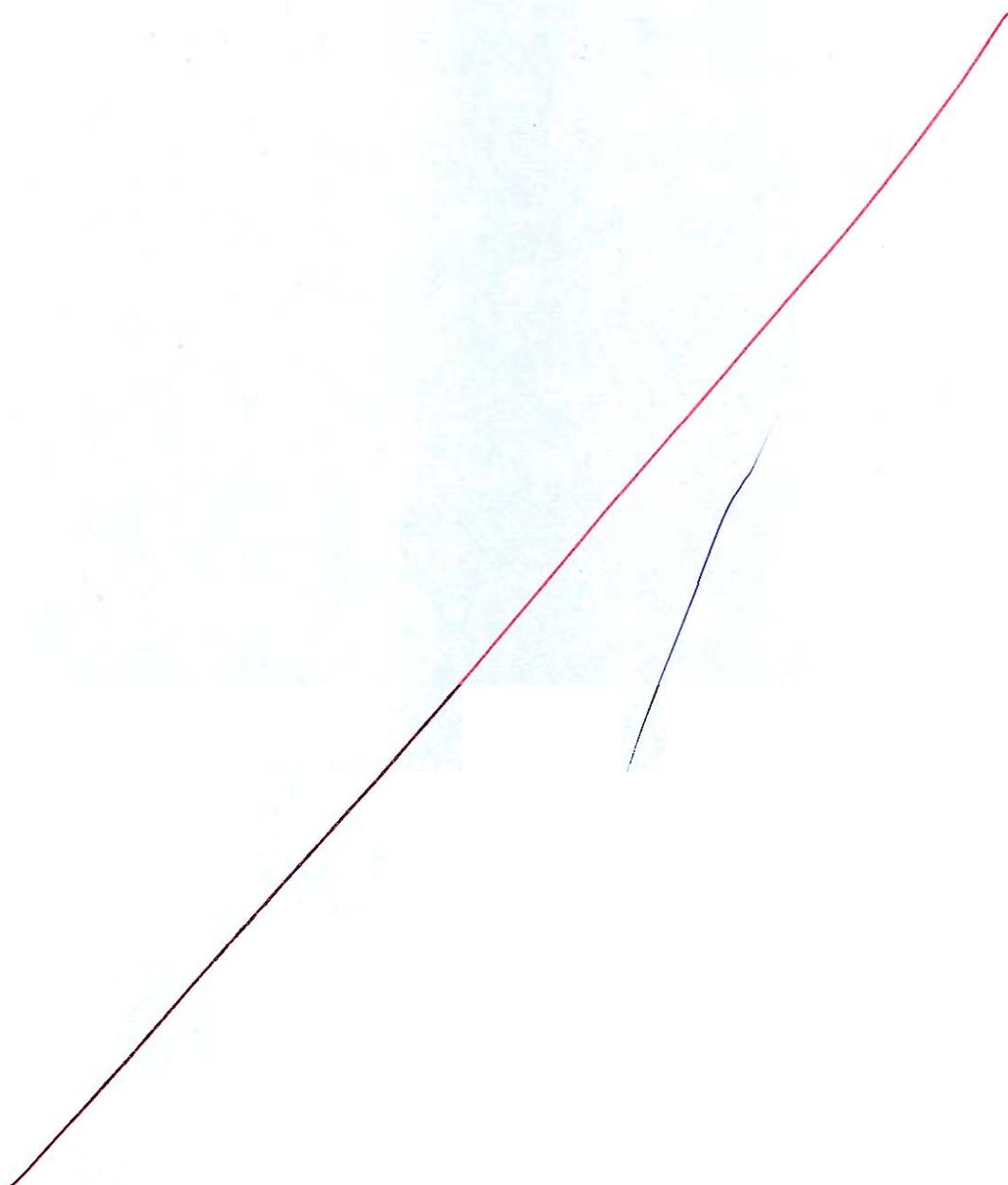


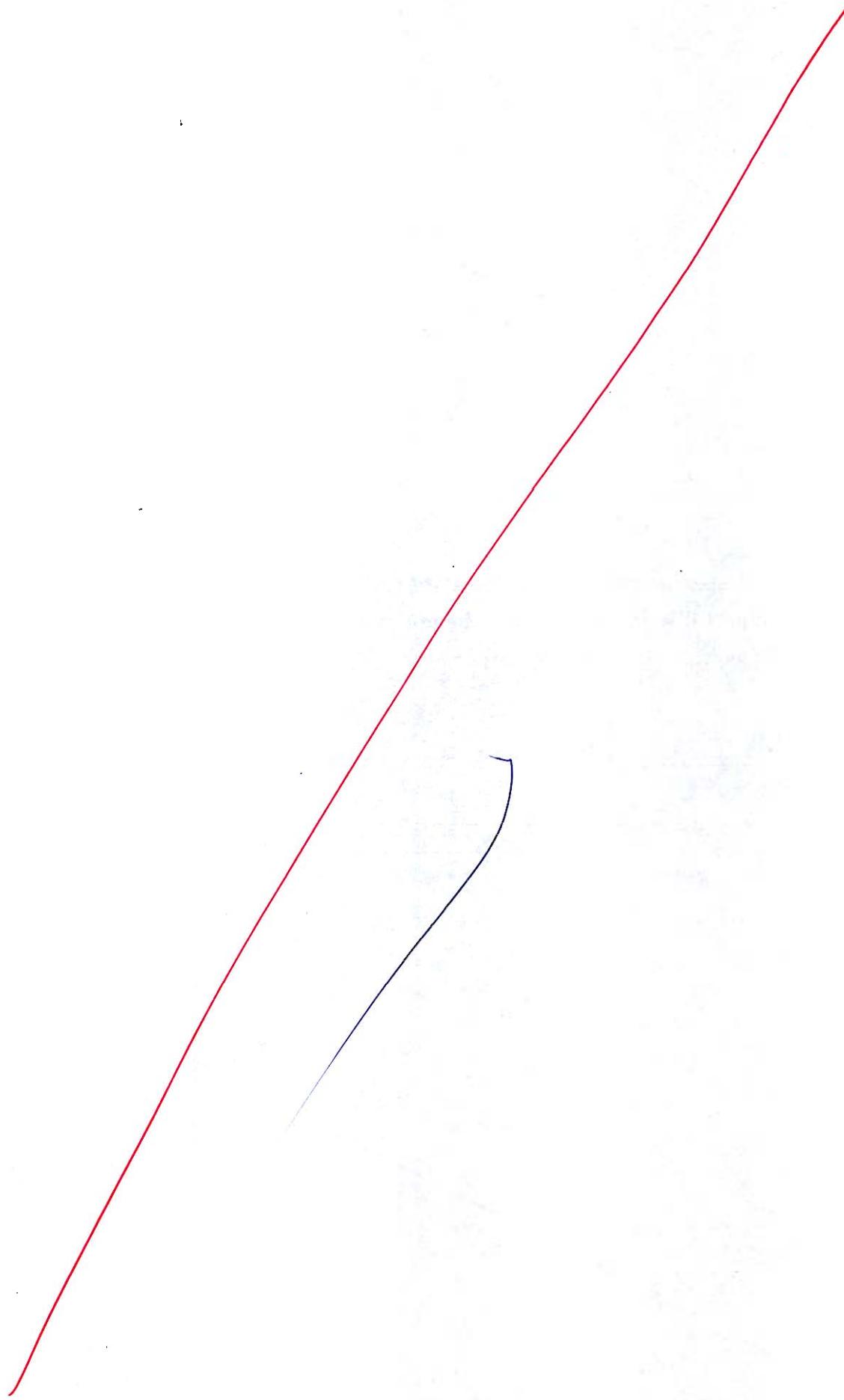
Q.6 (b)

