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IC ENGINE

MECHANICAL ENGINEERING

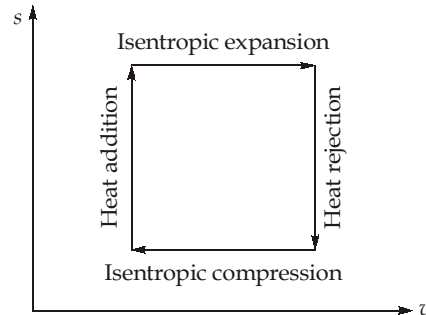
Date of Test : 10/09/2025

ANSWER KEY ➤

- | | | | | |
|--------|---------|---------|---------|---------|
| 1. (c) | 7. (b) | 13. (b) | 19. (a) | 25. (c) |
| 2. (b) | 8. (c) | 14. (d) | 20. (b) | 26. (d) |
| 3. (c) | 9. (d) | 15. (a) | 21. (d) | 27. (b) |
| 4. (a) | 10. (b) | 16. (b) | 22. (b) | 28. (c) |
| 5. (b) | 11. (c) | 17. (d) | 23. (a) | 29. (d) |
| 6. (b) | 12. (d) | 18. (a) | 24. (a) | 30. (c) |

DETAILED EXPLANATIONS

2. (b)



3. (c)

for N_2 gas, $\gamma = 1.66$

$$\text{Compression ratio, } r = \frac{v_1}{v_2} = \frac{v_1}{0.125v_1} = 8$$

$$\therefore \eta_{th} = 1 - \frac{1}{(r)^{\gamma-1}} = 1 - \frac{1}{(8)^{1.66-1}} = 0.7465$$

$$\text{Now } \eta_{th} = \frac{W_{net}}{Q_{in}}$$

$$\Rightarrow W_{net} = 0.7465 \times 150 = 111.9765 \text{ kW}$$

$$= \frac{111.9765 \times 1000}{746} = 150.10 \text{ hP}$$

4. (a)

$$BP = \frac{2\pi NT}{60 \times 1000} = \frac{2\pi \times 2500 \times 12}{60 \times 1000} = 3.14 \text{ kW}$$

$$\text{Friction power} = 3.5 - 3.14 = 0.358 \text{ kW}$$

$$\% \text{loss} = \frac{0.358}{3.5} \times 100 = 10.22 \%$$

5. (b)

$$\eta_m = \frac{\text{Brake power}}{\text{Indicated power}}$$

$$\Rightarrow \text{Indicated power} = \frac{\text{Brake power}}{\eta_m} = \frac{60}{0.8} = \frac{60}{8} \times 10 = 75 \text{ kW}$$

7. (b)

$$\eta = 1 - \frac{1}{(r)^{\gamma-1}} = 1 - \frac{1}{(9)^{0.5}} = 1 - \frac{1}{3} = 0.6667$$

$$\therefore \eta = 67 \%$$

9. (d)

Using characteristic gas equation

$$Pv = RT$$

$$v_1 = \frac{RT_1}{P_1} = \frac{287 \times 300}{1 \times 10^5} = 0.861 \text{ m}^3/\text{kg}$$

$$v_2 = \frac{0.861}{14} = 0.0615 \text{ m}^3/\text{kg}$$

$$\text{Swept volume} = v_1 - v_2 = 0.861 - 0.0615 = 0.7995 \text{ m}^3/\text{kg}$$

10. (b)

