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STRENGTH OF MATERIALS

CIVIL ENGINEERING

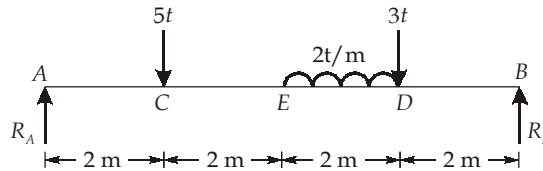
Date of Test : 14/10/2025

ANSWER KEY ➤

1. (a)	7. (b)	13. (c)	19. (d)	25. (a)
2. (c)	8. (d)	14. (c)	20. (a)	26. (a)
3. (a)	9. (c)	15. (b)	21. (a)	27. (b)
4. (b)	10. (a)	16. (a)	22. (c)	28. (d)
5. (c)	11. (c)	17. (b)	23. (d)	29. (c)
6. (d)	12. (b)	18. (b)	24. (b)	30. (b)

DETAILED EXPLANATIONS

1. (a)



Taking moments about B,

$$R_A = \left(\frac{6}{8}\right) \times 5 + \frac{3}{8} \times 4 + 3 \times \frac{2}{8}$$

$$= 3.75 + 1.5 + 0.75 = 6t$$

Taking moment about A,

$$R_B = (5 + 3 + 2 \times 2) - R_A$$

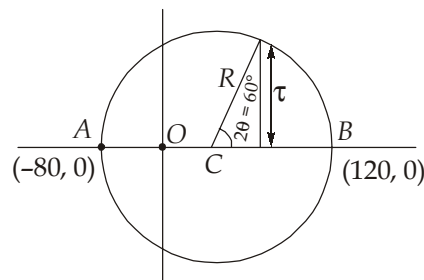
$$= 12t - 6t = 6t$$

$$\therefore \frac{R_A}{R_B} = \frac{6t}{6t} = 1$$

2. (c)

Radius,

$$R = \frac{120 - (-80)}{2} = 100$$



Tangential stress,

$$\tau = R \sin 2\theta$$

$$= 100 \sin 60^\circ$$

$$= 100 \times \frac{\sqrt{3}}{2}$$

$$= 50\sqrt{3}$$

3. (a)

The value of shear stress at the top edge and bottom edge of any section will always be zero.

4. (b)

In flitched beam has a composite section made of two or more materials joined together in such a manner that they behave as a unit piece and each material bends to the same radius of curvature. The total moment of resistance of a flitched beam is equal to the sum of the moments of resistance of individual sections.

5. (c)

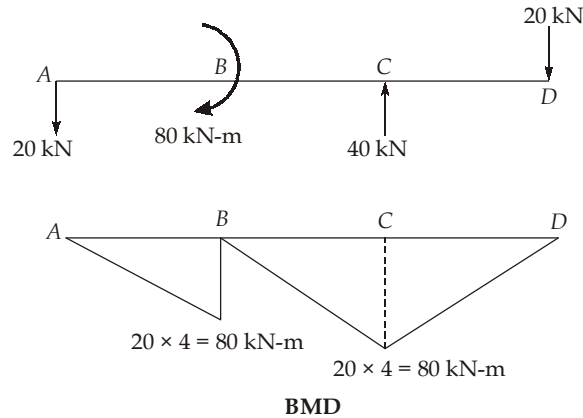
The stress induced in metal-1 due to restriction = $E_1 \alpha_1 \Delta T$

So, force required in metal-1 = $E_1 \alpha_1 \Delta T \times A_1$

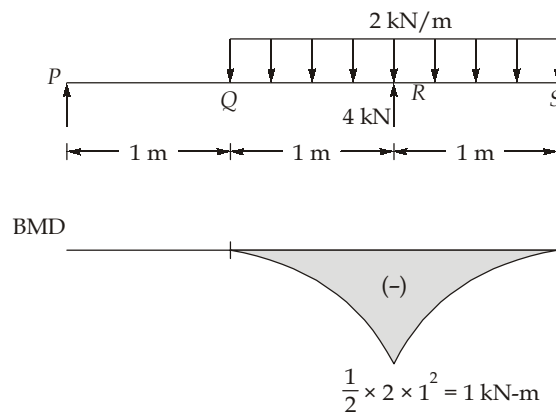
Similarly for metal-2, force required = $E_2 \alpha_2 \Delta T \times A_2$

