

### ANSWER KEY > Strength of Material

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1. (a)	7. (a)	13. (b)	19. (d)	25. (a)
2. (b)	8. (c)	14. (c)	20. (b)	26. (b)
3. (a)	9. (a)	15. (d)	21. (c)	27. (a)
4. (b)	10. (d)	16. (b)	22. (b)	28. (b)
5. (c)	11. (b)	17. (c)	23. (d)	29. (a)
6. (c)	12. (b)	18. (d)	24. (b)	30. (d)

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### DETAILED EXPLANATIONS

1. (a)

$$D = 20 \text{ mm (Diameter)}$$

$$L = 210 \text{ mm (Length)}$$

$$\Delta L = 0.30 \text{ mm (Expansion)}$$

$$P = 50 \text{ kN (Load)}$$

$$\text{Strain} = \frac{\Delta L}{L} = \frac{0.30}{210} = 1.428 \times 10^{-3}$$

2. (b)

$$E = 1.25 \times 10^5 \text{ N/mm}^2$$

$$\mu = 0.25$$

$$E = 2G(1 + \mu)$$

$$1.25 \times 10^5 = 2G(1 + 0.25)$$

$$G = \frac{1.25 \times 10^5}{2 \times 1.25} = 0.5 \times 10^5 \text{ N/mm}^2$$

3. (a)

Since fluid element will be subjected to hydrostatic loading therefore Mohr's circle will reduce into a point on  $\sigma$ -axis.

$$\therefore \text{Radius of Mohr circle} = 0 \text{ unit}$$

4. (b)

Strain energy due to torsion per unit volume,

$$U = \frac{\tau^2}{4G} (1+K^2)$$

where,

$$K = \frac{d}{D}$$

$d$  = inner diameter of hollow shaft

$D$  = outer diameter of hollow shaft

Given,

$$U = \frac{\tau^2}{3G}$$

$$\therefore \frac{\tau^2}{3G} = \frac{\tau^2}{4G} (1+K^2)$$

$$\Rightarrow K^2 = \frac{1}{3}$$

$$\Rightarrow K = \frac{d}{D} = \sqrt{\frac{1}{3}}$$

$$\therefore \frac{D}{d} = \sqrt{3}$$

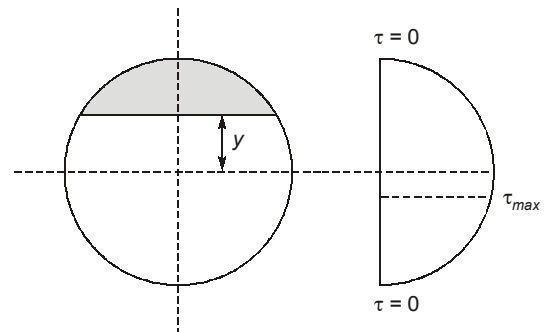
5. (c)

Shear stress in circular cross-section

$$\tau = \frac{F}{3I} (r^2 - y^2)$$

We observe

1. Variation of  $\tau$  versus  $y$  is a parabolic curve.
2.  $\tau$  increases as  $y$  decreases.
3.  $\tau = 0$  ( $y = r$ )
4. at  $y = 0$ ,  $\tau$  is maximum



$$\tau = \frac{F}{3I} (r^2 - y^2)$$

$$\tau_{y=0} = \tau_{\max} = \frac{F \times (d/2)^2}{3 \times \frac{\pi}{64} d^4} = \frac{F}{3 \left( \frac{\pi}{16} d^2 \right)}$$

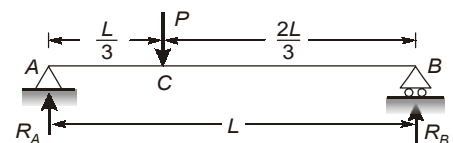
$$\Rightarrow \tau_{\max} = \frac{4}{3} \frac{F}{A} = \frac{4}{3} \tau_{\text{avg.}}$$

$$\Rightarrow \frac{\tau_{\max}}{\tau_{\text{avg}}} = \frac{4}{3} = 1.33$$

6. (c)

$$R_A + R_B = P$$

Taking moment about A



$$R_B \times L = P \times \frac{L}{3}$$

$$R_B = \frac{P}{3}$$

$$\text{Bending moment at } C = R_B \times \frac{2L}{3} = \frac{P}{3} \times \frac{2L}{3} = \frac{2PL}{9}$$

