CLASS TEST SL.: 01SPME_ABCD_0208202						82023			
EESE MADE EASE India's Best Institute for IES, GATE & PSUs									
Web: www.madeeasy.in E-mail: info@madeeasy.in Ph: 011-45124612									
INTERNAL COMBUSTION ENGINE									
MECHANICAL ENGINEERING									
	Date of Test : 02/08/2023								
			Date c	of Test :	: 02/08	8/2023	3		
			Date o	of Test :	: 02/08	8/2023	3		
AN	SWER KEY	>	Date c	of Test :	: 02/08	3/2023	3		
AN 1.	SWER КЕҮ (b)	>	(a)	of Test : 13.	(d)	3/2023 19.	3 (d)	25.	(b)
AN 1. 2.	SWER KEY (b) (c)	> 7. 8.	(a) (a)	of Test : 13. 14.	(d) (c)	3 /2023 19. 20.	3 (d) (a)	25. 26.	(b) (d)
AN 1. 2. 3.	SWER KEY (b) (c) (c)	 7. 8. 9. 	(a) (a) (b)	of Test : 13. 14. 15.	(d) (c) (c)	3 /2023 19. 20. 21.	3 (d) (a) (c)	25. 26. 27.	(b) (d) (b)
AN 1. 2. 3. 4.	SWER KEY (b) (c) (c) (a)	 7. 8. 9. 10. 	(a) (a) (b) (c)	of Test : 13. 14. 15. 16.	(d) (c) (c) (b)	3 /2023 19. 20. 21. 22.	3 (d) (a) (c) (c)	25. 26. 27. 28.	(b) (d) (b) (c)
AN 1. 2. 3. 4. 5.	SWER KEY (b) (c) (c) (a) (c)	 7. 8. 9. 10. 11. 	(a) (a) (b) (c) (d)	of Test : 13. 14. 15. 16. 17.	(d) (c) (c) (b) (c)	19. 20. 21. 22. 23.	 3 (d) (a) (c) (c) (b) 	25. 26. 27. 28. 29.	(b) (d) (b) (c) (a)

DETAILED EXPLANATIONS

1. (b)

A good CI engine fuel should have a short ignition lag and will ignite more quickly.

5. (c)

The pour point indicates the lowest temperature at which an oil stops flowing to the pump, bearings or cylinder walls. It is particularly important for immediate oil circulation in respect of starting of engines in very cold climate with gravity lubricating systems. The fluidity is a factor affecting of pour point and viscosity of the cold oil.

7. (a)

Given data:

Compression ratio: r = 10

$$\eta_{\text{Otto}} = 1 - \frac{1}{r^{\gamma - 1}} = 1 - \frac{1}{10^{1.4 - 1}}$$

= $1 - \frac{1}{10^{0.4}} = 0.6018 \text{ or } 60.18\%$

9. (b)

Crank radius, r = 50 mmDiameter of cylinder, D = 90 mmSwept volume = ? \therefore Stroke length, $L = 2r = 2 \times 50 \text{ mm} = 100 \text{ mm}$ Swept volume = $\frac{\pi}{4}D^2L$ = $\frac{\pi}{4} \times (90)^2 \times 100 \times 10^{-3}$ = 636.172 cm³

10. (c)

$$r = 7.5 = \frac{1 + \frac{V_s}{V_c}}{V_c}$$

...

$$\frac{V_c}{V_s} = \frac{1}{6.5}$$

$$Work \text{ output} = 19.5 \times 10^{5} \times V_{c}$$

$$= IMEP \times V_{s}$$

$$Werk = \frac{19.5 \times 10^{5} \times V_{c}}{V_{s}} = \frac{19.5 \times 10^{5}}{6.5} = 3 \times 10^{5} = 3 \text{ bar}$$

$$BMEP = \eta_{m} \times IMEP$$

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11. (d)

Fuel consumption per hour =
$$0.250 \times 90 = 22.5$$
 kg
 \therefore Fuel consumption per cylinder = $\frac{22.5}{6} = 3.75$ kg/hr
Fuel consumption per cycle per cylinder = $\frac{3.75}{60 \times \left(\frac{2500}{2}\right)} = 5 \times 10^{-5}$ kg = 0.05 gm

1

13. (d)

Given data:

$$bp = 12 \text{ kW}$$

$$\eta_{b, \text{th}} = 28\% = 0.28$$

C.V. = 40000 kJ/kg

$$\eta_{b, \text{th}} = \frac{bp}{m_f \times C.V.}$$

$$0.28 = \frac{12}{m_f \times 40000}$$

$$m_f = 1.071 \times 10^{-3} \text{ kg/s} = 3.85 \text{ kg/hour}$$

or 14. (c)