Compression ratio(r) = 8

$$v_1 = 0.9 \text{ m}^3/\text{kg}$$

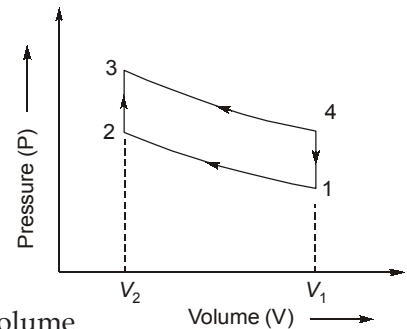
$$r = \frac{v_1}{v_2}$$

$$\Rightarrow v_2 = \frac{0.9}{8} = 0.1125 \text{ m}^3/\text{kg}$$

Work done = Mean effective pressure $\times$ swept volume

$$1575 \times 10^3 = P_m \times (0.9 - 0.1125)$$

$$\Rightarrow P_m = 20 \text{ bar}$$



11. (c)

$$\eta_I = 1 - \frac{1}{(r_1)^{r-1}} = 1 - \frac{1}{(4)^{0.4}} = 0.4256$$

$$\eta_{II} = 1 - \frac{1}{(r_2)^{r-1}} = 1 - \frac{1}{(6)^{0.4}} = 0.5116$$

$$\% \text{ Increase in thermal efficiency} = \frac{\eta_{II} - \eta_I}{\eta_I} \times 100 = \frac{0.5116 - 0.4256}{0.4256} \times 100 = 20.2\%$$

12. (d)

Given: $r = 14$, $R = 287 \text{ J/kgK}$, $c_p = 1005 \text{ J/kgK}$, $\gamma = 1.4$

$$T_1 = 27^\circ\text{C} = 300 \text{ K}$$

$$P_1 = 1 \text{ bar}$$

$$Q = c_p (T_3 - T_2)$$

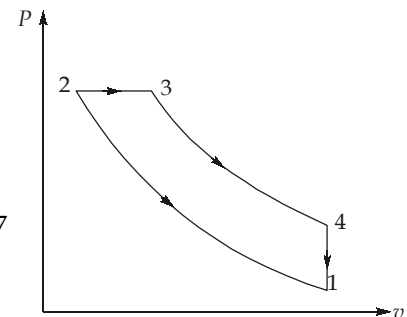
$$\frac{T_2}{T_1} = \left(\frac{V_1}{V_2} \right)^{\gamma-1} = (r)^{\gamma-1} = (14)^{1.4-1} = 2.8737$$

$$T_2 = 300 \times 2.8737 = 862.11 \text{ K}$$

$$\frac{T_3}{T_2} = \frac{v_3}{v_2} = 2$$

$$T_3 = 2T_2 = 1724.22 \text{ K}$$

$$Q = c_p (T_3 - T_2) = 1.005 \times (1724.22 - 862.11) = 866.4 \text{ kJ/kg}$$



13. (b)

Let BP at full load = x kW

BP at 70% of load = $0.70x$ kW

IP at 70% of load = $0.70x + FP$

$$\text{At 70\% load} \quad \eta_m = \frac{0.70x}{0.70x + FP} = 0.80$$

$$\Rightarrow 0.70x = 0.56x + 0.80FP$$

$$0.14x = 0.80 FP$$

$$FP = 0.175x$$

FP remains constant at all loads

$$\text{At full load} \quad IP = BP + FP = 50$$

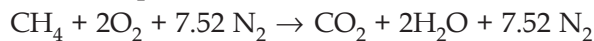
$$x + 0.175x = 50$$

$$x = \frac{50}{1.175} = 42.55 \text{ kW}$$

14. (d)

Air consists of 21% oxygen and 79% nitrogen thus for each mole of oxygen in air there are 3.76 moles of nitrogen.

The chemical equation of combustion of methane is given as



$$16 \text{ kg of CH}_4 \text{ requires} = 2 \times 32 + 7.52 \times 28$$

$$= 274.56 \text{ kg of air}$$

$$\Rightarrow \text{A/F ratio} = \frac{274.56}{16} = 17.16$$

$$\text{The density of the air} = \frac{P}{RT} = \frac{1.01 \times 10^5}{287 \times 295} = 1.1929 \text{ kg/m}^3$$

$$\text{Volume of air required per kg of fuel} = \frac{17.16}{1.1929} = 14.38 \text{ m}^3/\text{kg of fuel}$$