So, Total force required = $E_1 \alpha_1 \Delta T A_1 + E_2 \alpha_2 \Delta T A_2$
= $(E_1 \alpha_1 A_1 + E_2 \alpha_2 A_2) \Delta T$

6. (d)



7. (b)



8. (d)

As we know, $\frac{dV}{dx} = w$

$$\Rightarrow w = \frac{d}{dx}(5x^2) = 10x$$

For midspan, $x = 1 \text{ m}$

So, load intensity, $w = 10 \text{ N/m}$

9. (c)

$$\text{Deflection due to load} = \frac{wl^4}{8EI} = \frac{10 \times (3000)^4}{8 \times 5 \times 10^{11}} = 202.5 \text{ mm}$$

Since gap is only 3 mm.

$$\therefore 202.5 - 3 = \frac{Rl^3}{3EI}$$

$$\Rightarrow \frac{(202.5 - 3) \times 3 \times 5 \times 10^{11}}{(3000)^3} = R$$

$$\Rightarrow R = 11.083 \text{ kN} \simeq 11.08 \text{ kN}$$

10. (a)

$$V = \frac{\pi}{4} D^2 l = \frac{\pi}{4} (120)^2 (1.5 \times 10^3) = 16.96 \times 10^6 \text{ mm}^3$$

So strain energy stored in shaft

$$\begin{aligned} U &= \frac{\tau^2}{4G} \text{ Volume} \\ &= \frac{50^2}{4 \times 80 \times 10^3} \times 16.96 \times 10^6 \\ &= 132.5 \times 10^3 \text{ N-mm} \\ &= 132.5 \text{ N-m} \end{aligned}$$

11. (c)

$$\text{Rankine's crippling load} = \frac{\sigma_{cs} A}{1 + a \left(\frac{l_e}{k} \right)^2}$$

As both ends are hinged

$$\text{So } l_e = l = 2.3 \text{ m}$$

$$\begin{aligned} \therefore P_R &= \frac{335 \times 88.75 \pi}{1 + \frac{1}{7500} \left[\frac{2.3 \times 10^3}{12.6} \right]^2} \\ &= 17161.04 \text{ N} = 17.16 \text{ kN} \end{aligned}$$

12. (b)

Since section is symmetric about $x-x$ and $y-y$, therefore centre of section will lie on the geometrical centroid of section.

The semi-circular grooves may be assumed together and consider one circle of diameter 60 mm.

$$\begin{aligned} \text{So, } I_{xx} &= \frac{80 \times (100)^3}{12} - \frac{\pi}{64} (60)^4 \\ &= 6.03 \times 10^6 \text{ mm}^4 \end{aligned}$$

Now for shear stress at neutral axis, consider the area above the neutral axis,

$$\begin{aligned} A\bar{y} &= [80 \times 50 \times 25] - \frac{\pi}{2} (30)^2 \times \frac{4 \times 30}{3\pi} \\ &= 100000 - 18000 = 82000 \text{ mm}^3 \\ b &= 20 \text{ mm} \end{aligned}$$

So,

$$\begin{aligned} \tau &= \frac{VA\bar{y}}{Ib} = \frac{20 \times 10^3 \times 82000}{6.03 \times 10^6 \times 20} \\ &= 13.60 \text{ MPa} \end{aligned}$$

13. (c)

Area of tube,

$$A = \frac{\pi}{4} [40^2 - 25^2] = 765.8 \text{ mm}^2$$

$$I = \frac{\pi}{64} [D^4 - d^4] = \frac{\pi}{64} [40^4 - 25^4] = 106488.9 \text{ mm}^4$$

We also know strain in alloy tube is,

$$\epsilon = \frac{\delta l}{l} = \frac{4.8}{4 \times 10^3} = 0.0012$$

Modulus of elasticity for the alloy

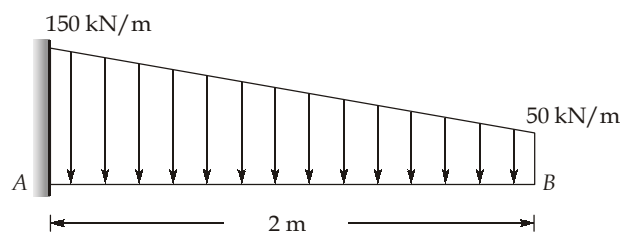
$$\begin{aligned} E &= \frac{\text{Load}}{\text{Area} \times \text{Strain}} = \frac{60 \times 10^3}{765.8 \times 0.0012} \\ &= 65291.2 \text{ N/mm}^2 \end{aligned}$$