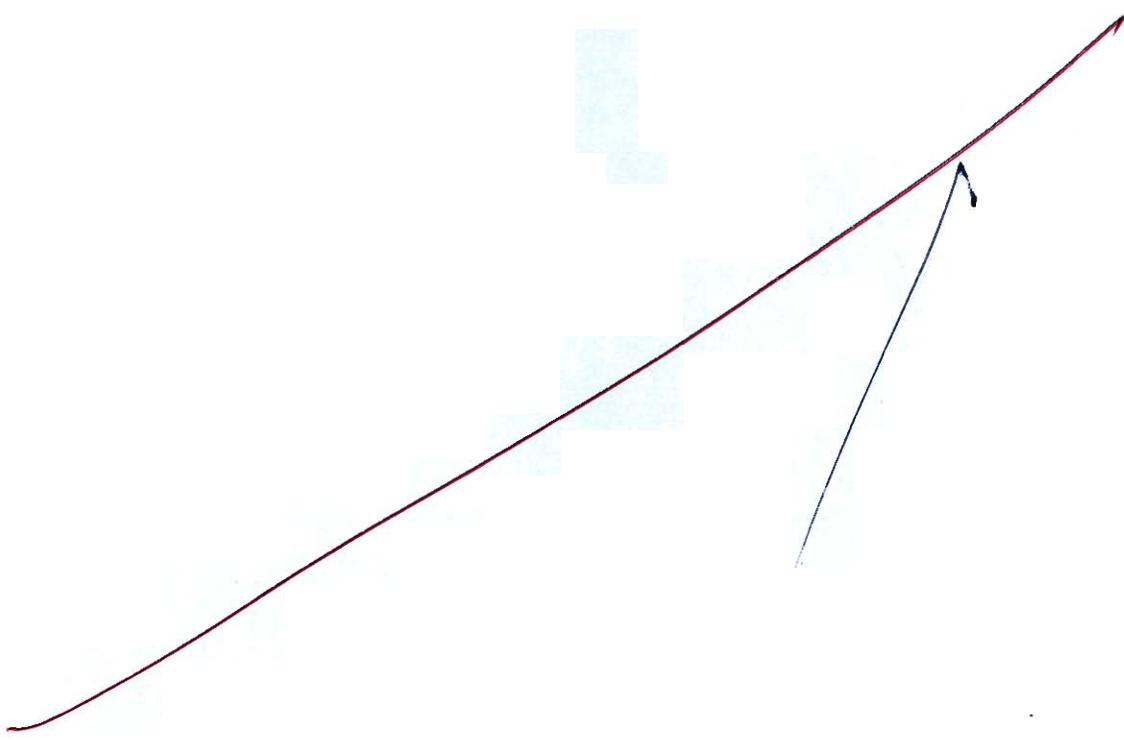
An air preheater is used to heat up the air used for combustion by cooling the outgoing products of combustion from  $600^{\circ}\text{C}$  to  $450^{\circ}\text{C}$ . The rate of flow of the products is  $12 \text{ kg/sec}$ , and the products are cooled from  $600^{\circ}\text{C}$  to  $450^{\circ}\text{C}$  and the specific heat at constant pressure for products and for air are  $1.09 \text{ kJ/kgK}$  and  $1.005 \text{ kJ/kgK}$  respectively. The rate of air flow is  $8 \text{ kg/sec}$  and the initial air temperature is  $40^{\circ}\text{C}$ . Determine the following :

- the initial and final availability of the products.
- the irreversibility of the process.
- if the heat transfer from the products were to take place reversibly through heat engine, what would be the final temperature of the air and the power developed by heat engine? Take ambient temperature  $T_0 = 300 \text{ K}$ .

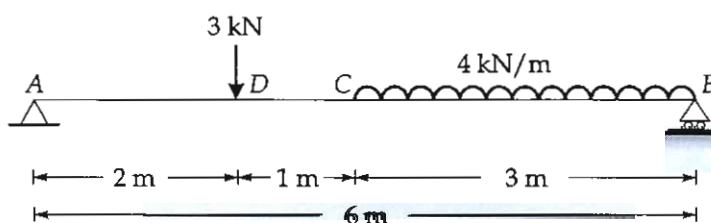
[20 marks]