15. (a)

$$\text{If } V_2 = V_3 = 1, V_1 = V_4 = r$$

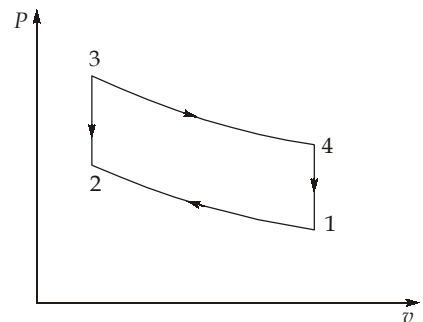
$$\frac{P_2}{P_1} = \left(\frac{V_1}{V_2} \right)^\gamma = (r)^\gamma = \frac{P_3}{P_4}$$

$$\therefore \frac{P_3}{P_2} = \frac{P_4}{P_1} = \frac{140}{95} = 1.473$$

Work done = area of P-v diagram

$$= \frac{P_3 v_3 - P_4 v_4}{\gamma - 1} - \frac{P_2 v_2 - P_1 v_1}{\gamma - 1}$$

$$= \frac{1}{\gamma - 1} \left[P_4 r \left(\frac{P_3}{P_4 \times r} - 1 \right) - P_1 r \left(\frac{P_2}{P_1 \times r} - 1 \right) \right]$$



$$= \frac{r}{\gamma-1} [P_4 (r^{\gamma-1} - 1) - P_1 (r^{\gamma} - 1)] = \frac{r}{\gamma-1} (r^{\gamma-1} - 1) (P_4 - P_1)$$

$$= \frac{7}{1.4-1} [(7)^{1.4-1} - 1] (140 - 95) = 927.60 \text{ kJ}$$

$$\text{Length of diagram} = r - 1 = 7 - 1 = 6$$

$$\text{mep} = \frac{\text{area of diagram}}{\text{length of the diagram}} = 927.60 / 6 = 154.6 \text{ kPa}$$

16. (b)

$$\eta_m = 0.8 = \frac{b.p.}{i.p.} \Rightarrow i.p. = \frac{50}{0.8} = 62.5 \text{ kW}$$

$$\Rightarrow \eta_{\text{ith}} = 0.35 = \frac{i.p.}{\dot{V}_f (C.V.)_f} \Rightarrow \dot{V}_f = \frac{62.5}{34 \times 1000 \times 0.35}$$

$$\dot{V}_f = 5.252 \times 10^{-3} \text{ m}^3/\text{s}$$

$$\Rightarrow \frac{A}{F} = \frac{10}{1} \Rightarrow \frac{\dot{V}_a}{\dot{V}_f} = 10 \Rightarrow \dot{V}_a = 0.05252 \text{ m}^3/\text{s}$$

$$\eta_v = \frac{(\dot{V}_a)_{\text{air}}}{\dot{V}_s} = \frac{0.05252}{2000 \times 10^{-6} \times \frac{N}{120}} = 0.7$$

$$\Rightarrow N = 4501.71 \text{ rpm}$$

17. (d)

$$\text{Compression ratio} = \text{expansion ratio} \times \text{cut-off ratio}$$

$$r = 15 \times 1.5$$

$$r = 22.5$$

$$\text{Now, } \eta_{\text{air-std}} = 1 - \frac{1}{r^{\gamma-1}} \frac{\rho^{\gamma} - 1}{\gamma(\rho - 1)}$$

$$\eta_{\text{air-std}} = 1 - \frac{1}{22.5^{0.4}} \times \frac{1.5^{1.4} - 1}{1.4(1.5 - 1)}$$

$$\eta_{\text{air-std}} = 0.6858$$

$$\text{Now, } \eta_{\text{rel}} = \frac{\eta_{\text{bth}}}{\eta_{\text{air-std}}}$$