Since, column is pinned at its both ends, therefore equivalent length of the column

$$L_e = l = 4 \times 10^3 \text{ mm}$$

$$\begin{aligned} \text{Euler's buckling load, } P_E &= \frac{\pi^2 EI}{l_e^2} = \frac{\pi^2 \times 65291.2 \times 106488.9}{(4 \times 10^3)^2} \text{ N} \\ &= 4.29 \text{ kN} \end{aligned}$$

14. (c)



$$I = \frac{bd^3}{12} = \frac{150 \times (200)^3}{12} = 10^8 \text{ mm}^4$$

We will split this load distribution i.e. trapezoidal load into a uniformly distributed load (w_1) of 50 kN/m and a triangular load (w_2) of 100 kN/m at A to zero at B.

So, deflection at free end

$$y_B = \frac{w_1 l^4}{8EI} + \frac{w_2 l^4}{30EI}$$

$$\begin{aligned}
 &= \frac{50 \times (2 \times 10^3)^4}{8 \times (100 \times 10^3) \times 10^8} + \frac{100 \times (2 \times 10^3)^4}{30 \times (100 \times 10^3) \times 10^8} \\
 &= 10 + 5.3 = 15.3 \text{ mm}
 \end{aligned}$$

15. (b)

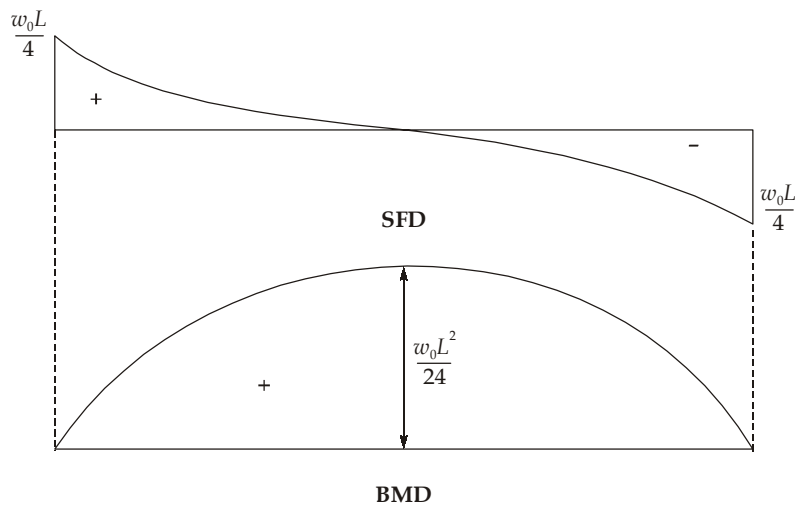
$$\text{Total load} = 2 \left[\frac{1}{2} \left(\frac{L}{2} \right) \times w_0 \right] = \frac{w_0 L}{2}$$

By symmetry, $R_1 = R_2 = \frac{1}{2} \times \text{Total load}$

$$\Rightarrow R_1 = R_2 = \frac{w_0 L}{4}$$

Bending moment at B,

$$\begin{aligned}
 (M_B) &= R_1 \times \frac{L}{2} - \frac{1}{2} w_0 \times \frac{L}{2} \times \frac{2}{3} \left(\frac{L}{2} \right) \\
 &= \frac{w_0 L}{4} \times \frac{L}{2} - \frac{w_0}{2} \times \frac{L}{2} \times \frac{2}{3} \left(\frac{L}{2} \right) \\
 &= \frac{w_0 L^2}{8} - \frac{w_0 L^2}{12} = \frac{w_0 L^2}{24}
 \end{aligned}$$