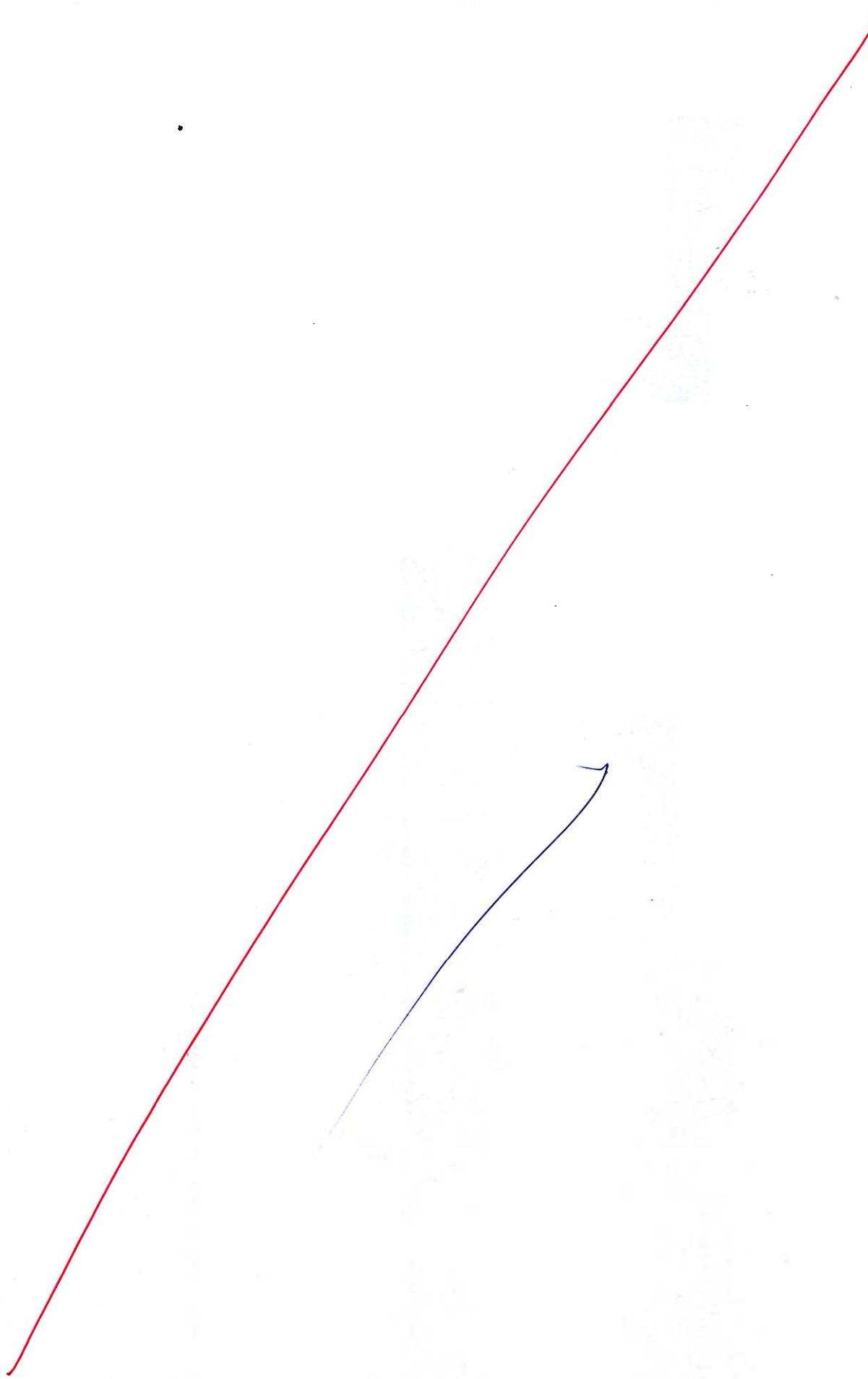


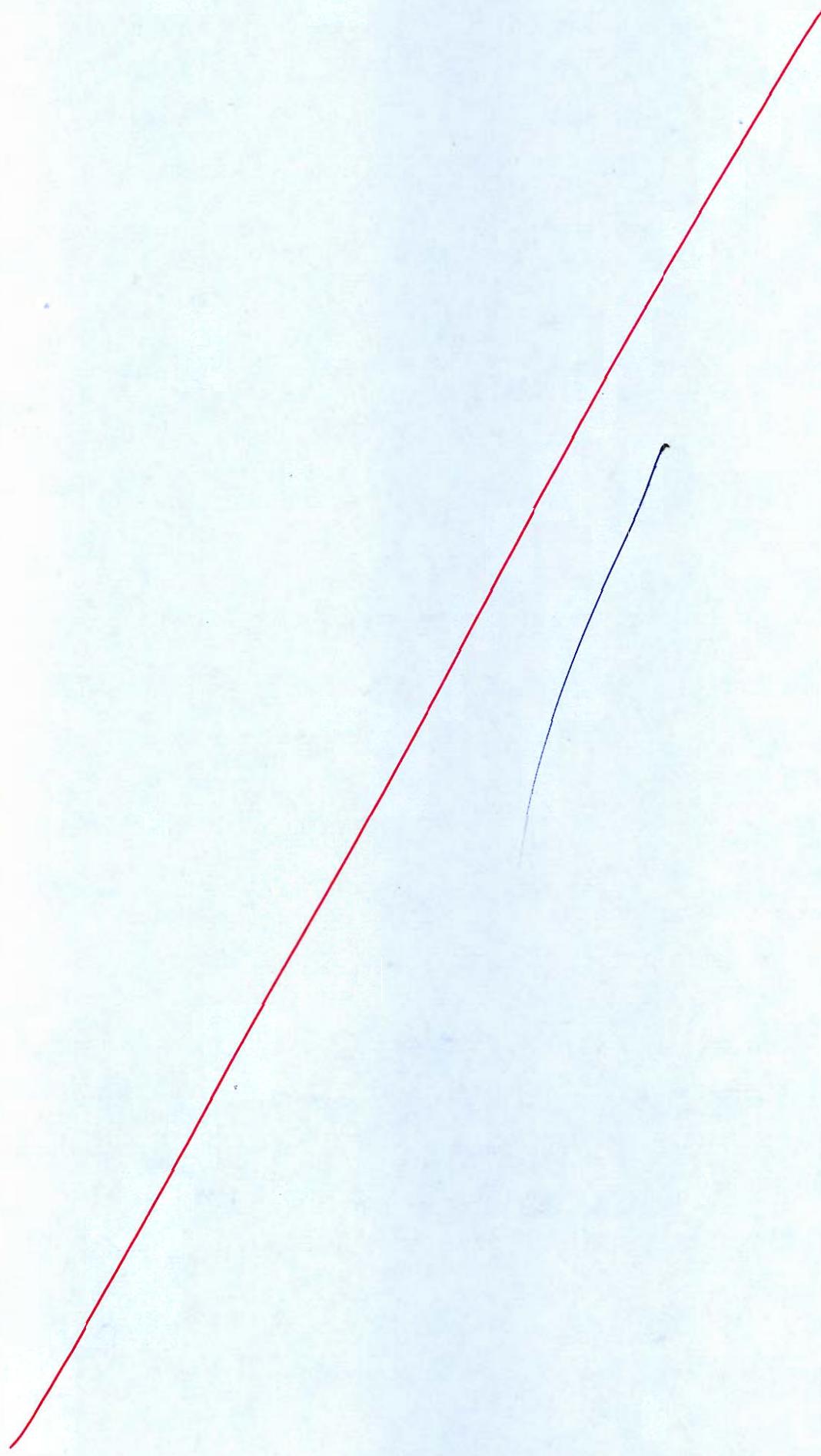


- Q.6 (c) A beam  $AB$  of 6 m span is simply supported at the ends and is loaded as shown in figure below. Determine : (i) Deflection at  $C$ , (ii) Maximum deflection and (iii) Slope at end  $A$ . Take  $E = 2 \times 10^5 \text{ N/mm}^2$  and  $I = 2500 \text{ cm}^4$ .



[20 marks]







.7 (a) An ammonia ice plant operates between condenser temperature of 35°C and an evaporator temperature of -15°C. It produces 10 tons of ice per day from water at 27°C to ice at -5°C. The NH<sub>3</sub> enters the compressor as dry saturated vapour and leaves the condenser as saturated liquid. Determine,

- (i) The capacity of the refrigerating plant.
- (ii) Mass flow rate of the refrigerant.
- (iii) Power of the compressor motor if the isentropic efficiency of the compressor is 82% and mechanical efficiency of the compressor is 94%.
- (iv) Relative efficiency

Take latent heat of ice = 335 kJ/kg

Specific heat of ice = 1.94 kJ/kgK

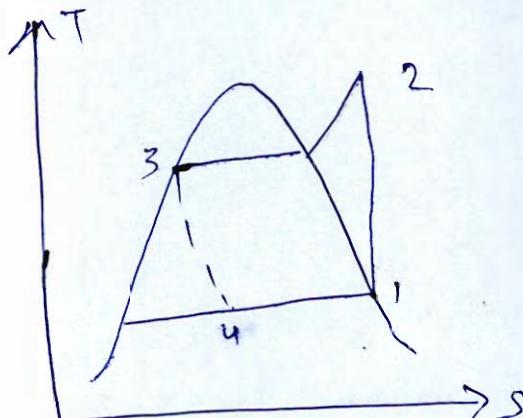
Specific heat of water = 4.2 kJ/kgK

Use the following properties of NH<sub>3</sub>



| Saturation | Enthalpy (kJ/kg) |       | Entropy kJ/kg-K |       | Specific heat (kJ/kg-K) |                  |
|------------|------------------|-------|-----------------|-------|-------------------------|------------------|
| Temp. (°C) | $h_f$            | $h_g$ | $s_f$           | $s_g$ | Liquid, $C_{pf}$        | Vapour, $C_{pg}$ |
| -15        | 112.3            | 1426  | 0.457           | 5.549 | —                       | —                |
| 35         | 347.5            | 1471  | 1.282           | 4.930 | 4.6                     | 2.8              |

[20 marks]



$RC = 10 \text{ ton ice/day}$   
from  $27^\circ\text{ water to}$   
 $-5^\circ\text{C ice}$

$$h_1 = 1426 \text{ kJ/kg}$$

$$h_3 = h_4 = 347.5 \text{ kJ/kg}$$

$$RC = \frac{10 \times 10^3 [4.2 \times 27 + 335 + 1.94 \times 5]}{24 \times 3600}$$

$$RC = 53.02083 \text{ kW} \quad \boxed{\text{Ans}} \quad \checkmark$$

$$s_1 = s_2$$

$$5.549 = 4.93 + 2.8 \ln \frac{T_2}{273+35=308}$$

$$T_2 = 384.203 \text{ K}$$

$$h_2 = 1471 + 2.8(384.203 - 308) = 1684.3685 \text{ kJ/kg}$$

$$\dot{m} = \frac{RC}{RE} = \frac{53.02083}{1426 - 347.5} = \boxed{0.04916 \text{ kg/s}} \quad \boxed{\text{Ans}} \quad \checkmark$$

Power of Compressor motor =

$$\frac{0.04916(1684.3685 - 1426)}{0.82 \times 0.94} \\ = \boxed{16.4787 \text{ kW}} \quad \boxed{\text{Ans}} \quad \checkmark$$

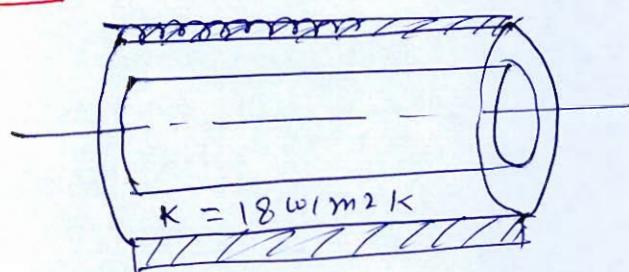
$$(iv) \eta_{\text{Relative}} = 0.82 \times 0.94 = \boxed{77.08\%} \quad \boxed{\text{Ans}}$$

Refer solution



Q.7 (b)

Calculate the temperature difference between outer and inner surface of a thin hollow tube having 4 mm and 6 mm inside and outside radius, respectively, when a current of 2.1 kA flows through it. For cooling the tube, the coolant water at temperature 20°C circulates inside the tube. The heat transfer coefficient of water side is 12500 W/m<sup>2</sup>K. The outer surface of the tube is insulated. The electrical resistivity of material is 0.11 Ωmm<sup>2</sup>/m. The thermal conductivity of the material is 18 W/m<sup>2</sup>K. Find the heat flow rate. Also obtain the expression for temperature difference between inner and outer radius.