$$\frac{2 \times 50 \times 3}{9} = 33.33 \text{ Nm}$$

7. (a)

$$P = T\omega$$

So,

$$T = \frac{P}{N}$$

Now,

$$\tau = \frac{16T}{\pi d^3} = \frac{16P}{\pi N d^3}$$

So,

$$d \propto \left(\frac{P}{N}\right)^{1/3}$$

9. (a)

$$\frac{\sigma}{y} = \frac{M}{I} = \frac{E}{R}$$

$$\sigma_{\max} = \frac{E}{R} \times y_{\max} = \frac{2 \times 10^5}{10 \times 1000} \times \frac{20}{2} = 200 \text{ N/mm}^2 = 0.2 \text{ kN/mm}^2$$

10. (d)

$$\text{Area of the bar} = A = \frac{\pi}{4} \times 15^2 = 176.7145 \text{ mm}^2$$

$$\text{Stress in the bar} = \sigma = \frac{10 \times 10^3}{176.7145} = 56.59 \text{ N/mm}^2$$

Strain energy stored per unit volume

$$= \frac{\sigma^2}{2E} = \frac{56.59^2}{2 \times 2 \times 10^5} = 8.006 \times 10^{-3} \text{ N/mm}^2$$

11. (b)

$$D = 60 \text{ mm}$$

$$P = 180 \text{ kW}$$

$$f = 25 \text{ Hertz}$$

$$\text{thickness (t)} = \frac{D-d}{2}$$

$D \rightarrow$  outer diameter

$d \rightarrow$  Inner diameter

$$P = \frac{2\pi NT}{60} \text{ or } \frac{2\pi fT}{1000}$$

$$T = \frac{180 \times 1000}{2 \times \pi \times 25}$$

$$T = 1145.92 \text{ Nm}$$

By torsional rigidity:

$$\frac{\tau}{R} = \frac{T}{J} = \frac{G\theta}{L}$$

$$\frac{J}{R} = \frac{T}{\tau}$$

$\therefore$

$$\frac{J}{R} = \text{Polar section modulus}$$

$$Z_P = \frac{1145.92 \times 10^3}{60} = 19098.67 \text{ mm}^3$$

$$Z_P = \frac{\pi}{16D} (D^4 - d^4)$$

$$d = 51.66 \text{ mm}$$

$$\text{thickness} = \frac{D-d}{2} = \frac{60-51.66}{2} = 4.17 \text{ mm}$$

12. (b)

$$\frac{\tau}{R} = \frac{G\theta}{L}$$

$$\theta = 2.5^\circ = \frac{\pi}{180} \times 2.5^\circ \text{ radian}$$

$$L = 450 \text{ mm}$$

$\therefore$

$$G = \frac{\text{Shear stress}}{\text{Shear strain}}$$

$\Rightarrow$

$$R = \frac{\tau}{G} \times \frac{L}{\theta}$$

$$= \frac{0.0006 \times 450 \times 180}{\pi \times 2.5}$$

$$= 6.1879 \text{ mm}$$

$$D = 12.3758 \text{ mm}$$

$$[\text{strain} = \frac{\tau}{G} = 0.0006]$$

13. (b)

$$L = 1 \text{ m}$$

$$P = 4500 \text{ N/m}$$

$$\text{Maximum deflection} = \frac{wL^4}{384EI} = \frac{4.5 \times 1000^4}{384 \times 200000 \times 5 \times 10^7} = 1.1718 \times 10^{-3} \text{ mm}$$

14. (c)

$$\begin{aligned}\epsilon_{\theta_1} &= \epsilon_x \cos^2\theta_1 + \epsilon_y \sin^2\theta_1 + \gamma \sin\theta_1 \cos\theta_1 \\ \epsilon_{\theta_2} &= \epsilon_x \cos^2\theta_2 + \epsilon_y \sin^2\theta_2 + \gamma \sin\theta_2 \cos\theta_2 \\ \epsilon_{\theta_3} &= \epsilon_x \cos^2\theta_3 + \epsilon_y \sin^2\theta_3 + \gamma \sin\theta_3 \cos\theta_3\end{aligned}$$

$$\epsilon_{\theta'} = \epsilon_x + \epsilon_{45^\circ} = \frac{1}{2}(\epsilon_x + \epsilon_y) + \frac{\gamma}{2}\epsilon_{90^\circ} = \epsilon_y$$