> Net brake power: $bp_{1,2} = 10 \text{ kW}$ Brake power of cylinder-1,

. 1

$$bp_1 = 4.75 \, \text{kW}$$

Brake power of cylinder-2,

$$bp_2 = 4 \text{ kW}$$

Net indicated power,

$$ip_{1,2} = bp_{1,2} + fp$$

= 10 + fp ...(1)

Indicated power of cylinder-1, when spark cut-off of cylinder-2,

$$ip_1 = bp_1 + fp$$

 $ip_1 = 4.75 + fp$...(2)

Indicated power of cylinder-2,

$$ip_2 = ip_{1,2} - ip_1$$

= 10 - 4.75 = 5.25 kW

Indicated power of cylinder-2, when spark cut-off of cylinder-1,

$$ip_2 = bp_2 + fp$$
$$= 4 + fp \qquad \dots(3)$$

Indicated power of cylinder-1,

$$ip_1 = ip_{1,2} - ip_2$$

= 10 - 4 = 6 kW
Net indicated power: $ip_{1,2} = ip_1 + ip_2$
= 6 + 5.25 = 11.25 kW

	Mechanica	l efficiency: $\eta_m =$	Net brake power Net indicated power
		=	$\frac{bp_{1,2}}{ip_{1,2}} = \frac{10}{11.25} = 0.8888 = 88.88\%$
15.	(c)		
	Given data:		
		$V_s =$	0.0259 m ³
		P =	950 kW
	We know that	N =	2200 rpm
		power output,	p Alar p V ar
		P =	$\frac{p_m v_{\text{m}}}{60} \text{ kW} = \frac{p_m v_{\text{s}}}{60}$
	where <i>P</i> is in I p_m is in kPa V _s is in m ³	<w< th=""><th></th></w<>	
	5		Ν
		n =	- rpm
		x =	1, number of cylinder
		950 =	$\frac{p_m \times 0.0259}{60} \times \frac{N}{2} \times 1$
		950 =	$\frac{p_m \times 0.0259 \times 2200}{120}$
	or	$p_m =$	2000 kPa = 2 MPa
16.	(b)		
	Let	BP at full load =	x kW
	BP	at 70% of load =	0.70 <i>x</i> kW
	IP	at 70% of load =	0.70 <i>x</i> + FP
	At 70% load	$\eta_m =$	$\frac{0.70x}{0.70x + FP} = 0.80$

 \Rightarrow

0.70x = 0.56x + 0.80FP 0.14x = 0.80 FP

$$FP = 0.175x$$

Since, FP remains constant at all loads

At full load
$$IP = BP + FP = 50$$

$$x + 0.175 x = 50$$

$$x = \frac{50}{1.175} = 42.55 \text{ kW}$$

17. (c)

Given,

$$r = 18, \alpha_c = 1.9, T_1 = 40^{\circ}\text{C}$$

 $T_2 = T_1(r)^{\gamma-1}$



and,

18. (a)

 $\frac{T_2}{T_1} = \left(\frac{V_1}{V_2}\right)^{\gamma-1}$ $= (14)^{1.4-1} = 2.873$ $T_1 = 16 + 273 = 289 \text{ K}$ $T_2 = 2.873 \times 289 = 830.5 \text{ K}$ For process 2 to 3 (V = C) $\frac{P_3}{T_2} = \frac{T_3}{T_2}$

$$P_{2}^{-} = I_{2}^{-}$$

$$T_{3} = \frac{P_{3}}{P_{2}} \times T_{2} = \frac{60}{P_{2}} \times 830.5$$

 \Rightarrow



and	$\frac{P_2}{P_1} = \left(\frac{V_1}{V_2}\right)^{\gamma} = (14)^{1.4} = 40.23$
	$P_2 = 40.23 \times 1 = 40.23$ bar
\Rightarrow	$T_3 = \frac{60 \times 830.5}{40.23} = 1238.54 \text{ K}$
Giver	$(Q_s)_{V=C} = (Q_s)_{P=C}$
\Rightarrow	$C_{v}(T_{3} - T_{2}) = C_{p}(T_{4} - T_{3})$
\Rightarrow	$0.718(1238.54 - 830.5) = 1.005(T_4 - 1238.54)$
\Rightarrow	$T_4 = \frac{0.718 \times 408.044}{1.005} + 1238.54 = 1530.058 \text{ K}$
Now	$\frac{V_4}{V_3} = \frac{T_4}{T_3} = \frac{1530.058}{1238.54} = 1.2354$
<i>.</i> .	$\frac{V_5}{V_4} = \frac{V_1}{V_4} = \frac{V_1}{V_2} \times \frac{V_3}{V_4} = \frac{14 \times 1}{1.2354} = 11.332$
	$\frac{T_4}{T_5} = \left(\frac{V_5}{V_4}\right)^{\gamma-1} = (11.332)^{0.4} = 2.640755$
	$T_5 = \frac{1530.058}{2.640755} = 579.402 \text{ K}$
÷	Heat supplied = $c_v(T_3 - T_2) + c_p(T_4 - T_3)$
	$= 2c_{v}(T_{3}-T_{2})$
	$= 2 \times 0.718(1238.54 - 830.5)$
	= 585.94 kJ/kg
	and heat rejected = $c_v(T_5 - T_1) = 0.718(579.402 - 289) = 208.508 \text{ kJ/kg}$
<i>.</i> :	$\eta = 1 - \frac{Q_{\text{rej}}}{Q_s} = 1 - \frac{208.508}{585.94} = 0.6442 \text{ or } 64.42\%$