[20 marks]

$$r_1 = 4 \text{ mm}$$

$$r_2 = 6 \text{ mm}$$

$$I = 2.1 \text{ kA}$$

$$T_{ci} = 20^\circ\text{C}, h_i = 12500 \frac{\text{W}}{\text{m}^2\text{K}}$$

$$\rho_e = 0.11 \Omega \text{mm}^2/\text{m}$$

find Q & Temp diff.

$$V = IR, R = \rho_e \cdot \frac{L}{A}$$

$$\rho = \frac{R A}{L}$$

$$R = \frac{0.11 \times (0.006^2 - 0.004^2)}{\pi} = 1750.704 \Omega$$

$$V = IR = 3.676479 \cdot 185 \text{ V}$$

$$Q = V \times I$$

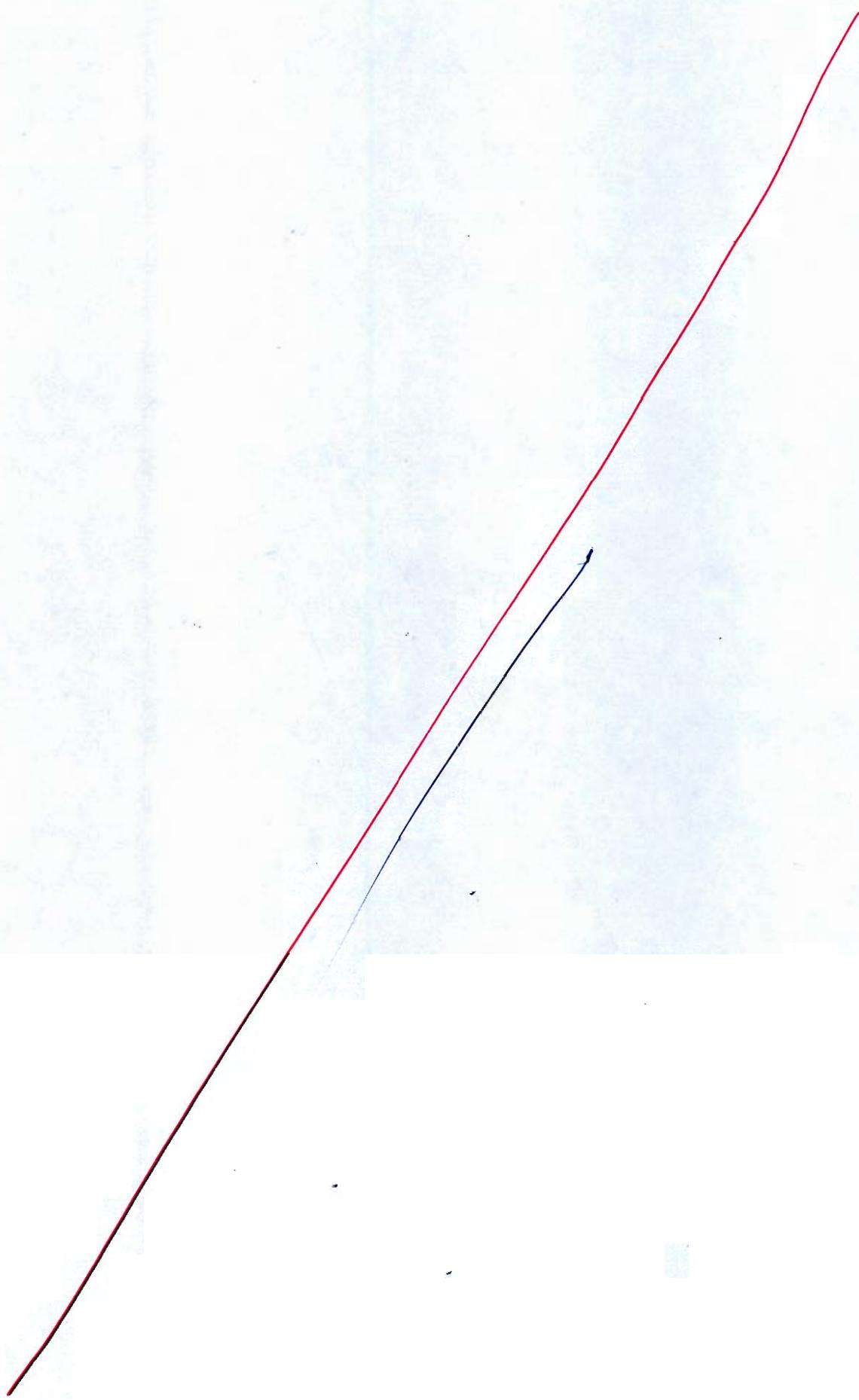
$$R = \frac{0.11}{\pi(6^2 - 4^2)} = 1.7507 \times 10^{-3} \Omega$$

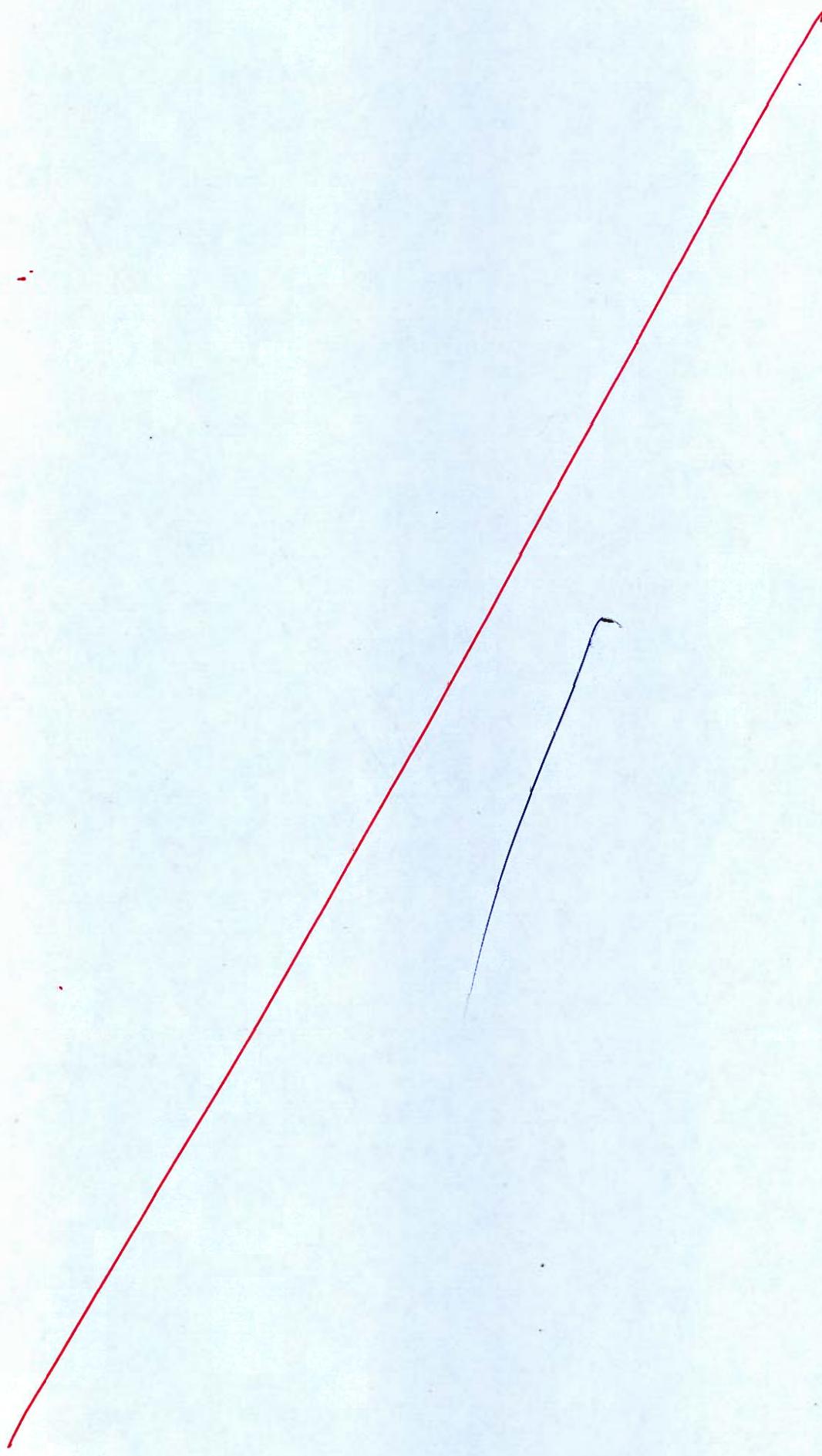
$$V = IR = 3.6764 \text{ V}$$

$$Q = V \times I = 7720.606 \text{ W}$$

7

Incomplete Solution



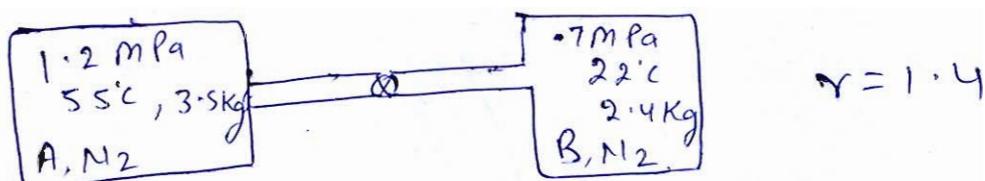


- 7 (c) Two vessels, A and B both containing nitrogen, are connected by a valve which is opened to allow the contents to mix and achieve an equilibrium temperature of 25°C. Before mixing the following information is known about the gases in the two vessels.

|            | Pressure | Temperature | Content |
|------------|----------|-------------|---------|
| Vessel 'A' | 1.2 MPa  | 55°C        | 3.5 kg  |
| Vessel 'B' | 0.7 MPa  | 22°C        | 2.4 kg  |

Calculate the final equilibrium pressure and the amount of heat transferred to the surroundings. If the vessel had been perfectly insulated. Calculate the final temperature and pressure which would have been reached. Take  $\gamma = 1.4$

[20 marks]



$$T_f = 25^\circ\text{C} = 298 \text{ K}$$

$$R_{N_2} = \frac{R}{m} = \frac{8.314}{28} = 0.2967 \text{ kJ/kg K}$$

$$V_A = \frac{m_A R T_A}{P_A} = \frac{3.5 \times 0.2967 \times 328}{1200} = 0.283843 \text{ m}^3$$

$$V_B = \frac{2.4 \times 0.2967 \times 295}{700} = 0.3 \text{ m}^3$$

$$V_{\text{total}} = 0.58393 \text{ m}^3$$

$$P_f V_f = m_f R T_f$$

$$P_f = \frac{(3.5 + 2.4) \times 0.2967 \times 298}{0.58393} = 893.35 \text{ kPa}$$

The tank are rigid, so process is constant volume.