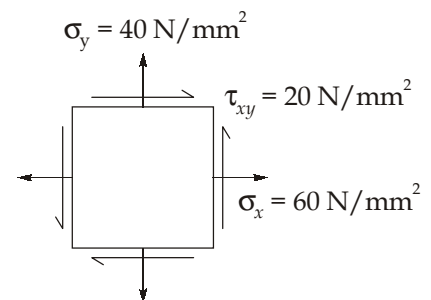
16. (a)

$$\begin{aligned}
 \sigma_{1/2} &= \frac{P_1 + P_2}{2} \pm \sqrt{\left(\frac{P_1 - P_2}{2} \right)^2 + q^2} \\
 &= \frac{60 + 40}{2} \pm \sqrt{\left(\frac{60 - 40}{2} \right)^2 + 20^2} \\
 &= 50 \pm 22.36
 \end{aligned}$$

$$\therefore \sigma_1 = 72.36 \text{ N/mm}^2 \text{ (T)}$$

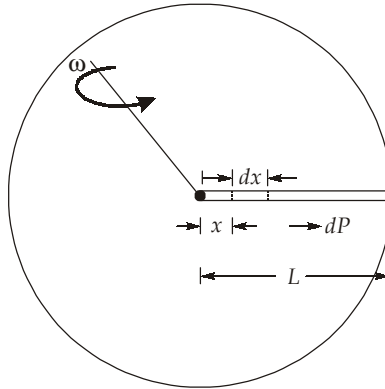
$$\text{and } \sigma_2 = 27.64 \text{ N/mm}^2 \text{ (T)}$$

$$\tan 2\theta_p = \frac{2q}{P_1 - P_2} = \frac{2 \times 20}{60 - 40} = 2$$



$$\begin{aligned}\therefore 2\theta_p &= 63.43^\circ = 63^\circ 26' \\ \therefore \theta_{p_1} &= 31^\circ 43' \text{ and } \theta_{p_2} = 121^\circ 43' \\ \therefore \tau_{\max} &= \frac{\sigma_1 - \sigma_2}{2} = \frac{72.36 - 27.64}{2} = 22.36 \text{ N/mm}^2 \\ \theta_{s_1} &= 45^\circ + \theta_{p_1} = 45^\circ + 31^\circ 43' = 76^\circ 43'\end{aligned}$$

17. (b)



$$\delta = \frac{PL}{AE}$$

From the figure,

$$d\delta = \frac{dPx}{AE}$$

Centrifugal force on differential mass dM ,

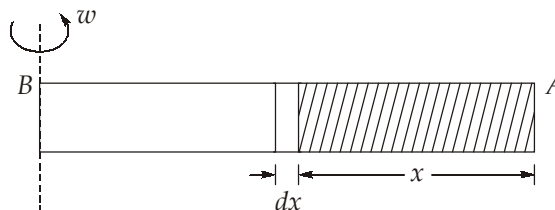
$$dP = dM \cdot \omega^2 x = (\rho A dx) \omega^2 x$$

$$\therefore d\delta = \frac{(\rho A \omega^2 x dx) x}{AE}$$

$$\therefore \delta = \frac{\rho \omega^2}{E} \int_0^L x^2 \cdot dx = \frac{\rho \omega^2}{E} \left[\frac{x^3}{3} \right]_0^L$$

$$\Rightarrow \delta = \frac{\rho \omega^2}{3E} [L^3 - 0^3] = \frac{\rho \omega^2 L^3}{3E}$$

Alternate Solution:



$$m_x = \rho Ax$$

Distance between the C.G. of mass to the centre of rotation,

$$r = L - \frac{x}{2}$$

Centrifugal force, $F = m_x w^2 r = \rho A x \left(w^2 \right) \left(L - \frac{x}{2} \right)$

Element elongation, $d\delta = \frac{F dx}{AE} = \frac{A x \rho \left(L - \frac{x}{2} \right) w^2 dx}{AE}$

$$d\delta = \frac{\rho w^2 \left(L - \frac{x}{2} \right)}{E} x dx$$

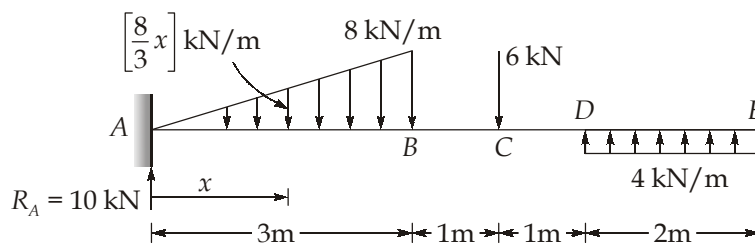
Total elongation, $\delta = \int_0^L \frac{\rho w^2}{E} \left(L - \frac{x}{2} \right) x dx$

$$= \frac{\rho w^2}{E} \left[L \left(\frac{L^2}{2} \right) - \left(\frac{x^3}{6} \right)_0^L \right]$$

$$= \frac{\rho w^2}{E} \left[\frac{L^2}{2} - \frac{L^3}{6} \right]$$

$$\delta = \frac{\rho w^2 L^2}{3E}$$

18. (b)