19. (d)

Air consists of 21% oxygen and 79% nitrogen thus for each mole of oxygen in air there are 3.76 moles of nitrogen.

The chemical reaction of combustion of methane is given as

CH₄ + 2O₂ + 7.52 N₂ → CO₂ + 2H₂O + 7.52 N₂
16 kg of CH₄ requires = 2 × 32 + 7.52 × 28
= 274.56 kg of air
A/F ratio =
$$\frac{274.56}{16}$$
 = 17.16

 \Rightarrow

The density of the air =
$$\frac{P}{RT} = \frac{1.01 \times 10^5}{287 \times 295} = 1.1929 \text{ kg/m}^3$$

Volume of air required per kg of fuel = $\frac{17.16}{1.1929}$ = 14.38 m³/kg of fuel

20. (a)

Given:

n:

$$P_{1} = 1 \text{ bar}, T_{1} = T_{2} = 300 \text{ K}$$

$$P_{3} = 15 \text{ bar}, T_{3} = 700 \text{ K}$$
Compression ratio,

$$\frac{V_{1}}{V_{2}} = \frac{V_{1}}{V_{3}} = \frac{V_{4}}{V_{3}} = r$$
Net work done = $W_{\text{expansion}} - W_{\text{compression}}$

$$= \text{RT}_{3} \log_{e} r - \text{RT}_{1} \log_{e} r$$

$$= R(T_{3} - T_{1}) \log_{e} r$$

Assume 1 kg of air at the beginning of compression process.

$$v_{1} = \frac{RT_{1}}{P_{1}} = 287 \times \frac{300}{10^{5}} = 0.861 \text{ m}^{3}/\text{kg}$$

$$v_{2} = v_{3} = \frac{RT_{3}}{P_{3}} = 287 \times \frac{700}{15 \times 10^{5}} = 0.134 \text{ m}^{3}/\text{kg}$$

$$r = \frac{v_{1}}{v_{2}} = 6.425$$
Stroke volume = $v_{1} - v_{2} = 0.727 \text{ m}^{3}/\text{kg}$

$$W = R(T_{3} - T_{1})\log_{e} r = 287 \times (700 - 300)\log_{e} 6.425$$

$$= 213.55 \text{ kJ/kg of air}$$

Mean effective pressure =
$$\frac{\text{Work done}}{\text{Stroke volume}} = \frac{213.55}{0.727} = 2.94 \text{ bar}$$

21. (c)

$$\eta_m = 0.8 = \frac{b.p}{i.p.} \Rightarrow i.p. = \frac{50}{0.8} = 62.5 \text{ kW}$$

$$\Rightarrow \qquad \eta_{ith} = 0.35 = \frac{i.p.}{\dot{V}_f (C.V.)_f} \Rightarrow \dot{V}_f = \frac{62.5}{34 \times 1000 \times 0.35}$$
$$\dot{V}_f = 5.252 \times 10^{-3} \text{ m}^3/\text{s}$$
$$\Rightarrow \qquad \frac{A}{F} = \frac{10}{1} \Rightarrow \frac{\dot{V}_a}{\dot{V}_f} = 10 \Rightarrow \dot{V}_a = 0.05252 \text{ m}^3/\text{s}$$
$$\eta_v = \frac{(\dot{V}_a)_{air}}{\dot{V}_s} = \frac{0.05252}{2000 \times 10^{-6} \times \frac{N}{120}} = 0.7$$
$$\Rightarrow \qquad N = 4501.71 \text{ rpm}$$

22. (c)

