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Strength of Materials

MECHANICAL ENGINEERING

Date of Test : 20/09/2026

ANSWER KEY ➤

1. (c)	7. (a)	13. (a)	19. (d)	25. (b)
2. (d)	8. (b)	14. (c)	20. (d)	26. (c)
3. (c)	9. (a)	15. (b)	21. (c)	27. (d)
4. (b)	10. (c)	16. (b)	22. (b)	28. (c)
5. (a)	11. (c)	17. (c)	23. (a)	29. (b)
6. (d)	12. (b)	18. (d)	24. (a)	30. (a)

DETAILED EXPLANATIONS

1. (c)

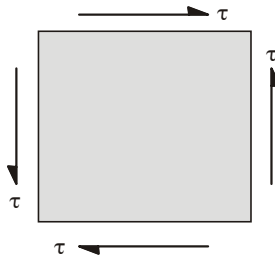
Deflection of a simply supported beam is proportional to,

$$\delta \propto \frac{PL^3}{EI}$$

∴ Increasing I , decreasing L or P will reduce deflection.

2. (d)

The stress system for an element on the surface is



All normal stresses are zero.

Stress on an inclined plane is given by,

$$\sigma_n = \frac{\sigma_1 + \sigma_2}{2} + \frac{\sigma_1 - \sigma_2}{2} \cos 2\theta + \tau \sin 2\theta$$

$$\sigma_t = \frac{\sigma_1 - \sigma_2}{2} \sin 2\theta - \tau \cos 2\theta$$

Now at $\theta = 45^\circ$,

$$\therefore \sigma_n = \tau \sin 90^\circ = \tau \text{ and } \sigma_t - \tau \cos 90^\circ = 0$$

∴ 45° plane is the principal plane as shear stress on it is zero. Value of maximum principal stress is ' τ '.

3. (c)

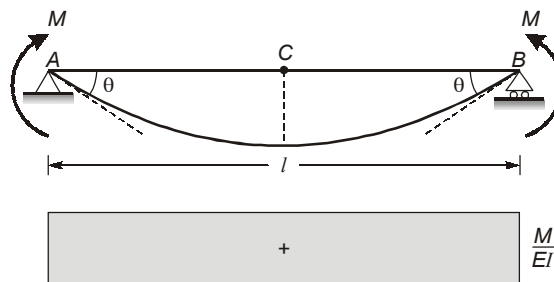
A ductile material fails through a cup and cone type of failure.

4. (b)

$$\text{Volumetric strain, } \frac{\Delta V}{V} = \frac{\sigma}{E}(1 - 2\mu)$$

$$= \frac{180(1 - 2 \times 0.3)}{205 \times 1000} = 3.5 \times 10^{-4} \text{ or } 0.035\%$$

5. (a)



$$\theta_C - \theta_A = \text{Area of } \frac{M}{EI} \text{ diagram}$$

$$0 - \theta_A = +\frac{M}{EI} \times \frac{l}{2} = \frac{Ml}{2EI}$$

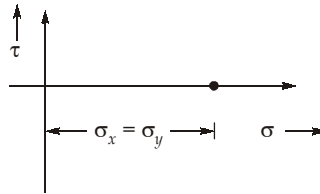
$$\theta_A = -\frac{Ml}{2EI}$$

$$\theta_A = \frac{Ml}{2EI} \text{ (Anti-clockwise)}$$

$$\therefore \frac{Ml}{EI} = 2\theta_A = 2\theta$$

6. (d)

When normal stresses are equal (same magnitude) and of same nature, then Mohr's circle will be reduced to a point.



Ex. Hydrostatic pressure, internal contact pressure in hollow shafts due to press-fitting of bearing.