$$\eta_{\text{bth}} = 0.4 \times 0.6858$$

$$\eta_{\text{bth}} = 0.27432$$

$$\frac{BP}{\dot{m}_f \times CV} = 0.27432$$

$$\dot{m}_f = \frac{30 \times 3600}{0.27432 \times 42000}$$

$$\dot{m}_f = 9.3738 \text{ kg/h}$$

18. (a)

$$\eta_{bth} = \frac{BP}{m_f \times CV}$$

$$\Rightarrow m_f = \frac{50}{0.25 \times 44000} = 4.545 \times 10^{-3} \text{ kg/sec} = 16.36 \text{ kg/hr}$$

$$= \frac{16.36}{0.74} = 22.1 \text{ l/hr}$$

$$\eta_{ith} = \frac{\eta_{bth}}{\eta_m} \quad \left\{ \eta_m = \frac{b.P.}{i.P.} = \frac{\eta_{bth}}{\eta_{ith}} \right\}$$

$$= \frac{0.25}{0.75} = 33.3 \%$$

19. (a)

In four stroke cycle there is one effective working stroke in every two revolution of the engine

crankshaft i.e. $n = \frac{N}{2}$

$$\text{Indicated power} = \frac{P_m L A n k}{1000} \text{ kW}$$

$$= \frac{4.5 \times 10^5 \times 0.25 \times \frac{\pi}{4} \times 0.2^2 \times \frac{360}{2 \times 60} \times 1}{1000} = 10.6 \text{ kW}$$

$$\text{Indicated thermal efficiency} = \frac{i.p.}{m_f \times CV} = \frac{10.6}{\frac{5}{3600} \times 44000} = 17.35 \%$$

20. (b)

$$\eta_v = \frac{V_a}{V_s} = 0.9$$

$\Rightarrow$

$$V_a = 0.9 V_s$$

mass of air,

$$m_a = \rho_{\text{air}} \times V_a = 1.0 \times 0.9 V_s = 0.9 V_s$$

$$\frac{m_f}{m_a} = \frac{F}{A}$$

$\Rightarrow$

$$m_f = \frac{F}{A} \times m_a = 0.05 \times 0.9 V_s = 0.045 V_s$$

$$\eta_{th} = \frac{P_{mep} \times V_s}{m_f \times CV} = \frac{6 \times 10^5 \times V_s \times 100}{0.045 \times V_s \times 45000 \times 1000} = 29.63 \%$$

21. (d)

$$\eta_{\text{Otto}} = 1 - \left(\frac{v_c}{v_c + v_s} \right)^{\gamma-1}$$

$$v_c = 196.3 \text{ cm}^3$$

$$v_s = \frac{\pi}{4} \times D^2 L = \frac{\pi}{4} \times 10^2 \times 15 = 1178.097 \text{ cm}^3$$

$$\gamma = 1.4$$

$$\therefore \eta_{\text{Otto}} = 1 - \left(\frac{196.3}{196.3 + 1178.097} \right)^{1.4-1} = 0.5408$$

$$\therefore \eta_{\text{Otto}} = 54.08\%$$

$$\therefore \text{Work output} = \eta_{\text{Otto}} \times (\text{Heat supplied}) = 0.5408 \times 1800 \text{ kJ} = 973.58 \text{ kJ}$$

22. (b)

$$\frac{V_1}{V_2} = 17, \quad \frac{C_p}{C_v} = \gamma = 1.4$$

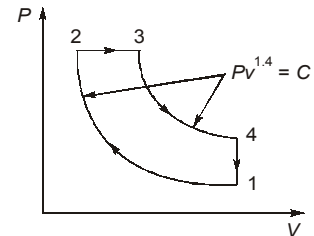
$$\text{or } V_3 - V_2 = 0.1 (V_1 - V_2)$$

$$\text{or } \frac{V_3}{V_2} - 1 = 0.1 \left(\frac{V_1}{V_2} - 1 \right)$$

$$\text{or } \frac{V_3}{V_2} = 0.1 \times 16 + 1 = 2.6$$

$$\eta_{\text{diesel}} = 1 - \frac{1}{r^{\gamma-1}} \left[\frac{r_c^{\gamma} - 1}{\gamma(r_c - 1)} \right] = 1 - \frac{1}{17^{0.4}} \left[\frac{2.6^{1.4} - 1}{1.4(2.6 - 1)} \right]$$

$$= 1 - \frac{1}{17^{0.4}} \left[\frac{3.81 - 1}{1.4 \times 1.6} \right] = 0.596 \text{ or } 59.6\%$$