$$\begin{aligned}\epsilon_x &= 400 \times 10^{-6}, \quad \epsilon_y = -100 \times 10^{-6}, \quad \gamma = 2\epsilon_{45^\circ} - \epsilon_x - \epsilon_y \\ \gamma &= 2 \times 200 \times 10^{-6} - (400 - 100) \times 10^{-6} = 100 \times 10^{-6}\end{aligned}$$

$$\epsilon_{1,2} = \frac{\epsilon_x + \epsilon_y}{2} \pm \frac{1}{2}\sqrt{(\epsilon_x - \epsilon_y)^2 + \gamma^2}$$

$$\epsilon_{1,2} = \frac{10^{-6}}{2} \left[ (400 - 100) \pm \sqrt{(500)^2 + (100)^2} \right]$$

$$\epsilon_{1,2} = 404.95 \times 10^{-6} \text{ and } -104.95 \times 10^{-6}$$

$$\sigma_1 = \frac{E(\nu \epsilon_2 + \epsilon_1)}{1 - \nu^2} = \frac{210 \times 10^3 (-0.3 \times 104.95 + 404.95) \times 10^{-6}}{1 - (0.3)^2}$$

$$\sigma_1 = 86.2 \text{ MPa}$$

$$\sigma_2 = \frac{E(\nu \epsilon_1 + \epsilon_2)}{1 - \nu^2} = \frac{210 \times 10^3 (0.3 \times 404.95 - 104.95) \times 10^{-6}}{1 - (0.3)^2}$$

$$\sigma_2 = 3.82 \text{ MPa}$$

$$\text{Major principal stress} = 86.2 \text{ MPa}$$

15. (d)

$$\text{Specific weight, } w = \rho g = 10.2 \text{ kN/m}^3$$

$$\text{Pressure at the given depth, } P = \rho gh$$

$$P = 10.2 \times 600 = 6120 \text{ kN/m}^2$$

$$P = 6.12 \text{ N/mm}^2$$

$$\text{Bulk modulus, } K = 160 \times 10^3 \text{ N/mm}^2$$

$$\text{Volumetric strain, } \epsilon_v = \frac{P}{K} = \frac{6.12}{160 \times 10^3} = 3.825 \times 10^{-5}$$

$$\text{Initial volume, } V = \frac{4}{3}\pi R^3 = \frac{\pi D^3}{6} = \frac{3.14 \times 160^3}{6} = 2.144 \times 10^6 \text{ mm}^3$$

$$\frac{\Delta V}{V} = \epsilon_v$$

or

$$\Delta V = 2.144 \times 10^6 \times 3.825 \times 10^{-5}$$

$$\text{Change in volume, } \Delta V = 82 \text{ mm}^3$$

16. (b)

$$\Delta L_{AC} = \Delta L_{CB}$$

$$\frac{R_A l}{AE} = \frac{2R_B l}{AE}$$

⇒

$$R_A = 2R_B$$

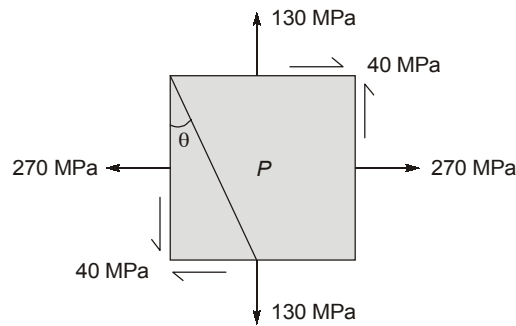
Also

$$R_A + R_B = 10$$

⇒

$$R_A = \frac{20}{3} \text{ kN} = 6.67 \text{ kN}$$

17. (c)



$$\begin{aligned}\sigma_{1,2} &= \frac{\sigma_x + \sigma_y}{2} \pm \frac{1}{2} \sqrt{(\sigma_x - \sigma_y)^2 + 4\tau_{xy}^2} \\ &= \frac{270 + 130}{2} \pm \frac{1}{2} \sqrt{(270 - 130)^2 + 4 \times 40^2} \\ &= 200 \pm \frac{1}{2} \sqrt{140^2 + 4 \times 1600} \\ &= 200 \pm 80.62 \text{ MPa} \\ \sigma_1 &= 280.62 \text{ MPa} \\ \sigma_2 &= 119.38 \text{ MPa}\end{aligned}$$

$$\text{Maximum shearing stress} = \frac{1}{2}(\sigma_1 - \sigma_2) = \frac{1}{2}(280.62 - 119.38) = 80.62 \text{ MPa}$$