7. (a)

$$\tan 2\theta_{p_1} = \frac{\tau_{xy}}{\left(\frac{\sigma_x - \sigma_y}{2}\right)}$$

$$\begin{aligned} \theta_{p_1} &= \frac{1}{2} \tan^{-1} \left(\frac{2 \times (-5)}{8 - 5} \right) \\ &= \frac{1}{2} \tan^{-1} \left(\frac{-10}{3} \right) = \frac{1}{2} \times -73.3 = -36.65^\circ \end{aligned}$$

and $\theta_{p_2} = \theta_{p_1} + 90 = -36.65 + 90 = 53.349^\circ$

8. (b)

We know that,

$$E = \frac{9KG}{3K + G}$$

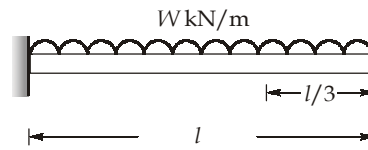
$$\Rightarrow 3KE + EG = 9KG$$

$$\Rightarrow 9KG - 3KE = EG$$

$$\Rightarrow K(9G - 3E) = GE$$

$$\begin{aligned} \Rightarrow K &= \frac{GE}{9G - 3E} = \frac{120 \times 42}{9 \times 42 - 3 \times 120} \\ &= 280 \text{ GPa} \end{aligned}$$

9. (a)



Maximum bending will be at fixed end,

i.e.
$$M = (Wl) \times \frac{l}{2} = \frac{Wl^2}{2}$$

Bending moment at $\frac{l}{3}$ distance from free end

$$\Rightarrow M' = \left(\frac{Wl}{3}\right) \times \left(\frac{l}{6}\right) = \frac{Wl^2}{18}$$

$$M' = \frac{2}{9} \times \frac{Wl^2}{2} = \frac{M}{9} = 0.1111 = 11.11\% \text{ of } M$$

10. (c)

$$\begin{aligned} \text{Normal stress, } \sigma_n &= \sigma_x \cos^2 \theta + \sigma_y \sin^2 \theta \\ &= 80 \cos^2 30^\circ + 40 \sin^2 30^\circ = 70 \text{ MPa} \end{aligned}$$

$$\text{Shear stress, } \tau = -\frac{\sigma_x - \sigma_y}{2} \sin 2\theta = -\frac{80 - 40}{2} \sin 60^\circ$$

$$\Rightarrow \tau = -17.32 \text{ MPa}$$

$$\tau = 17.32 \text{ MPa (Numerically)}$$

$$\sigma_r = \sqrt{70^2 + 17.32^2} = 72.1 \text{ MPa}$$

11. (c)

Taking moment about 'B'

$$R_a \times l = \frac{wl}{2} \times \frac{2l}{3}$$

$$\therefore R_a = \frac{wl}{3}$$

and
$$R_b = \frac{wl}{2} - \frac{wl}{3} = \frac{wl}{6}$$

Intensity of loading at a distance 'x' from end 'B' = $\frac{w}{l}x$

$$\therefore F_x = -R_b + \frac{1}{2} \frac{wx}{l} x = -\frac{wl}{6} + \frac{wx^2}{2l} \quad (\text{Parabolic})$$

Shear force is zero at

$$-\frac{wl}{6} + \frac{wx^2}{2l} = 0$$

$$\therefore \frac{wl}{6} = \frac{wx^2}{2l}$$

$$\Rightarrow x = \frac{l}{\sqrt{3}}$$

12. (b)

Bending moment at section 'x'

$$M_x = \frac{wlx}{6} - \frac{wx^2}{2l} \cdot \frac{x}{3}$$

$$M_x = \frac{wlx}{6} - \frac{wx^3}{6l} \quad (\text{Cubic})$$

$$\text{At } x = \frac{l}{\sqrt{3}}$$

$$\begin{aligned} \therefore M_{\max} &= \frac{wl}{6} \times \frac{l}{\sqrt{3}} - \frac{w}{6l} \times \left(\frac{l}{\sqrt{3}} \right)^3 \\ &= \frac{wl^2}{9\sqrt{3}} \end{aligned}$$

13. (a)

$$\begin{aligned} \text{Given, } \sigma_1 &= 100 \text{ MPa,} & \sigma_2 &= 50 \text{ MPa,} \\ \sigma_3 &= 25 \text{ MPa,} & S_{yt} &= 220 \text{ MPa,} \end{aligned}$$