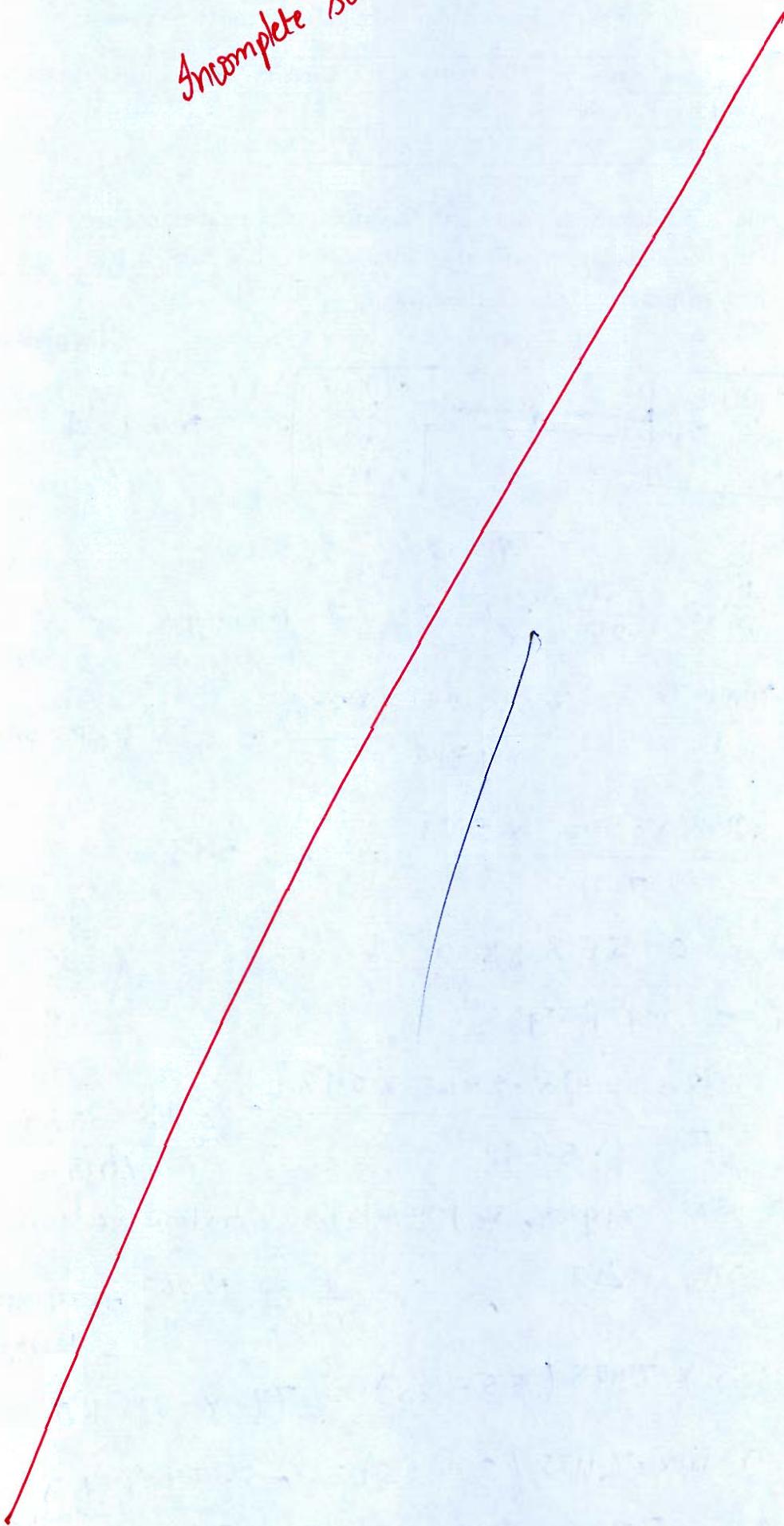
$$Q_A = m_A C_v \Delta T, \quad C_v = \frac{R}{\gamma - 1} = \frac{0.2967}{0.4} = 0.74175 \text{ kJ/kg K}$$