18. (d)

$$\text{Thermal Stress} = \alpha E \Delta t = (12 \times 10^{-6})(200 \times 10^3)(120 - 20) = 240 \text{ MPa}$$

19. (d)

Hoop stress or circumferential stress is given by  $\frac{pd}{2t} = \sigma_c$  longitudinal stress is given by  $\frac{pd}{4t} = \sigma_l$

$$\text{Volumetric strain} = \frac{dV}{V} = \frac{2d(d)}{d} + \frac{dl}{l}$$

$$\epsilon_v = 2\epsilon_d + \epsilon_l$$

Also,

$$\epsilon_d = \epsilon_c = \frac{\sigma_c}{E} - \frac{\nu\sigma_l}{E}$$

and

$$\epsilon_l = \frac{\sigma_l}{E} - \frac{\nu\sigma_c}{E}$$

$$\begin{aligned}\therefore \epsilon_v &= \frac{1}{E} [2\sigma_c - 2\nu\sigma_l + \sigma_l - \nu\sigma_c] \\ &= \frac{1}{E} [(2 - \nu)\sigma_c + (1 - 2\nu)\sigma_l]\end{aligned}$$

Here,

$$\sigma_c = \frac{pd}{2t} = \frac{3 \times 7.5}{2 \times 0.15} = 75 \text{ MPa}$$

$$\sigma_l = \frac{pd}{4t} = 37.5 \text{ MPa}$$

and  $\nu = 0.3$

$$\epsilon_v = \frac{1}{2 \times 10^5} [1.7 \times 75 + 0.4 \times 37.5] = 0.7125 \times 10^{-3}$$

$$\therefore \quad \% \text{ change in volume} = 0.07\%$$

20. (b)

Modulus of resilience = Area under the  $\sigma - \epsilon$  curve till the yield point.

$$\text{So,} \quad \text{MOR} = \frac{1}{2} \times 2 \times 14 = 14$$

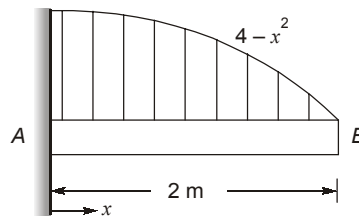
Modulus of toughness = Area under  $\sigma - \epsilon$  curve till the fracture point.

$$\text{MOT} = \frac{1}{2} \times 2 \times 14 + \frac{1}{2} \times 4 \times (20 + 14) + \frac{1}{2} \times 2 \times (20 + 12)$$

$$\text{MOT} = 14 + 68 + 32 = 114$$

$$\text{So,} \quad \frac{\text{MOR}}{\text{MOT}} = \frac{14}{114} = 0.1228$$

21. (c)



According to the question,

$$\frac{dM}{dx} = 4 - x^2$$

$$\text{So,} \quad \int_A^B dM = \int_0^2 (4 - x^2) dx$$

$$\text{So,} \quad M_B - M_A = 8 - \frac{8}{3} = \frac{24 - 8}{3} = \frac{16}{3} = 5.33$$

22. (b)

Outer radius =  $R$

Inner radius =  $R - t$

So, Approximate polar moment of inertia

$$J = 2\pi R^3 t$$

Now, we know,

$$\frac{T}{J} = \frac{\tau}{r}$$

For  $\tau_{\max}$ ,  $r = R$

$$\text{So,} \quad T = \frac{\tau_{\max} J}{R} = 2\pi R^2 t \tau_{\max}$$

Now, weight of tube ( $W$ ) =  $\gamma LA$ , where  $A$  is the cross-section area,

So,  $A \approx 2\pi Rt$

So,  $W = 2\pi RtL\gamma$

$$\therefore \frac{T}{W} = \frac{2\pi R^2 t \tau_{\max}}{2\pi RtL\gamma} = \frac{R\tau_{\max}}{L\gamma}$$

**23. (d)**

From the given figure,

$$\Rightarrow b^2 + d^2 = (25^2)$$

$$\Rightarrow d^2 = 625 - b^2 \quad \dots(1)$$

Now for the strongest section, section modulus  $Z$  must be maximum.