Let at a distance 'x' from A, shear force is zero

$$\therefore SF_x = R_A - \frac{1}{2}(x) \frac{8}{3} x = 0$$

$$\Rightarrow 10 = \frac{4x^2}{3}$$

$$\Rightarrow x = 2.7386 \text{ m}$$

$\therefore$ Beam at distance 'x' (BM_x)

$$= 10x - \frac{1}{2}x \left(\frac{8}{3}x \right) \frac{x}{3}$$

$$= 10(2.7386) - \frac{4}{9}(2.7386)^3$$

$$= 18.26 \text{ kNm}$$

19. (d)

Change in temperature (fall) = $400 - 300 = 100^\circ\text{C}$

Area of cross-section of rod = $\frac{\pi}{4}(20)^2 = 314.16 \text{ mm}^2$

$\therefore$ Wall yield by 5 mm

$$\therefore \frac{p_t L}{E} = \Delta = (L\alpha t - a), \quad \text{where } a = 5 \text{ mm}$$

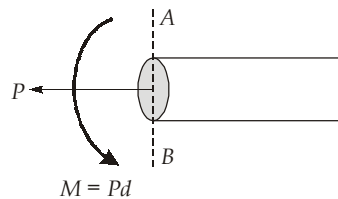
$$\begin{aligned} \Rightarrow p_t &= \frac{E}{L}(L\alpha t - 5) = E\alpha t - \frac{5E}{L} \\ &= \left(2 \times 10^5 \times 12 \times 10^{-6} \times 100\right) - \frac{5 \times 2 \times 10^5}{8000} \\ &= 240 - 125 = 115 \text{ N/mm}^2 \\ \therefore \text{ Pull, } P &= p A = 115 \times 314.16 = 36128 \text{ N} = 36.128 \text{ kN} \end{aligned}$$

20. (a)

Case 1:

Maximum stress, $\sigma_1 = \frac{P}{A} = \frac{4P}{\pi d^2}$

Case 2:



$$\begin{aligned} \text{Maximum stress, } \sigma_2 &= \frac{P}{A} + \frac{My}{I} \\ \Rightarrow \sigma_2 &= \frac{4P}{\pi d^2} + \frac{(Pd) \times (d/2)}{\frac{\pi}{64} d^4} = \frac{4P}{\pi d^2} + \frac{32P}{\pi d^2} = \frac{36P}{\pi d^2} \\ &= 9 \left(\frac{4P}{\pi d^2} \right) = 9\sigma_1 = n\sigma_1 \\ \therefore n &= 9 \end{aligned}$$

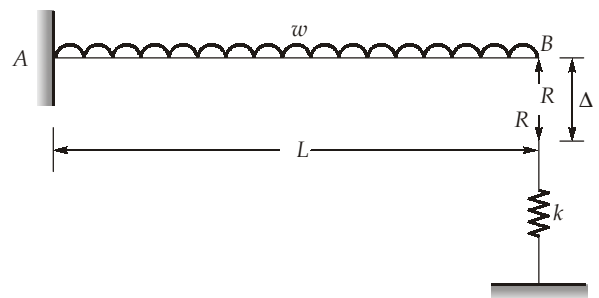
21. (a)

$$\Delta_B = \frac{R}{k} = \frac{wl^4}{8EI} - \frac{Rl^3}{3EI}$$

Since, $k = \frac{EI}{l^3}$

$$\therefore \frac{Rl^3}{EI} + \frac{Rl^3}{3EI} = \frac{wl^4}{8EI}$$

$$\Rightarrow R = \frac{3wL}{32}$$

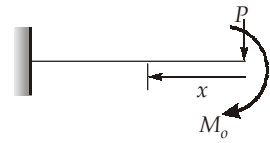


22. (c)

$$\text{Strain energy due to bending} = \int \frac{M_x^2 dx}{2EI}$$

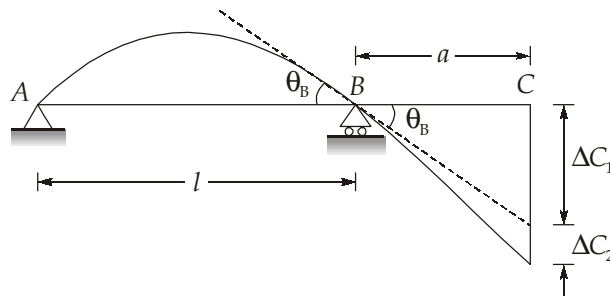
Since it varies with square, we can't apply principle of superposition.