$$Q_A = 3.5 \times 0.74175 (55 - 25) = 77.88375 \text{ kJ}$$

$$Q_B = 2.4 \times 0.74175 (22 - 25) = -5.3406 \text{ kJ}$$

$$Q_{\text{net}} = 77.88375 - 5.3406 = 72.543 \text{ kJ}$$

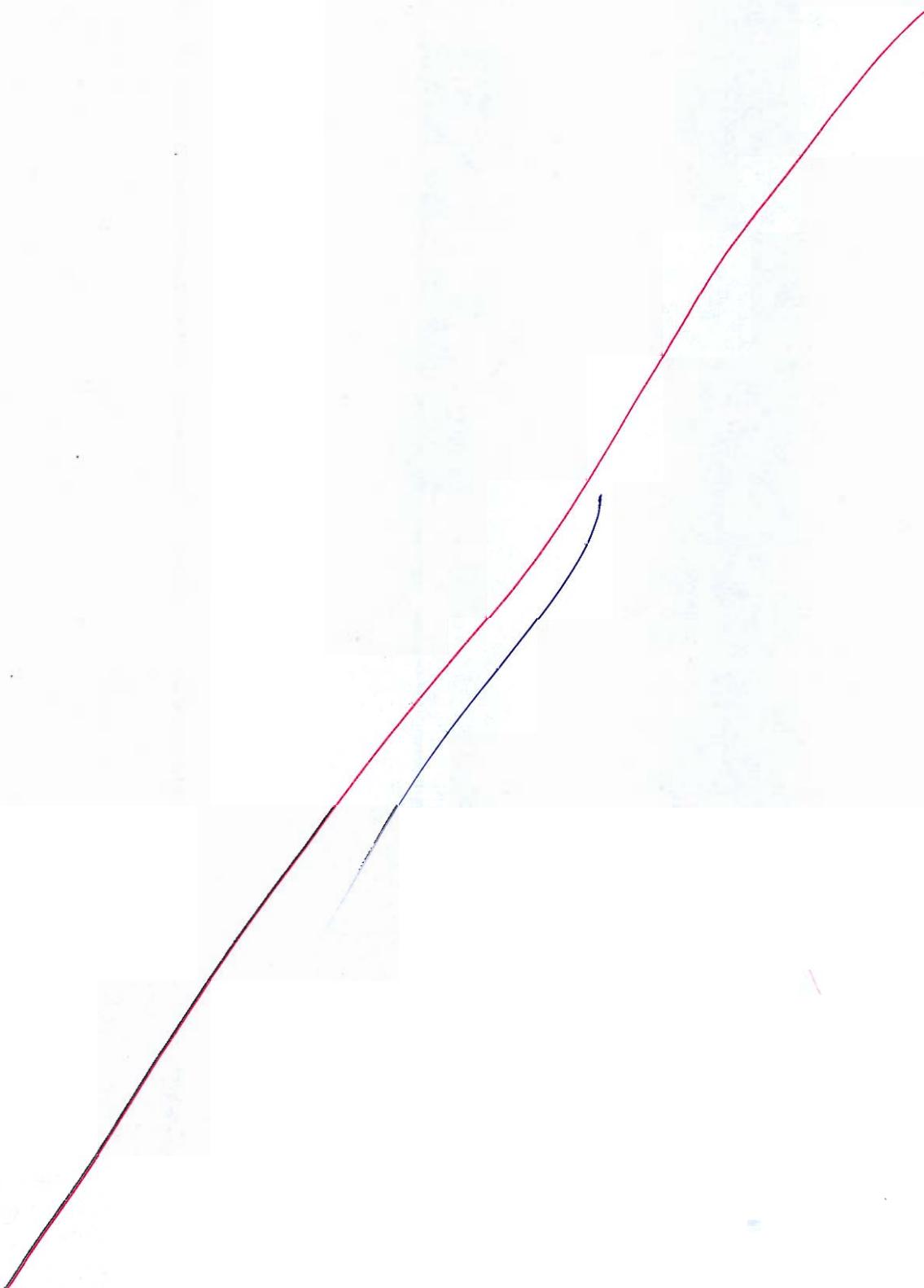
Incomplete Solution

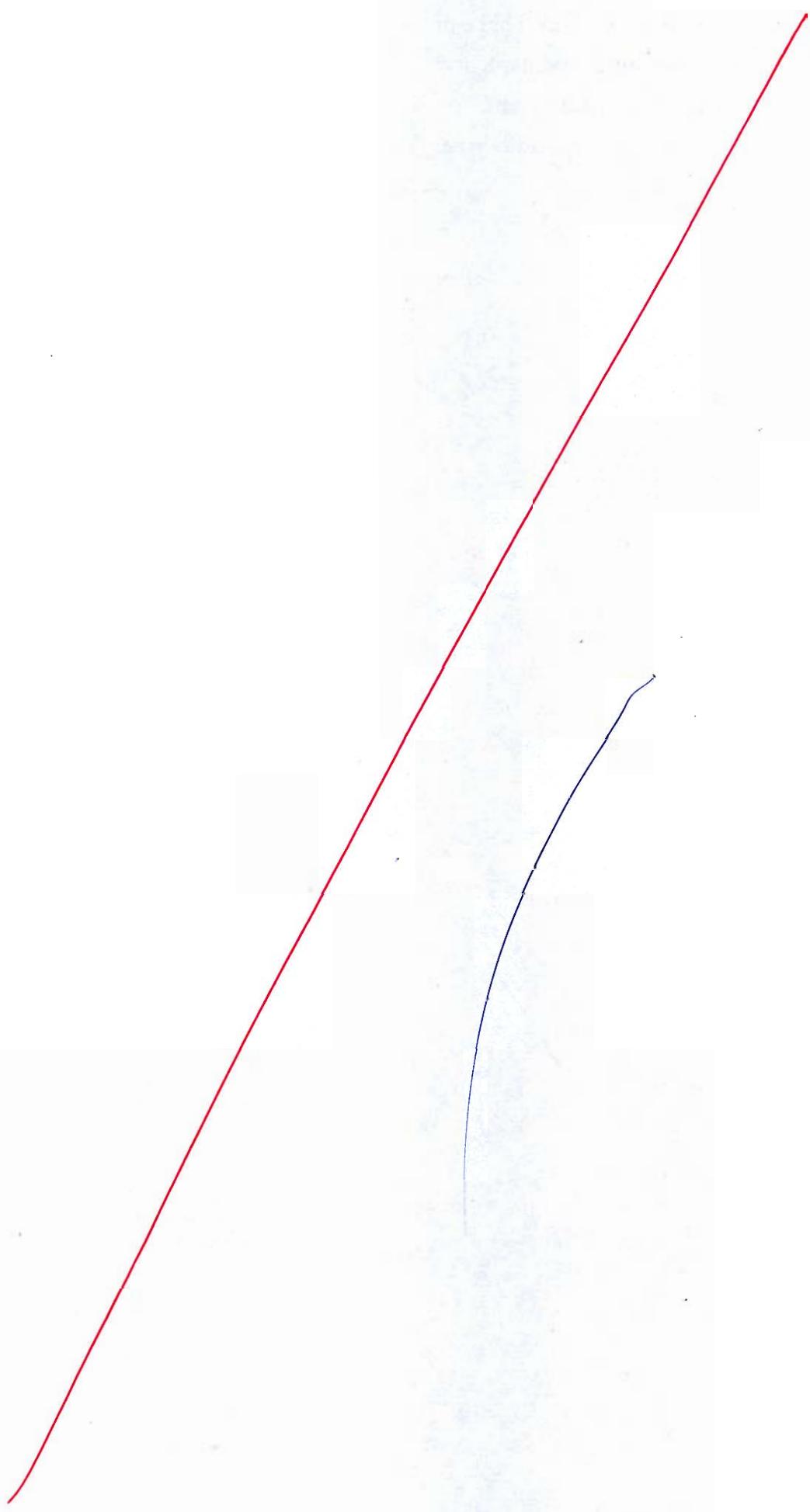


- 8 (a) A simply supported beam beam of span  $L$  carries a uniformly distributed load  $W$  per unit length on the whole span. Find the shape of beam of uniform strength, if
- (i) the width is to be maintained constant, and
  - (ii) the depth is to be maintained constant.

Also, determine depth in case (i) and width in case (ii) at the mid-span. Take permissible bending stress of beam as  $\sigma_{\text{per}}$ .

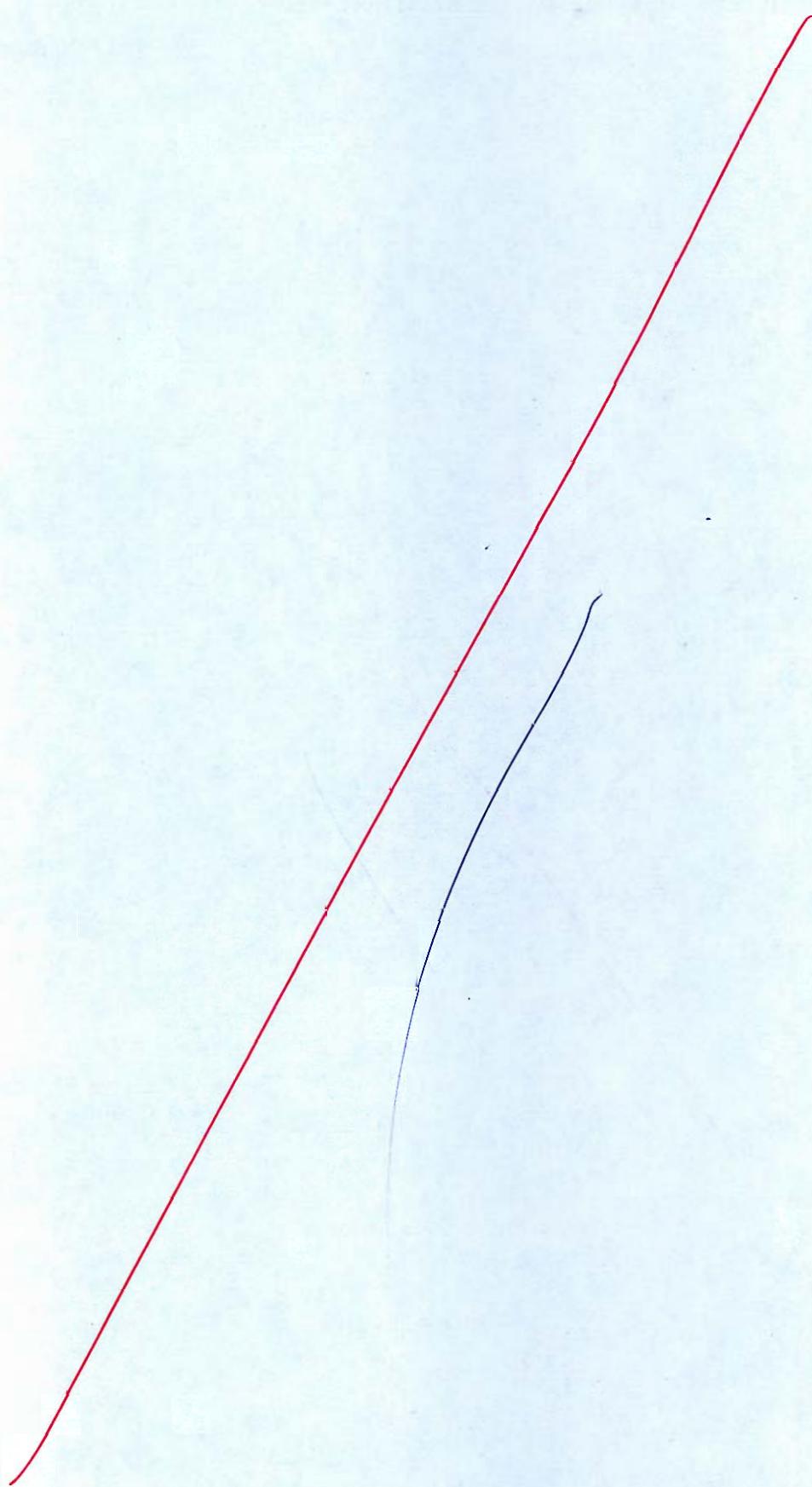
[20 marks]





- .8 (b) (i) Explain harmful effects of R-12 and R-22 refrigerant. Also suggest new eco-friendly substitutes of these two refrigerants.
- (ii) Define availability of a thermodynamic system. Also, derive the expression for availability function or availability for a non flow process.