$$\Rightarrow Z = \frac{I}{Y} = \frac{bd^3}{12 \times d/2} = \frac{bd^2}{6}$$

$$\Rightarrow = \frac{b}{6}(625 - b^2) = \frac{625}{6}b - \frac{b^3}{6}$$

For  $Z_{\max}$ ,  $\frac{dZ}{db} = 0$

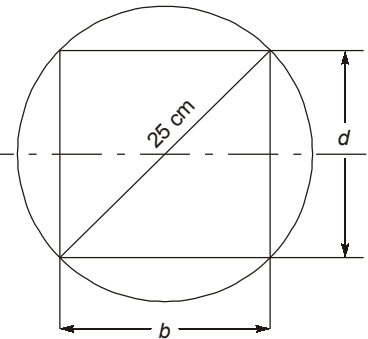
$$\Rightarrow \frac{625}{6} - \frac{3b^2}{6} = 0$$

$$\Rightarrow b^2 = \frac{625}{3}$$

$$\Rightarrow b = 14.4 \text{ cm}$$

From equation (1)

$$\Rightarrow d = 20.41 \text{ cm}$$



**24. (b)**

Outer radius =  $R$

Inner radius =  $R - t$

So, Approximate polar moment of inertia

$$J = 2\pi R^3 t$$

Now, we know,

$$\frac{T}{J} = \frac{\tau}{r}$$

For  $\tau_{\max}$ ,  $r = R$

So,  $T = \frac{\tau_{\max} J}{R} = 2\pi R^2 t \tau_{\max}$

Now, weight of tube ( $W$ ) =  $\gamma LA$ , where  $A$  is the cross-section area,

So,  $A \approx 2\pi Rt$

So,  $W = 2\pi RtL\gamma$

$$\therefore \frac{T}{W} = \frac{2\pi R^2 t \tau_{\max}}{2\pi RtL\gamma} = \frac{R\tau_{\max}}{L\gamma}$$



25. (a)

$$\begin{aligned} \frac{P_2}{P_1} &= \frac{I_2}{I_1} = \left(\frac{d_2}{d_1}\right)^4 \\ &= (0.8)^4 = 0.4096 \\ \frac{P_1 - P_2}{P_1} &= 1 - 0.4096 = 0.5904 = 59.04\% \end{aligned}$$

26. (b)

The strain energy per unit volume may be given as

$$\begin{aligned} u &= \frac{1}{2} \times \frac{\sigma_y^2}{E} = \frac{1}{2} \times \frac{(250)^2}{2 \times 10^5} \\ &= 0.156 \text{ Nmm/mm}^3 \end{aligned}$$

27. (a)

$$\begin{aligned} K &= \frac{\text{Direct stress}}{\text{Volumetric strain}} \\ \frac{dV}{V} &= \frac{r^3 - (r - \Delta r)^3}{r^3} = 1 - \left\{1 - \frac{\Delta r}{r}\right\}^3 = 1 - \left\{1 - \frac{0.55}{2.5}\right\}^3 \\ &= 6.5985 \times 10^{-4} \\ K &= \frac{250}{6.5985 \times 10^{-4}} = 378,871.2243 \text{ MPa} \\ &= 378.871 \text{ GPa} \end{aligned}$$

28. (b)

$$\begin{aligned} M &= 80 \cdot x - 64(x - 1) \quad \forall x \in (1, 4) \\ \text{At centre} \quad x &= 4 \text{ m} \\ M &= (80 \times 4) - 64(3) = 128 \text{ kNm} \end{aligned}$$

29. (a)

Given,

$$\begin{aligned} \sigma_1 &= 100 \text{ MPa} \\ \sigma_2 &= 50 \text{ MPa}, \\ \sigma_3 &= 25 \text{ MPa} \\ S_{yt} &= 220 \text{ MPa}, \end{aligned}$$

For maximum shear strain energy theory,

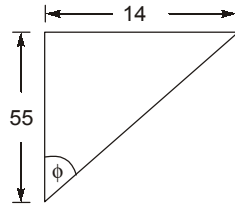
$$(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 \leq 2 \left(\frac{S_{yt}}{N}\right)^2 \quad [\text{Where, } N = \text{factor of safety}]$$

$$(100 - 50)^2 + (50 - 25)^2 + (25 - 100)^2 = 2 \left(\frac{220}{N}\right)^2$$

After solving,

$$\therefore \text{Factor of safety, } N = 3.326 \simeq \mathbf{3.33}$$

30. (d)



$$\text{Shear strain, } \gamma = \tan \phi = \frac{14}{55} = 0.2545$$

$$\tau = \frac{P}{ab} = \frac{16000}{150 \times 225} = 0.474 \text{ MPa}$$

$$\text{Shear modulus, } G = \frac{\tau}{\gamma} = \frac{0.474}{0.2545} = 1.86 \text{ MPa}$$

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