23. (a)

$$\text{Brake power} = T\omega = \frac{300 \times 2\pi \times 500}{60 \times 10^3} \text{ kW}$$

$$BP = 15.7 \text{ kW}$$

Calculating indicated power:

$$\text{Spring constant} = \frac{10 \times 10^4 P_a}{1.4 \text{ mm}} = 71.43 \text{ kPa/mm}$$

$$\text{Indicated pressure} = \frac{\text{Area of indicator diagram}}{\text{Length of indicator diagram}} \times \text{Spring constant}$$

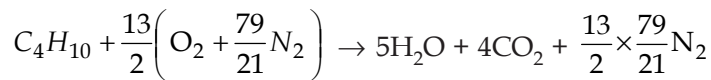
$$\text{Indicated pressure} = \frac{500}{60} \times 71.43 = 595.24 \text{ kPa}$$

$$\text{Indicated power} = 595.24 \times \frac{LAN}{60 \times 2} = 595.24 \times \frac{0.3 \times \frac{\pi}{4} \times 0.2^2 \times 500}{60 \times 2} = 23.375 \text{ kW}$$

$$\eta_m = \frac{bp}{ip} \times 100 = 67.165\%$$

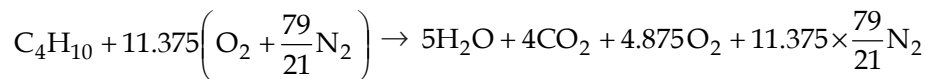
24. (a)

Stoichiometric reaction:



When 75% excess air is used:

$$\text{Moles of air per mole of butane} = 1.75 \times \frac{13}{2} = 11.375$$



$$\begin{aligned} \text{Mass percent of } CO_2 \text{ in dry exhaust} &= \frac{4m_{CO_2}}{4m_{CO_2} + 4.875m_{O_2} + 11.375 \times \frac{79}{21} m_{N_2}} \times 100 \\ &= \frac{4 \times 44}{4 \times 44 + 4.875 \times 32 + 11.375 \times \frac{79}{21} \times 28} \times 100 = 11.5020\% \end{aligned}$$

25. (c)

For compression, $pV^{1.3} = C$

$$\text{Now, } V_a = V_c + V_s - 0.05 V_s$$

$$V_a = V_c + 0.95 V_s$$

$$\text{and } V_b = V_c + V_s - 0.75 V_s$$

$$V_b = V_c + 0.25 V_s$$

$$\text{Now, } P_a V_a^{1.3} = P_b V_b^{1.3}$$

$$\frac{V_a}{V_b} = \left(\frac{P_b}{P_a} \right)^{1/1.3}$$

$$\frac{V_c + 0.95 V_s}{V_c + 0.25 V_s} = \left(\frac{4.8}{1.2} \right)^{1/1.3}$$

$$C = \frac{V_c}{V_s} = 0.1175$$

$$r = \frac{1}{C} + 1 = 9.5$$

$$\eta_{\text{airstd}} = 1 - \frac{1}{r^{\gamma-1}} = 0.5936$$

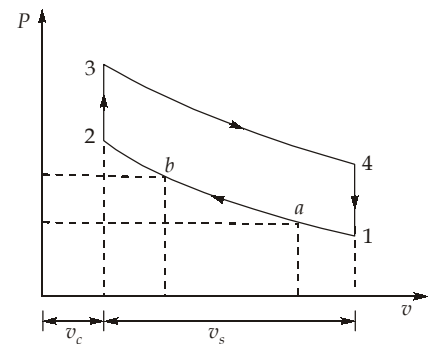
$$\eta_{\text{rel}} = \frac{\eta_{\text{bth}}}{\eta_{\text{airstd}}}$$