[10 + 10 marks]

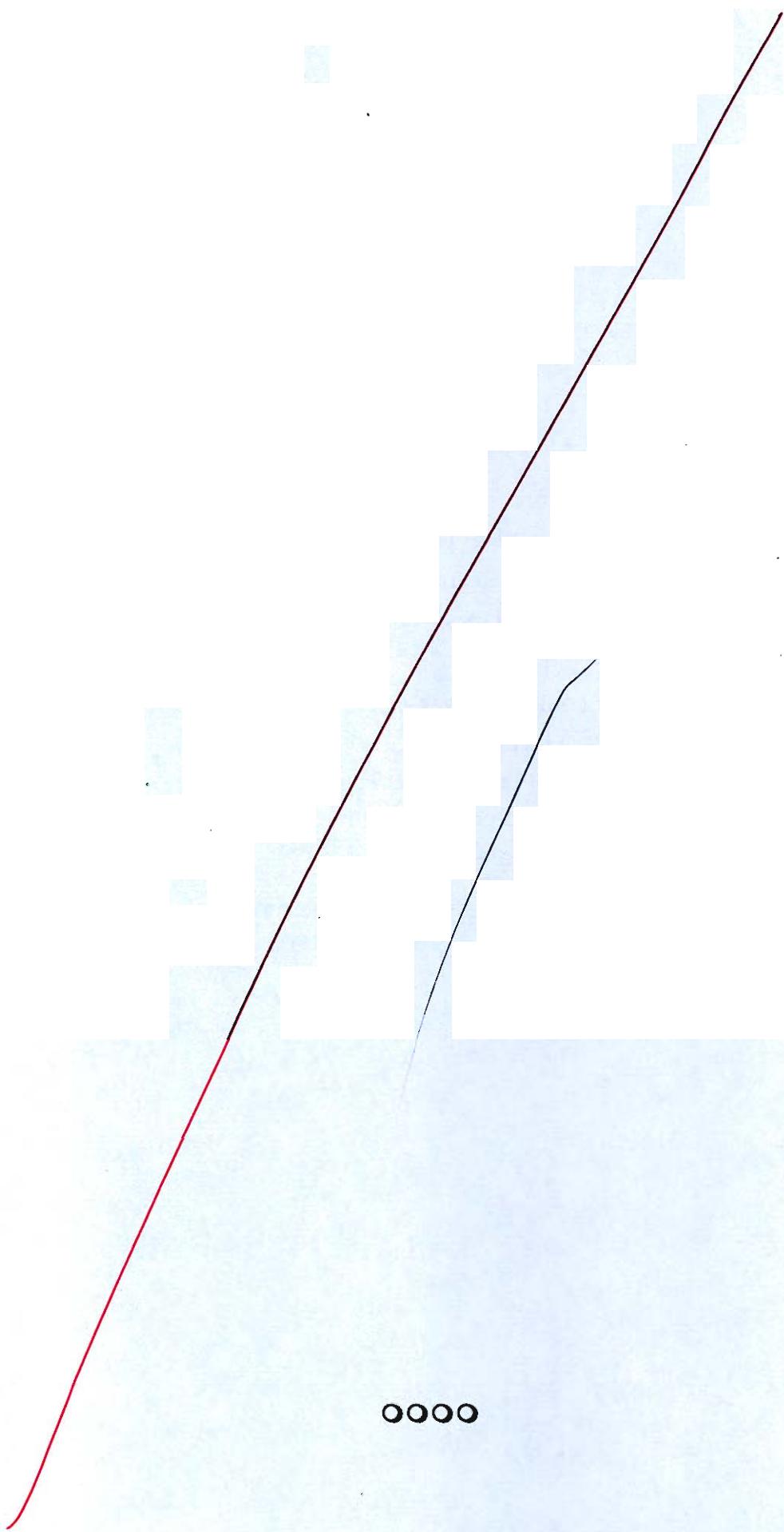


- 8 (c) Air at  $22^{\circ}\text{C}$  and at atmospheric pressure flows at a velocity of  $4.8 \text{ m/s}$  past a flat plate with a sharp leading edge. The entire plate is maintained at a temperature of  $58^{\circ}\text{C}$ . Assuming that transition occurs at critical Reynolds number of  $5 \times 10^5$ , find the distance from the leading edge at which the boundary layer changes from laminar to turbulent. At the location also calculate:

- (i) Thickness of hydrodynamic boundary layer
- (ii) Thickness of thermal boundary layer
- (iii) Local and average convective heat transfer coefficient
- (iv) Heat transfer from both side of plate per unit width of plate
- (v) Mass entrainment in the boundary layer

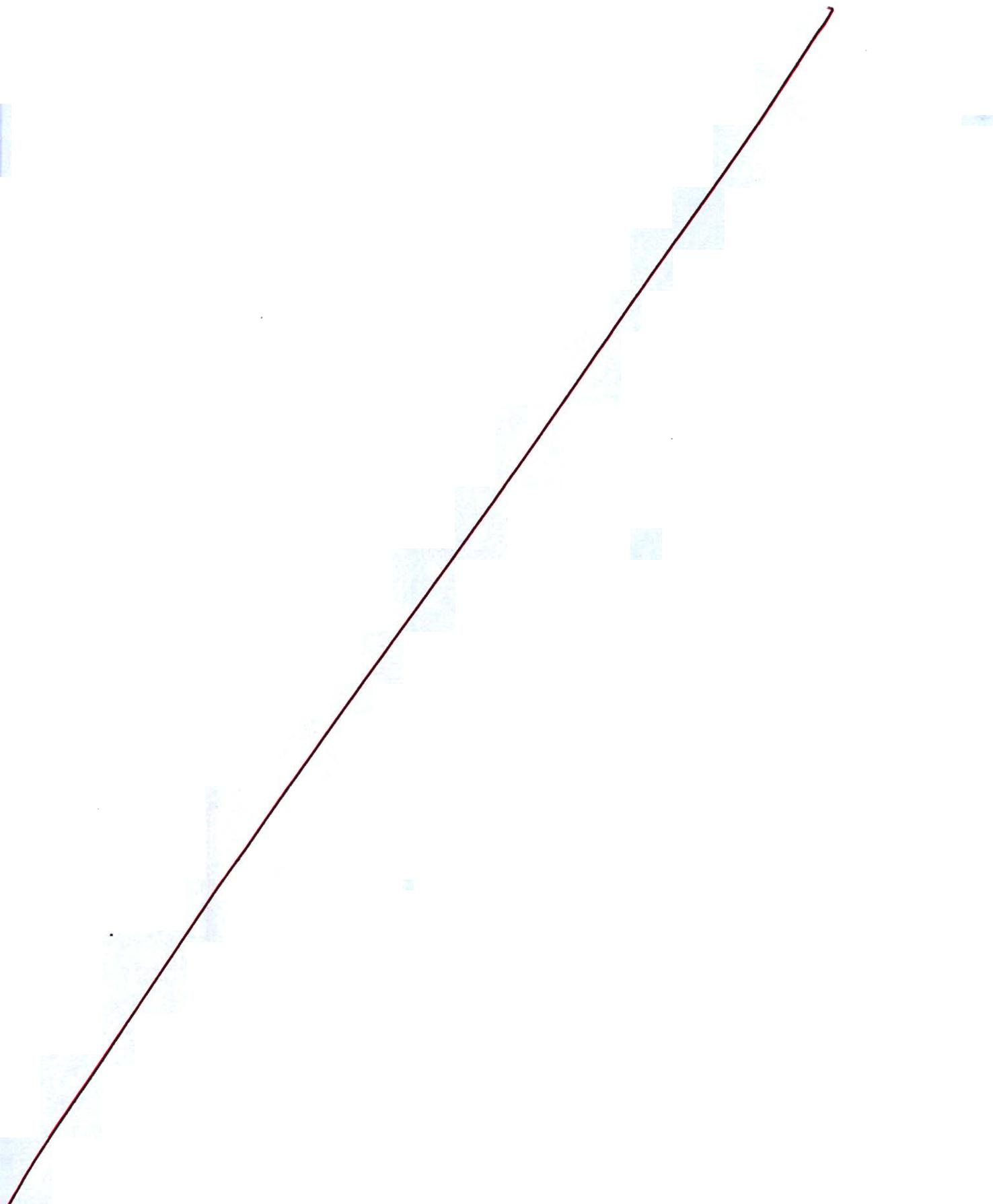
Assume cubic velocity profile and approximate method. Thermo-physical properties of air at mean film temperature of  $40^{\circ}\text{C}$  are:  $\rho = 1.128 \text{ kg/m}^3$ ;  $v = 16.96 \times 10^{-6} \text{ m}^2/\text{s}$ ;  $k = 0.02755 \text{ W/mK}$ ;  $\text{Pr} = 0.7$