$$\begin{aligned} \int_0^L \frac{(-Px - M_0)^2 dx}{2EI} &= \int_0^L \frac{P^2 x^2 + M_0^2 + 2PxM_0}{2EI} dx \\ &= \frac{1}{2EI} \left[\frac{P^2 x^3}{3} + M_0^2 x + PM_0 x^2 \right]_0^L \\ &= \frac{P^2 L^3}{6EI} + \frac{M_0^2 L}{2EI} + \frac{PM_0 L^2}{2EI} \end{aligned}$$



23. (d)

The deformation of the beam will be as shown below.



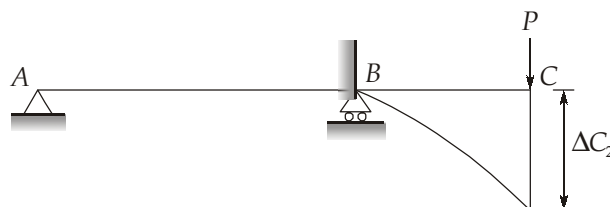
Now ΔC_1 is produced due to deflection of C as caused due to deformation of AB,

$$\Delta C_1 = \theta_B (BC) = \theta_B a$$

$$\theta_B = \frac{M_{BA} l}{3EI} = \frac{Pal}{3EI}$$

$$\therefore \Delta C_1 = \frac{Pala}{3EI} = \frac{Pa^2 l}{3EI}$$

ΔC_2 is produced due to deformation of BC

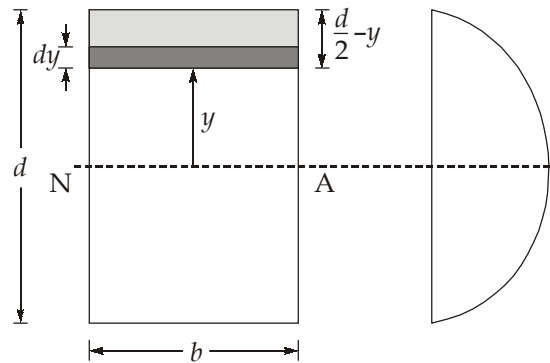


$$\Delta C_2 = \frac{Pa^3}{3EI}$$

So total deflection at C, $\Delta C = \Delta C_1 + \Delta C_2$

$$= \frac{Pa^2 l}{3EI} + \frac{Pa^3}{3EI}$$

24. (b)



Shear stress at 'y' distance from neutral axis.

$$\tau = \frac{VQ}{It}$$

Where

$$Q = A\bar{y} = \left(\frac{d}{2} - y\right)b \times \left(y + \frac{\frac{d}{2} - y}{2}\right)$$

$$\Rightarrow Q = \left(\frac{d}{2} - y\right)b \left(\frac{\frac{d}{2} + y}{2}\right)$$

$$\Rightarrow Q = \left(\frac{d^2}{4} - y^2\right)\frac{b}{2}$$

$$I = \frac{bd^3}{12}$$

So,

$$\tau = \frac{V\left(\frac{d^2}{4} - y^2\right)\frac{b}{2}}{\frac{bd^3}{12} \times b} = \frac{6V}{d^3} \left(\frac{d^2}{4} - y^2\right)$$

Now shear force carried by elementary portion

$$dF = \tau dA$$

$$= \tau b dy$$

$$dF = \frac{6V}{d^3} \left(\frac{d^2}{4} - y^2\right) dy$$

So, shear force carried by upper 1/3rd portion:

$$F = \int_{d/6}^{d/2} \frac{6V}{d^3} \left(\frac{d^2}{4} - y^2\right) dy = \frac{6V}{d^3} \left[\frac{d^2}{4} y - \frac{y^3}{3} \right]_{d/6}^{d/2}$$

$$= \frac{6V}{d^3} \left[\frac{d^2}{4} \cdot \frac{d}{2} - \frac{d^3}{24} - \left(\frac{d^2}{4} \cdot \frac{d}{6} - \frac{d^3}{216 \times 3} \right) \right] = \frac{6V}{d^3} \times \frac{7d^3}{162}$$

$$\therefore F = \frac{7V}{27}$$

25. (a)