For maximum shear strain energy theory,

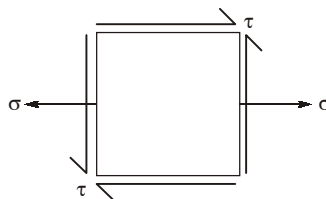
$$(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 \leq 2 \left(\frac{S_{yt}}{N} \right)^2 \quad [\text{Where, } N = \text{factor of safety}]$$

$$(100 - 50)^2 + (50 - 25)^2 + (25 - 100)^2 = 2 \left(\frac{220}{N} \right)^2$$

After solving,

$$\therefore \text{Factor of safety, } N = 3.326 \sim 3.33$$

14. (c)



$$\text{State of stress, } \sigma = \frac{P}{2\pi rt} \quad (\text{Area of cross-section for a thin tube} = 2\pi rt = \pi dt)$$

$$\sigma = \text{Simple tensile stress} = \frac{10 \times 10^3}{\pi \times 25 \times 1.6} = 79.6 \text{ N/mm}^2$$

Now,

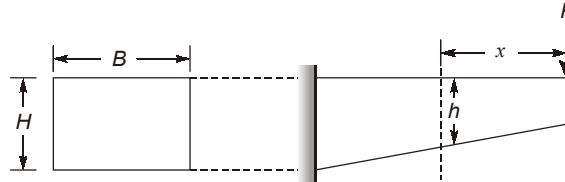
$$J = \frac{\pi D^3 t}{8} \text{ for a thin tube}$$

$$\therefore \tau = \frac{T}{J} r = \frac{8T}{\pi D^3 t} \times \frac{D}{2} = \frac{4T}{\pi D^2 t} = \frac{4 \times 23.5 \times 10^3}{\pi \times 25^2 \times 1.6} = 29.94 \text{ N/mm}^2$$

$$\begin{aligned}\text{Principal stresses, } \sigma_{1,2} &= \frac{\sigma}{2} \pm \sqrt{\left(\frac{\sigma}{2}\right)^2 + \tau^2} = 39.8 \pm \sqrt{39.8^2 + 29.94^2} \\ &= 39.8 \pm 49.8 = 89.6 \text{ N/mm}^2 \text{ and } -10 \text{ N/mm}^2\end{aligned}$$

∴ Maximum principal stress = 89.6 MPa

15. (b)



$$\text{Stress at support, } \sigma_1 = \frac{M}{Z} = \frac{P \times L}{\left(\frac{B \times H^3}{12}\right)} \times \frac{H}{2} = \frac{6PL}{BH^2}$$

$$\text{Stress at distance } x, \sigma_2 = \frac{M}{Z} = \frac{6Px}{Bh^2}$$

Equating,

$$\sigma_1 = \sigma_2$$

$$\frac{6PL}{BH^2} = \frac{6Px}{Bh^2}$$

$$h = \sqrt{\frac{x}{L}} \cdot H$$

16. (b)

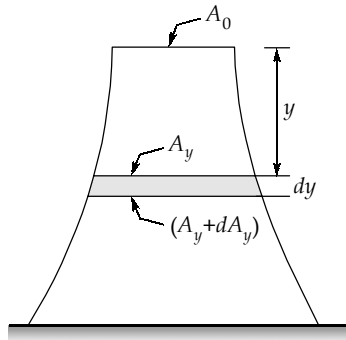
$$\text{Effective length of column } (L_e) = \frac{L}{\sqrt{2}}$$

$$\text{Least radius of gyration } (k) = \sqrt{\frac{I_{\min.}}{A}} = \left(\frac{\pi D^4}{64} \times \frac{4}{\pi D^2} \right)^{1/2} = \frac{D}{4}$$

$$\text{Slenderness ratio } (s) = \frac{L_e}{k} = \frac{\left(\frac{L}{\sqrt{2}}\right)}{\left(\frac{D}{4}\right)}$$

$$s = 2\sqrt{2} \frac{L}{D}$$

17. (c)



Let us consider a cross-sectional area A_y at a distance x from top and having thickness dy (as shown)

Force acting due to self weight at section dy

$$\Rightarrow W_y = \int_0^y \rho g A_y dy$$

Now, taking force balance at the section ' dy '

$$\Rightarrow \sigma A_y + \int \rho g A_y dy = \sigma (A_y + dA_y) \quad [\because \sigma = \text{Constant for uniform strength beam}]$$

$$\Rightarrow \sigma A_y + \int \rho g A_y dy = \sigma A_y + \sigma dA_y$$

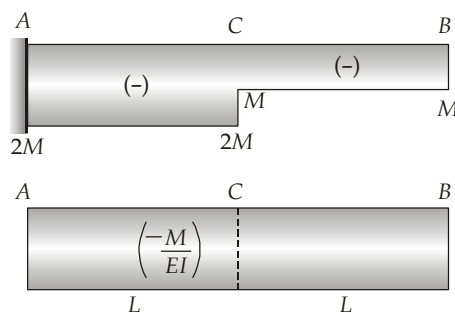
$$\Rightarrow \int_0^y \rho g dy = \int_{A_0}^{A_y} \frac{\sigma dA_y}{A_y}$$

$$\Rightarrow \rho g y = \sigma \log \left[\frac{A_y}{A_0} \right]$$

$$\Rightarrow A_y = A_0 e^{(\rho g y / \sigma)}$$

18. (d)