[20 marks]



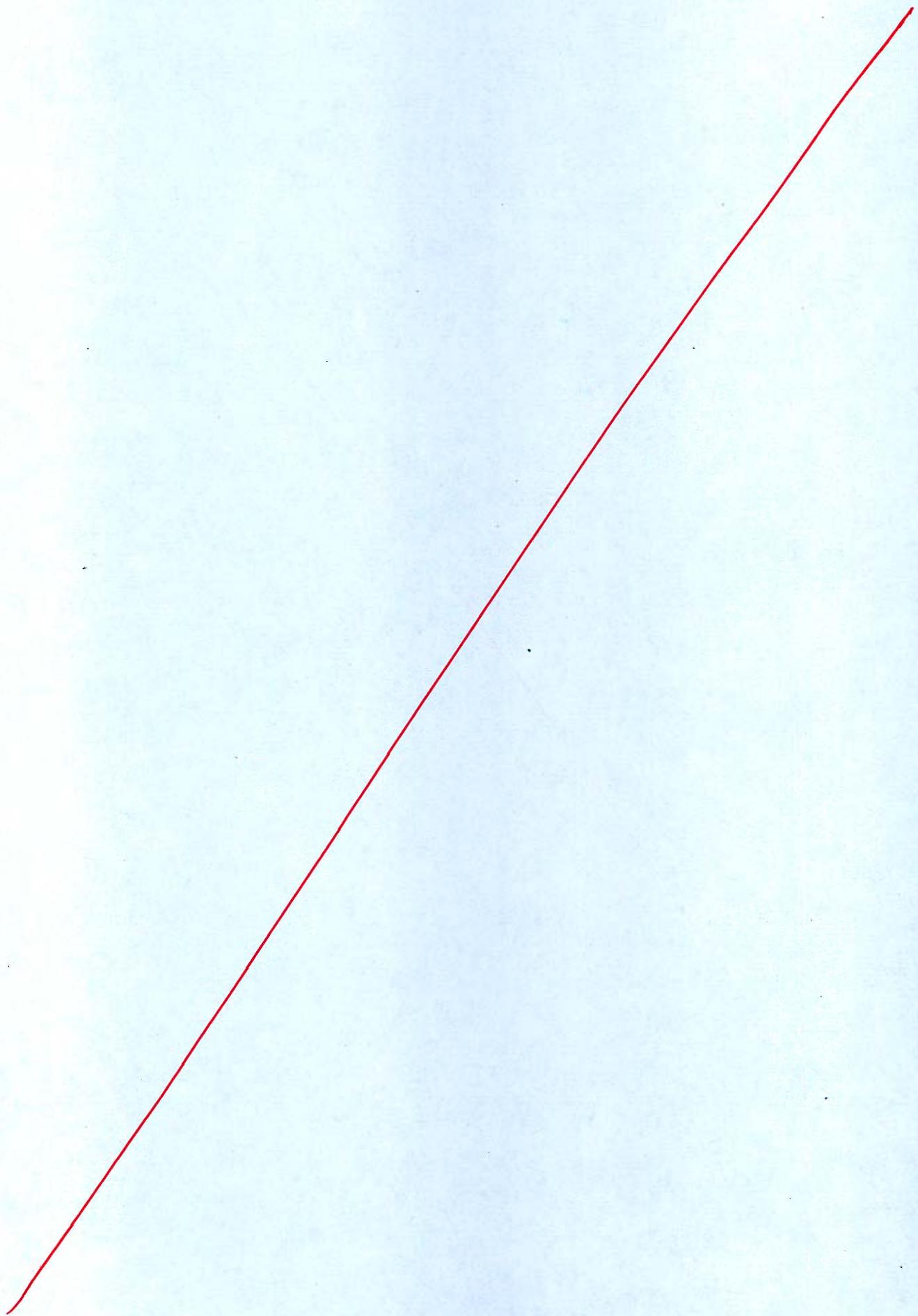
**Space for Rough Work**

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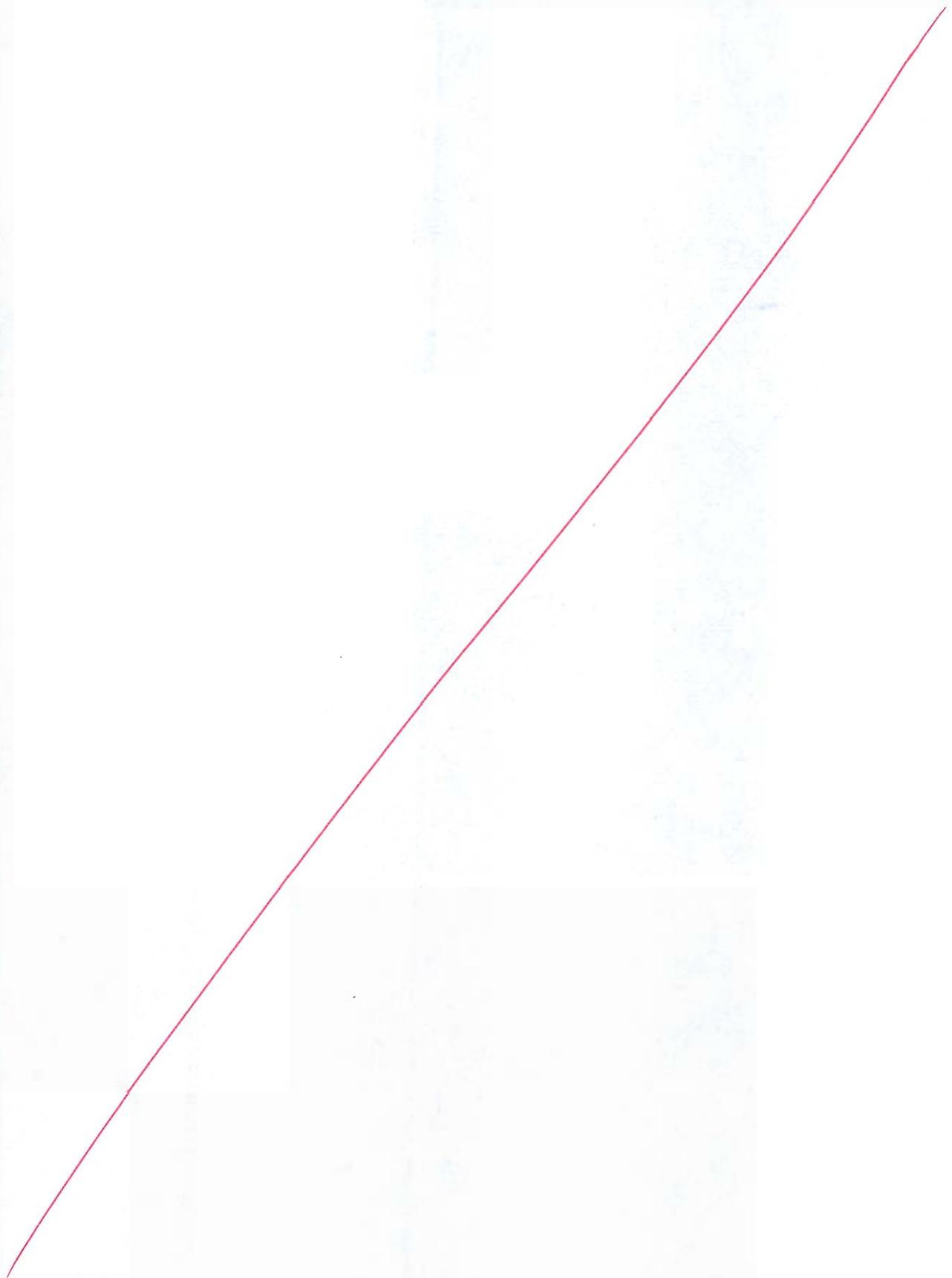
**Space for Rough Work**

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