Let D = Diameter of shaft (in mm)We know that power transmitted by shaft (P)

$$= \frac{2\pi NT}{60}$$

$$\Rightarrow 100 = \frac{2\pi NT}{60} = \frac{2\pi \times 160 \times T}{60}$$

$$\Rightarrow T = 5.968 \text{ kNm}$$

$$\begin{aligned} \therefore \text{Maximum torque, } T_{max} &= 1.2 T \\ &= 1.2 \times 5.968 \\ &= 7.162 \text{ kNm} \end{aligned}$$

$$T_{max} = f_{max} \frac{\pi}{16} D^3$$

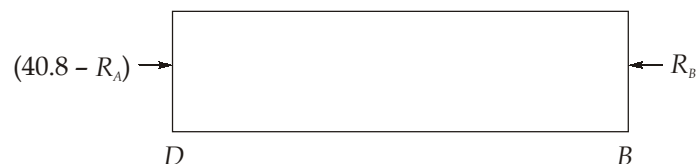
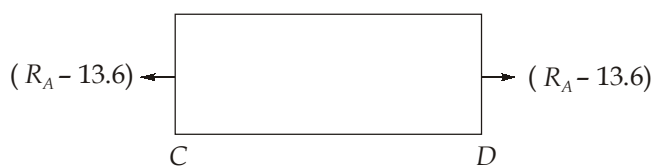
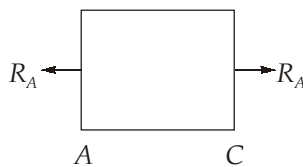
$$\Rightarrow 7.162 \times 10^6 = \frac{\pi}{16} \times \tau \times D^3 = \frac{\pi}{16} \times 70 \times D^3$$

$$\Rightarrow D^3 = \frac{7.162 \times 10^6 \times 16}{\pi \times 70}$$

$$\Rightarrow D = (521082.378)^{1/3} = 80.57 \text{ mm}$$

26. (a)

Free body diagram is shown below,



$$R_A + R_B = 40.8 \quad \dots(i)$$

$$\text{Total change in length} = 0$$

$$\text{So, } \delta_{AC} + \delta_{CD} + \delta_{DB} = 0$$

$$\frac{R_A \times 100}{AE} + \frac{(R_A - 13.6) \times 200}{AE} - \frac{(-R_A + 40.8) \times 300}{AE} = 0$$

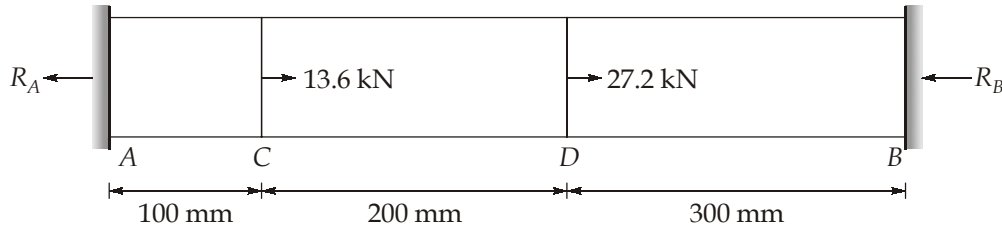
$$\Rightarrow R_A + 2R_A - 27.2 + 3R_A - 122.4 = 0$$

$$\Rightarrow R_A = 24.93 \text{ kN}$$

$$R_B = (40.8 - R_A) = -15.87 \text{ kN}$$

So ratio of magnitude of reactions i.e., $\frac{R_A}{R_B} = \frac{24.93}{15.87} = 1.57$

Alternatively:



$$R_A = \frac{13.6 \times 500}{600} + \frac{27.2 \times 300}{600} = 24.93 \text{ kN}$$

$$R_B = \frac{27.2 \times 300}{600} + \frac{13.6 \times 100}{600} = 15.87 \text{ kN}$$

$$\therefore \text{Ratio of } \left(\frac{R_A}{R_B} \right) = \frac{24.93}{15.87} = 1.57$$

27. (b)

Let the forces induced in aluminium and steel rods be P_a and P_s respectively.