BMD for the given beam



For deflection at B,

$$\begin{aligned} \delta_{B/A} &= \frac{-M}{EI} \times L \times \frac{L}{2} - \frac{M}{EI} \times L \times \frac{3L}{2} \\ &= \frac{-ML^2}{EI} \left[\frac{1}{2} + \frac{3}{2} \right] = -\frac{2ML^2}{EI} \end{aligned}$$

(-)Ve sign indicates downward deflection.

19. (d)

Effective equivalent length,

$$L_e = \frac{L}{3}$$

$$\begin{aligned} \text{As, } P_e &= \frac{\pi^2 EI}{L_e^2} \\ &= \frac{\pi^2 EI}{(L/3)^2} = \frac{9\pi^2 EI}{L^2} \end{aligned}$$

20. (d)

Actual column is having end constraints as fixed and free but it was considered as fixed and hinged column.

$$\begin{aligned} (L_{eq})_{\text{actual}} &= 2L \\ \Rightarrow (P_e)_{\text{actual}} &= \frac{\pi^2 EI}{L_e^2} = \frac{\pi^2 EI}{4L^2} \quad \dots(i) \\ (P_e)_{\text{assumed}} &= \frac{\pi^2 EI}{(L/\sqrt{2})^2} \quad \because (L_{eq}) = \frac{L}{\sqrt{2}} \\ &= \frac{2\pi^2 EI}{L^2} \\ \% \text{error} &= \frac{\text{Error}}{(P_e)_{\text{Actual}}} = \frac{\frac{\pi^2 EI}{L^2} \left[2 - \frac{1}{4} \right]}{\frac{\pi^2 EI}{L^2} \times \frac{1}{4}} \\ &= 700\% \end{aligned}$$

21. (c)

$$\text{Inside diameter, } D_i = D_o - 2t = 50 - 2 \times 5 = 40 \text{ mm}$$

$$\text{Permissible stress, } \sigma_y = \frac{240}{3} = 80 \text{ MPa}$$

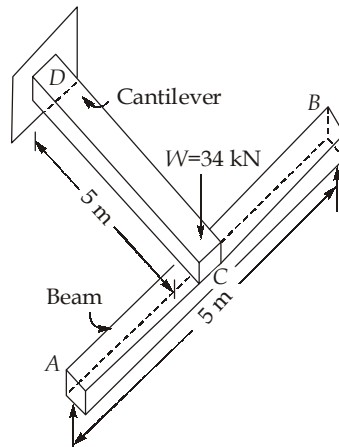
$$\text{Maximum shear stress, } \tau_{\max} = \frac{80}{2} = 40 \text{ MPa}$$

$$\text{or } \frac{16TD}{\pi(D^4 - d^4)} = 40$$

$$\frac{16 \times T \times 50}{\pi(50^4 - 40^4)} = 40$$

$$\begin{aligned} \therefore T &= \frac{40 \times \pi \times (50^4 - 40^4)}{16 \times 50} \\ T &= 579.6 \text{ Nm} \end{aligned}$$

22. (b)



Let the load shared by the cantilever be ' P '