$$\Rightarrow \eta_{\text{bth}} = 0.3562$$

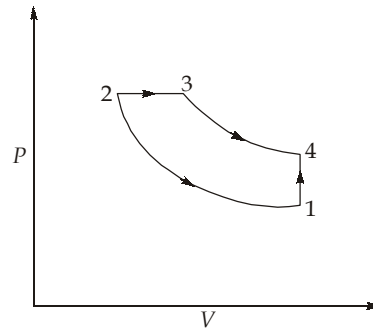
$$\text{Now } \eta_{\text{bth}} = \frac{bp}{\dot{m}_f \times CV} \Rightarrow \frac{\dot{m}_f}{bp} = \frac{1}{\eta_{\text{bth}} \times CV}$$

$$\text{So, specific fuel consumption} = \frac{\dot{m}_f}{bp} \times 3600 = \frac{3600}{\eta_{\text{bth}} \times CV}$$

$$sfc = 0.241 \text{ kg/kWh}$$



26. (d)



Compression ratio, $r = \frac{V_1}{V_2} = 15$

Expansion ratio, $r_e = \frac{V_4}{V_3} = 10$

Cut-off ratio, $r_c = \frac{r}{r_e} = \frac{V_1/V_2}{V_4/V_3} = \frac{V_3}{V_2} = \frac{15}{10} = 1.5$ [$V_1 = V_4$]

Air standard efficiency, $\eta = 1 - \frac{1}{\gamma} \frac{1}{r^{\gamma-1}} \left[\frac{r_c^\gamma - 1}{r_c - 1} \right] = 1 - \frac{1}{1.4} \times \frac{1}{15^{0.4}} \left[\frac{1.5^{1.4} - 1}{1.5 - 1} \right]$
 $= 0.6305 = 63.05\%$

27. (b)

Indicated power, $IP = P_{\text{eff imep}} \times V_s \times \frac{(\text{Number of explosion})}{\text{Time of test}}$... (i)

Now, effective indicated mean effective pressure,

$$P_{\text{eff, imep}} = P_{\text{imep}} - P_{\text{pumping mep}}$$

$$= 6.4 - 0.6 = 5.8 \text{ bar}$$

From equation (i),

$$IP = 5.8 \times 10^5 \times \frac{\pi}{4} \times \frac{0.25^2 \times (1.9 \times 0.25) \times 3300}{1000 \times 40 \times 60}$$

$$= 18.594 \text{ kW}$$

Brake power, $BP = (m_b \times g) \times r_d \times \left(\frac{2\pi \times \text{Number of revolution}}{\text{Time of test}} \right)$

$$= 90 \times 9.81 \times 0.8 \times \frac{2\pi \times 8080}{1000 \times 40 \times 60}$$

$$= 14.941 \text{ kW}$$

Mechanical efficiency, $\eta_m = \frac{BP}{IP} = \frac{14.941}{18.594} = 0.8035 = 80.35\%$

28. (c)

$$\text{Heat supplied/kg} = c_v (T_3 - T_2) + c_p (T_4 - T_3) \dots (i)$$

$$\frac{T_2}{T_1} = \left(\frac{V_2}{V_1} \right)^{\gamma-1} = (10)^{1.4-1}$$

$$\begin{aligned} T_2 &= (10)^{0.4} \times 300 \\ &= 753.57 \text{ K} = 480.57^\circ\text{C} \end{aligned}$$

