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Railway + Airport

CIVIL ENGINEERING

Date of Test : 18/03/2024

ANSWER KEY >

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|--------|---------|---------|---------|---------|
| 1. (c) | 6. (b) | 11. (b) | 16. (a) | 21. (d) |
| 2. (d) | 7. (b) | 12. (b) | 17. (a) | 22. (b) |
| 3. (a) | 8. (b) | 13. (a) | 18. (a) | 23. (c) |
| 4. (d) | 9. (b) | 14. (c) | 19. (a) | 24. (d) |
| 5. (d) | 10. (d) | 15. (c) | 20. (b) | 25. (c) |

DETAILED EXPLANATIONS

1. (c)

Composite Sleeper Index (CSI), measures the mechanical strength of timber, derived from its composite properties of strength and hardness

$$\text{CSI} = \frac{S + 10H}{20}$$

where,

S = Strength index at 12% moisture content

H = Hardness index at 12% moisture content.

2. (d)

Length of BG rail = 12.8 m

$$\text{Number of BG rails in 800 m} = \frac{800}{12.8} = 62.5 \simeq 63$$

Sleeper density = 12.8 + 5 = 17.8 \simeq 18 sleepers per rail

$$\therefore \text{Number of sleepers} = 18 \times 63 = 1134$$

3. (a)

Grade provided = Ruling gradient - Grade compensation

$$= 1 \text{ in } 250 - 0.04\% \times 4^\circ$$

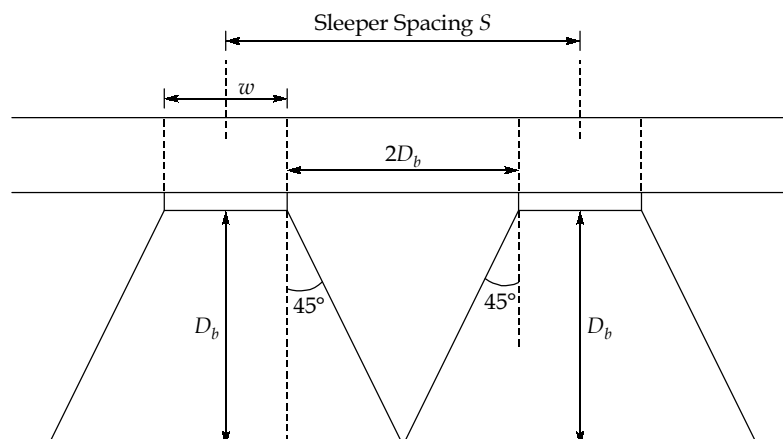
$$= \frac{1}{250} - \frac{0.16}{100}$$

$$= 0.0024 = 0.24\%$$

4. (d)

$$\text{Radius of exit taxiway, } R = \frac{V^2}{125f} = \frac{80^2}{125 \times 0.15} = 341.33 \simeq 342 \text{ m}$$

5. (d)



From figure,

$$S = 2 D_b + w$$

$$D_b = \frac{S - w}{2} = \frac{65 - 25}{2} = 20 \text{ cm}$$

6. (b)

$$\begin{aligned} \text{CL} &= 2 \text{ GN} \\ &= 2 \times 1.676 \times 16 \\ &= 53.63 \text{ m} \end{aligned}$$

7. (b)

8. (b)

9. (b)

Standard atmospheric temperature

$$\begin{aligned} &= 15^\circ - (0.0065 \times \text{Change in elevation above MSL}) \\ &= 15^\circ\text{C} - (0.0065 \times 700) \\ &= 15 - 4.55 \\ &= 10.45^\circ\text{C} \end{aligned}$$

10. (d)

Population, agriculture and industrial development factors are utilized in deciding the best alignment.

11. (b)

Grade resistance + Curve resistance = Gradient resistance

$$\Rightarrow W \tan \theta + 0.04\% \times 4 \times W = \frac{W}{200}$$

$$\Rightarrow W \tan \theta + 0.0004 \times 4 \times W = \frac{W}{200}$$

$$\Rightarrow \tan \theta = \frac{1}{200} - 0.0004 \times 4$$

$$= 3.4 \times 10^{-3} = \frac{1}{294}$$

12. (b)

$$R = 250 \text{ m}$$

$$e = 20 \text{ cm}$$

$$V_{\max} = 58.84 \text{ kmph}$$

$$\begin{aligned} \therefore \text{Length of transition curve} &= \max \begin{cases} 7.2e = 7.2 \times 20 = 144 \text{ m} \\ 0.073(e) V_{\max} = 0.073 \times 20 \times 58.84 = 85.91 \text{ m} \end{cases} \\ &= 144 \text{ m} \end{aligned}$$

13. (a)

$$\text{Length of track, } l = (D - G)N + G(4N - \sqrt{1 + N^2})$$

$$\text{Given, } N = 15, D = 7.5\text{m}, G = 1.676\text{ m}$$

$$\begin{aligned} l &= (7.5 - 1.676) \times 15 + 1.676(4 \times 15 - \sqrt{1 + 15^2}) \\ &= 87.36 + 75.36 \\ &= 162.73\text{ m} \end{aligned}$$

The length of straight distance

$$\begin{aligned} &= l - 4GN \\ &= 162.73 - 4 \times 1.676 \times 15 \\ &= 62.17\text{ m} \end{aligned}$$