For the cantilever,

$$\Delta_C = \frac{PL^3}{3EI}$$

For the beam,

$$\Delta_S = \frac{(W - P)L^3}{48EI}$$

$$\therefore \frac{PL^3}{3EI} = \frac{(W - P)L^3}{48EI}, \text{ on solving we get}$$

$$\therefore P = \frac{W}{17} = \frac{34}{17} = 2 \text{ kN}$$

23. (a)

$$\begin{aligned} \gamma_{xy} &= 2\varepsilon_{45^\circ} - (\varepsilon_0 + \varepsilon_{90^\circ}) \\ &= 2 \times 200 - (-500 + 300) \end{aligned}$$

$$\gamma_{xy} = 600 \mu\text{m/m}$$

$$\begin{aligned} \varepsilon_1, \varepsilon_2 &= \left(\frac{\varepsilon_x + \varepsilon_y}{2} \right) \pm \sqrt{\left(\frac{\varepsilon_x - \varepsilon_y}{2} \right)^2 + \left(\frac{\gamma_{xy}}{2} \right)^2} \\ &= \frac{-500 + 300}{2} \pm \sqrt{\left(\frac{-500 - 300}{2} \right)^2 + 300^2} \end{aligned}$$

$$\varepsilon_{1,2} = -100 \pm 500$$

$$\varepsilon_1 = -600 \mu\text{m/m}$$

$$\varepsilon_2 = 400 \mu\text{m/m}$$

$$\sigma_1 = \frac{E}{1 - \mu^2} [\varepsilon_1 + \mu\varepsilon_2]$$

$$= \frac{200 \times 10^3}{(1 - 0.3^2)} [-600 + 0.3 \times 400]$$

$$\sigma_1 = -105.49 \simeq -105 \text{ MPa}$$

24. (a)

$$\sigma_h = \frac{pd}{2t \times \eta_{LJ}} = \frac{6 \times 150}{2 \times 12.5 \times 0.8} = 45 \text{ MPa}$$

$$\sigma_l = \frac{pd}{4t \times \eta_{CJ}} = \frac{6 \times 150}{4 \times 12.5 \times 0.9} = 20 \text{ MPa}$$

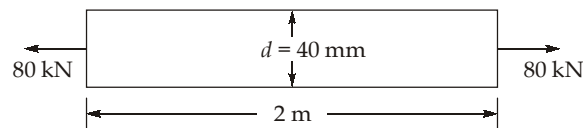
$$\frac{\delta d}{d} = \frac{1}{E}(\sigma_h - \mu \sigma_L) = \frac{1}{200 \times 10^3}(45 - 0.25 \times 20)$$

$$\frac{\delta d}{d} = 0.2 \times 10^{-3}$$

$$\delta d = 0.2 \times 150 \times 10^{-3} \text{ mm}$$

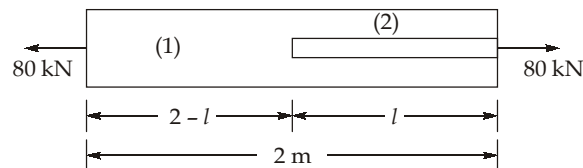
$$\delta d = 0.03 \text{ mm}$$

25. (b)



$$\text{Elongation of the rod, } \delta l_1 = \frac{PL}{AE} = \frac{80 \times 1000 \times 2000}{\frac{\pi}{4} \times 40^2 \times 2 \times 10^5} = 0.6366 \text{ mm}$$

Let the rod be bored to a length of l meters.



Elongation of the rod after boring,

$$\delta l_2 = 1.2 \times \delta l_1 = 1.2 \times 0.6366 = 0.7639 \text{ mm}$$

$$\delta l_2 = \frac{Pl}{A_2 E} + \frac{P(2-l)}{A_1 E}$$

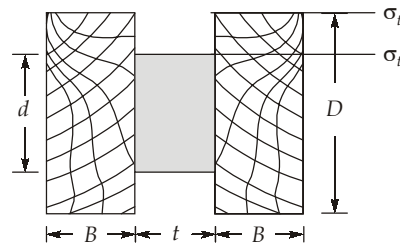
$$= \frac{80000}{2 \times 10^5} \left[\frac{l \times 1000}{\frac{\pi}{4} \times (40^2 - 20^2)} + \frac{(2-l) \times 1000}{\frac{\pi}{4} \times 40^2} \right]$$

$$= \frac{80000}{2 \times 10^5} [1.061l + (2-l)0.795]$$

$$= \frac{80000}{2 \times 10^5} [1.59 + 0.266l] = 0.7639$$

$$l = 1.202 \text{ m}$$

26. (c)
Refer figure,



Let the stress in extreme timber fiber be σ_t
Then the stress in timber at extreme surface of steel,