Now, from equation (i),

$$878 = 0.717 (987 - 480.57) + 1.004(T_4 - 987)$$

$$T_4 = 1499.84^\circ\text{C} = 1772.84 \text{ K}$$

$$\text{Cut-off ratio, } r_c = \frac{V_4}{V_3}$$

For constant pressure process (3-4)

$$\frac{V_4}{V_3} = \frac{T_4}{T_3}$$

$$\text{So, } r_c = \frac{T_4}{T_3} = \frac{1772.84}{1260} = 1.407 \quad [\because T_3 = 987^\circ\text{C} = 1260 \text{ K}]$$

29. (d)

$$\text{Efficiency of Otto cycle} = 1 - \frac{1}{(r)^{\gamma-1}} \text{ (where, } r = \text{compression ratio)}$$

$$\eta_{\text{Otto}} = 1 - \frac{1}{(6)^{1.4-1}} = 0.5116 = 51.16\%$$

$$\begin{aligned} \text{Expansion ratio, } (e) &= \frac{V_4'}{V_2} = \frac{V_4'}{V_3} = \left(\frac{P_3}{P_4'} \right)^{1/\gamma} = \left(\frac{P_3}{P_1} \right)^{1/\gamma} \\ &= \left(\frac{20}{1} \right)^{1/1.4} = 8.497 \end{aligned}$$

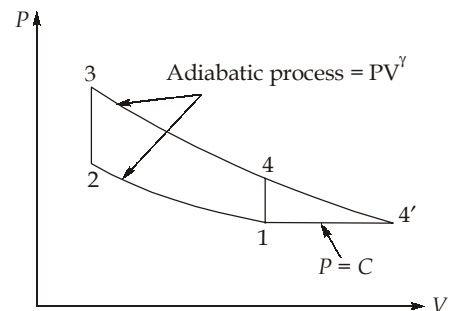
$$\text{Compression ratio, } r = 6$$

Efficiency of Atkinson cycle is given by

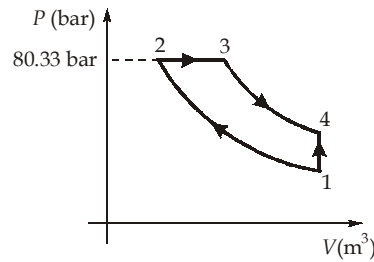
$$\eta_{\text{Atkinson}} = 1 - \gamma \frac{(e - r)}{(e^\gamma - r^\gamma)} = 1 - 1.4 \frac{(8.497 - 6)}{(8.497^{1.4} - 6^{1.4})}$$

$$= 0.5466 = 54.66\%$$

$$\text{So, Difference in efficiencies} = 54.66 - 51.16 = 3.5\%$$



30. (c)



$$P_1 = 1 \text{ bar}$$

$$T_1 = 300 \text{ K}$$

$$P_2 = P_3 = 80.33 \text{ bar}$$

$$\frac{T_2}{T_1} = (r)^{\gamma-1} \text{ or } \frac{P_2}{P_1} = r^{\gamma} \quad \left(\text{where, } r = \frac{V_1}{V_2} \right)$$

So,
$$r = \left(\frac{80.33}{1} \right)^{1/1.4} = 22.94$$

$$T_2 = T_1 \times (22.94)^{1.4-1} = 1050.44 \text{ K}$$

Heat energy added, $Q = C_p(T_3 - T_2)$

$$2000 = 1.005(T_3 - 1050.44)$$

$$T_3 = 3040.489 \text{ K}$$

Cut-off volume ratio,
$$\rho = \frac{V_3}{V_2} = \frac{T_3}{T_2} = \frac{3040.489}{1050.44} = 2.894$$

So, efficiency of diesel cycle

$$\begin{aligned}
 &= 1 - \frac{1}{(r)^{\gamma-1}} \frac{(\rho^{\gamma} - 1)}{\gamma(\rho - 1)} \\
 &= 1 - \frac{1}{(22.94)^{0.4}} \frac{((2.894)^{1.4} - 1)}{1.4(2.894 - 1)} = 0.63089 = 63.08\%
 \end{aligned}$$

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