14. (c)

$$\begin{aligned} w &= \frac{13(B + L)^2}{R} = \frac{13(6 + 0.05)^2}{250} \\ &= 1.903\text{ m} \end{aligned}$$

15. (c)

$$\begin{aligned} \text{Hauling capacity} &= \mu n w_d \\ &= 0.2 \times 3 \times 20 = 12\text{ tonnes} \end{aligned}$$

For train moving on straight and level track,

$$\text{Hauling capacity} = \text{Total train resistance}$$

$$\text{Total train resistance} = R_{T1} + R_{T2} + R_{T3} + R_g \quad (\because R_g = W + \tan\theta = 0)$$

$$R_{T1} = \text{resistance independent of speed} = 0.0016w$$

$$R_{T2} = \text{resistance dependent of speed} = 0.00008wv = (0.00008 \times 100)w = 0.008w$$

$$R_{T3} = \text{atmospheric resistance} = 0.0000006wv^2 = (0.0000006 \times 100^2)w = 0.006w$$

$$\therefore 12 = 0.0016w + 0.008w + 0.006w$$

$$\Rightarrow 12 = 0.0156w$$

$$\Rightarrow w = 769.23\text{ tonnes} \simeq 769\text{ tonnes}$$

16. (a)

Gate capacity for single gate,

$$\begin{aligned} G_C &= \frac{1}{\text{Weighted service time}} \\ &= \frac{1}{(0.2 \times 30) + (0.2 \times 40) + (0.6 \times 60)} \\ &= 0.02\text{ aircraft/min/gates} \end{aligned}$$

$$\text{Capacity of all gates } C, = G_C \times \text{Number of gate}$$

$$= 0.02 \times 20$$

$$= 0.4\text{ aircraft/min}$$

$$= 24\text{ aircraft/hour}$$

17. (a)

Type of airport	Maximum rate of change of longitudinal gradient
A and B	0.1% per 30 m length
C type	0.2% per 30 m length
D and E type	0.4% per 30 m length

18. (a)

$$\begin{aligned} \text{Hauling capacity} &= \mu WN \\ W &= \text{Load on each driving axle} \\ \Rightarrow W &= 10 \times 2 = 20 \text{ tonnes} \\ N &= \text{Number of axles} \\ \Rightarrow N &= 3 \\ \therefore \text{Hauling capacity} &= 0.3 \times 20 \times 3 = 18 \text{ tonnes} \end{aligned}$$

19. (a)

Radius of broad gauge curve,

$$R = \frac{1146}{3} = 382 \text{ m}$$

$$e_{\text{eq}} = \frac{GV^2}{127R} = \frac{1.676 \times 70^2}{127 \times 382} = 0.169 \text{ m} > 0.165 \text{ m}$$

Adopt

$$\begin{aligned} e_{\text{eq}} &= 0.165 \text{ m} \\ e_{\text{th}} &= e_{\text{eq}} + \text{CD} \\ &= 16.5 + 7.6 = 24.1 \text{ cm} \end{aligned}$$

$$\therefore 24.1 = \frac{1.676 \times V_{\text{max}}^2}{127 \times 382} \times 100$$

$$\Rightarrow V_{\text{max}} = 83.52 \text{ kmph}$$

20. (b)

$$R = \frac{0.388w^2}{\frac{T}{2} - S}$$

$$R = \frac{0.388w^2}{\frac{T}{2} - \left[6 + \frac{\text{Trade of main landing gear}}{2} \right]}$$

$$300 = \frac{0.388 \times 35^2}{\frac{22.5}{2} - \left[6 + \frac{\text{Trade of main landing gear}}{2} \right]}$$

Trade of main landing gear = 7.332 m \approx 7.33 m

21. (d)

Crosswind is considered to select the correct orientation of runway using wind-rose diagram.

22. (b)

$$\text{Internal force developed} = (E \times T)A$$

$$F = 25 \times 10^5 \times 2 \times 10^5 \times 40 \times 50 = 100000 \text{ kg}$$

$$\text{Resistance of track} = 1000 \text{ kg/km}$$

$$\text{Length to resist at one end} = \frac{100000}{1000} = 100 \text{ km}$$

$$\text{Total breathing length required} = 200 \text{ km}$$

23. (c)

$$\begin{aligned} \text{A.R.T.} &= T_a + \frac{1}{3}(T_m - T_a) \\ &= 25 + \frac{1}{3}(35 - 25) = 28.33^\circ\text{C} \end{aligned}$$

24. (d)

$$(i) \quad \text{Turning radius, } R = \frac{V^2}{125 \times f} = \frac{50^2}{125 \times 0.15} = 133.33 \text{ m}$$

(ii) From Horonjeff's equation,

$$R = \frac{0.388W^2}{\left[\frac{T}{2} - S\right]} = \frac{0.388 \times 18^2}{\frac{22.5}{2} - \left(6 + \frac{6.5}{2}\right)} = 62.86 \text{ m}$$

(iii) For super-sonic jet,

$$R = 180 \text{ m}$$

$$\therefore R_{\min} = [\text{Maximum of (i), (ii) and (iii)}] = 180 \text{ m}$$

25. (c)

- Zero fuel weight is the sum of empty operating weight and maximum payload.
- Maximum landing weight is less than maximum takeoff weight because fuel is burned during flight.
- Maximum ramp weight is greater than maximum takeoff weight because extra fuel is also required for taxing.