$$\sigma'_t = \sigma_t \frac{d/2}{D/2} = \sigma_t \frac{d}{D}$$

Then if the stress in the extreme steel fibre is to reach maximum value,

$$\sigma_s = m \sigma'_t \quad \left[\text{Modular ratio, } m = \frac{E_s}{E_t} \right]$$

$$130 = \frac{210}{12} \times 10 \times \frac{d}{D}$$

or

$$\frac{D}{d} = \frac{210 \times 10}{12 \times 130} = 1.346 \simeq 1.35$$

27. (d)

$$\text{Given: } y = \frac{1}{EI} \left(2x^3 - \frac{x^4}{6} - 36x \right)$$



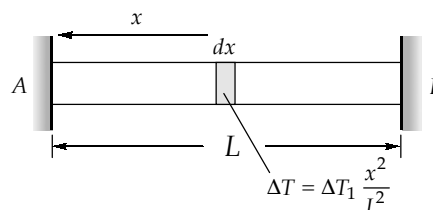
In a simply supported beam, bending moment at both the end of beam is zero.

$$\begin{aligned} M &= EI \frac{d^2 y}{dx^2} = EI \times \frac{1}{EI} \times \frac{d}{dx} \left(6x^2 - \frac{4x^3}{6} - 36 \right) \\ &= 12x - \frac{12x^2}{6} - 0 = 12x - 2x^2 \end{aligned}$$

$$\begin{aligned} \text{Put } M &= 0, \quad 12x - 12x^2 = 0 \\ 2x(6 - x) &= 0 \\ x &= 0, x = 6 \end{aligned}$$

So, the span of the beam is 6 metres.

28. (c)



Induced strain in dx element,

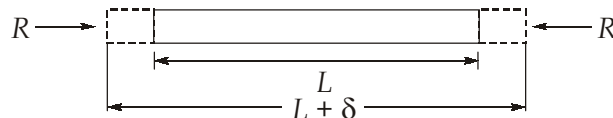
$$\frac{\delta x}{dx} = \alpha \Delta T$$

$$\delta x = \alpha \Delta T \cdot dx$$

Total change in length,

$$\int_0^L \delta x = \int_0^L \alpha \cdot \Delta T_1 \cdot \frac{x^2}{L^2} \cdot dx$$

$$\delta = \frac{\alpha \Delta T_1 L^3}{3 L^2} = \frac{\alpha \Delta T_1 L}{3}$$



As the both ends are rigid, no elongation will occur due to rigid support reactions induced R

$$\sigma_{\text{thermal}} = \frac{R}{A} = E \cdot \frac{\delta}{L}$$

$$\sigma_{\text{thermal}} = \frac{E \alpha \Delta T_1}{3}$$

29. (b)

Given : $L = 1 \text{ m}$; $W = 600 \text{ N}$; $EI = 20 \text{ kNm}^2$

For cantilever beam with transverse shear load W ,

$$\delta_1 = \text{Deflection at mid location} = \frac{WL^3}{3EI}$$

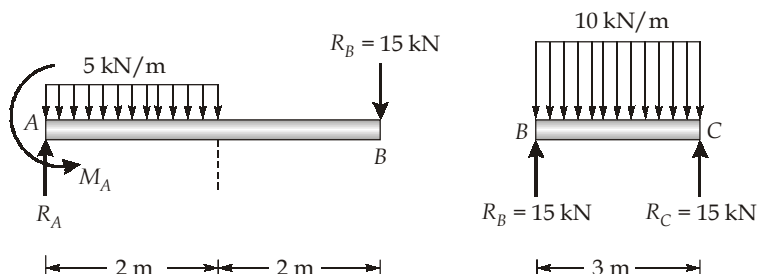
$$\delta_2 = (\text{Slope at mid location}) \times L = \frac{WL^2}{2EI} \times L$$

Deflection at the tip, $\delta = \delta_1 + \delta_2$

$$= \frac{WL^3}{3EI} + \frac{WL^3}{2EI} = \frac{5WL^3}{6EI}$$

$$= \frac{5 \times 600 \times 1^3}{6 \times 20000} = 0.025 \text{ m} = 25 \text{ mm}$$

30. (a)



Taking moment about point A,

$$M_A - (5 \times 2) \times 1 - 15 \times 4 = 0$$

$$\Rightarrow M_A = 10 + 60 = 70 \text{ kNm}$$