Now, $P_a + P_s = W$

Balancing moment about the point where aluminum rod is hinged we have,

$$W \times 1 = P_s \times 3$$

$$\Rightarrow P_s = \frac{W}{3}$$

But $P_a + P_s = W$

$$\Rightarrow P_a = W - \frac{W}{3} = \frac{2W}{3}$$

For the beam to remain horizontal, elongation in both the rods must be equal in the vertical direction.

$$\frac{P_a l_a}{A_a E_a} = \frac{P_s l_s}{A_s E_s}$$

$$\Rightarrow \frac{A_a}{A_s} = \frac{P_a l_a E_s}{P_s l_s E_a}$$

$$\Rightarrow \frac{A_a}{A_s} = \left(\frac{2W}{3} \right) \times \frac{2000}{(W/3)} \times \frac{200}{3000} \times \frac{70}{70}$$

$$\Rightarrow \frac{A_a}{A_s} = 3.8095 \approx 3.8$$

28. (d)

Strongest beam is the one for which section modulus Z is maximum.

Let rectangular section is of width b and depth d ,

Diameter of cylindrical log, $D = 300$ mm

$$\therefore b^2 + d^2 = D^2$$

$$\therefore Z = \frac{bd^2}{6} = \frac{b(D^2 - b^2)}{6} = \frac{bD^2 - b^3}{6}$$

For maximum Z ,

$$\frac{\partial Z}{\partial b} = 0$$

$$\Rightarrow \frac{D^2 - 3b^2}{6} = 0$$

$$\Rightarrow b = \frac{D}{\sqrt{3}}$$

$$\Rightarrow b = \frac{300}{\sqrt{3}} = 173.2 \text{ mm}$$

29. (c)

$$\text{Maximum principle stress} \leq \frac{f_y}{\text{FOS}}$$

$$\begin{aligned} \text{Equivalent BM, } M_e &= \frac{1}{2} \left[M + \sqrt{M^2 + T^2} \right] \\ &= \frac{1}{2} \left[20 + \sqrt{20^2 + 40^2} \right] \\ &= 32.36 \text{ kNm} \end{aligned}$$

$$\text{Maximum principle stress} = \frac{32 M_e}{\pi D^3} \leq \frac{250}{2}$$

$$\Rightarrow \frac{32 \times 32.36 \times 10^6}{\pi D^3} \leq 125$$

$$\Rightarrow D \geq 138.15 \text{ mm}$$

30. (b)

Let, P_s = Load shared by steel rod

P_c = Load shared by copper rod

Taking moments about A,

$$P_s \times 1 + P_c \times 3 = 20 \times 4$$

$$\Rightarrow P_s + 3P_c = 80 \quad \dots(i)$$

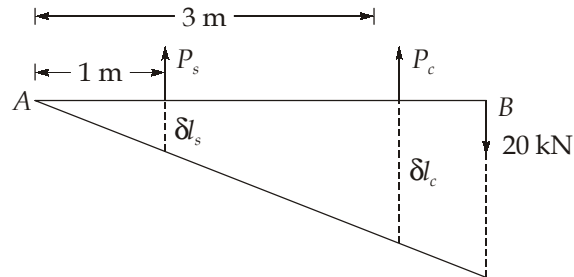
Now deformation of steel rod due to load P_s is,

$$\delta l_s = \frac{P_s l_s}{A_s E_s} = \frac{P_s \times 1 \times 10^3}{200 \times 200 \times 10^3} = 0.025 \times 10^{-3} P_s$$

And deformation of copper rod due to load P_c is,

$$\delta l_c = \frac{P_c l_c}{A_c E_c} = \frac{P_c \times 2 \times 10^3}{400 \times 100 \times 10^3} = 0.05 \times 10^{-3} P_c$$

From the geometry of elongation of the steel rod and copper rod,



$$\frac{\delta l_c}{3} = \delta l_s$$

$$\Rightarrow \delta l_c = 3 \delta l_s$$

$$\Rightarrow 0.05 \times 10^{-3} P_c = 3 \times 0.025 \times 10^{-3} P_s$$

$$\Rightarrow P_c = 1.5 P_s$$

Substituting this in eq. (i)

$$P_s + 3 (1.5 P_s) = 80$$

$$\Rightarrow P_s = 14.5 \times 10^3 \text{ N}$$

$$\text{So, stress in steel rod, } \sigma_s = \frac{P_s}{A_s} = \frac{14.5 \times 10^3}{200} = 72.5 \text{ N/mm}^2$$