Compression ratio = expansion ratio × cut-off ratio $r = 15 \times 1.5$ r = 22.5Now, $\eta_{air-std} = 1 - \frac{1}{r^{\gamma-1}} \frac{\rho^{\gamma} - 1}{\gamma(\rho - 1)}$ $\eta_{air-std} = 1 - \frac{1}{22.5^{0.4}} \times \frac{1.5^{1.4} - 1}{1.4(1.5 - 1)}$ $\eta_{air-std} = 0.6858$ Now, $\eta_{rel} = \frac{\eta_{bth}}{\eta_{air-std}}$ $\eta_{bth} = 0.4 \times 0.6858$ $\eta_{bth} = 0.27432$ $\frac{BP}{\dot{m}_{f} \times CV} = 0.27432$ $\dot{m}_{f} = \frac{30 \times 3600}{0.27432 \times 42000}$ $\dot{m}_{f} = 9.3738 \text{ kg/h}$

23. (b)

Brake power =
$$T\omega = \frac{300 \times 2\pi \times 500}{60 \times 10^3}$$
 kW
BP = 15.7 kW

Calculating indicated power:

Spring constant =
$$\frac{10 \times 10^4 Pa}{1.4 \text{ mm}}$$
 = 71.43 kPa/mm
Indicated pressure = $\frac{\text{Area of indicator diagram}}{\text{Length of indicator diagram}} \times \text{Spring constant}$
Indicated pressure = $\frac{500}{60} \times 71.43 = 595.24 \text{ kPa}$
Indicated pressure = $595.24 \times \frac{LAN}{60 \times 2} = 595.24 \times \frac{0.3 \times \frac{\pi}{4} \times 0.2^2 \times 500}{60 \times 2} = 23.375 \text{ kW}$
 $\eta_m = \frac{bp}{i.p} \times 100 = 67.165\%$

24. (b)



$$V_c = V_2 = 30 \text{ cm}^3$$

 $r = 20 = 1 + \frac{V_s}{V_c}$
 $V_s = 570 \text{ cm}^3$

After 30% of compression stroke

$$V_1' = V_c + 0.7 V_s$$

= 30 + 0.7 × 570 = 429 cm³

25. (b)

 $\Rightarrow \qquad T_1 V_1^{\gamma-1} = T_2 V_2^{\gamma-1}$ $\Rightarrow \qquad T_2 = T_1 \left(\frac{V_1}{V_2}\right)^{\gamma-1}$ $\Rightarrow \qquad T_2 = 340(20)^{0.4} = 1126.92 \text{ K}$ Stroke volume = $(V_1 - V_2)$ $V_3 = 0.1(V_1 - V_2)$ $\frac{V_3}{V_2} = 0.1(20 - 1) = 1.9$ Heat addition takes place at constant pressure (i.e. process 2-3)

$$\frac{V_3}{V_2} = \frac{T_3}{T_2}$$
$$T_3 = 1.9 \times [1126.92]$$
$$= 2141.14 \text{ K}$$

 \Rightarrow



26. (d)

or	

or

$$\eta = 1 - \left(\frac{1}{r}\right)^{\gamma - 1}$$

$$\eta = 1 - \left(\frac{V_2}{V_1}\right)^{\gamma - 1}$$

$$\vdots$$

$$\eta = 1 - \frac{T_1}{T_2}$$

$$\eta = \frac{WD}{HA} = \frac{WD}{WD + HR} = \frac{1}{1 + HR / WD}$$

or
$$\eta = \frac{HA - HR}{HA} = 1 - \frac{HR}{HA}$$

or
$$\eta = 1 - \frac{mc_v (T_4 - T_1)}{mc_v (T_3 - T_2)}$$

:.
$$\eta = 1 - \frac{T_4 - T_1}{T_3 - T_2}$$

27. (b)

for N₂ gas,
$$\gamma = 1.66$$

Compression ratio,
$$r = \frac{v_1}{v_2} = \frac{v_1}{0.125v_1} = 8$$

...

$$\eta_{\text{th}} = \frac{1 - \frac{1}{(r)^{\gamma - 1}} = 1 - \frac{1}{(8)^{1.66 - 1}} = 0.7465$$

Now

Now
$$\eta_{th} = \frac{W_{net}}{Q_{in}}$$

 $\Rightarrow \qquad W_{net} = 0.7465 \times 150 = 111.9765 \,\text{kW}$
 $= \frac{111.9765 \times 1000}{100} = 150.10 \,\text{km}$

 $\eta_{th} =$

$$= \frac{111.9765 \times 1000}{746} = 150.10 \text{ hF}$$

28. (c)

With the increase of load in CI engine, cut off ratio increases.

30. (c)

Upper limit of compression ratio in SI engine is fixed by antiknock quality of the fuel.

